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Find necessary and sufficient conditions for a numerical scheme to be derivable as a Godunov scheme (for some reconstruction)!

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Notes:

- Multi-d Cartesian grids
- Systems of hyperbolic PDEs
- Assume that the exact solution is known for all times and all kinds of initial data
- The reconstruction can be discontinuous and vary arbitrarily inside the cell

Vorticity preservation and low Mach number

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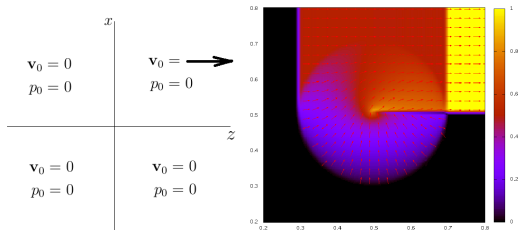
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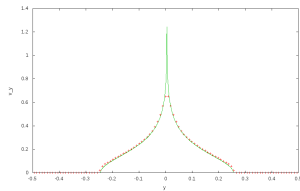
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Logarithmic singularity: $\mathbf{v} \sim -\log |\mathbf{x}|$

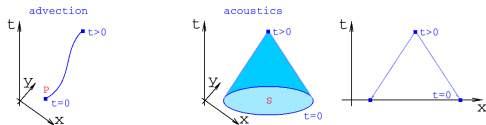


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Multi-d: characteristics become characteristic cones:

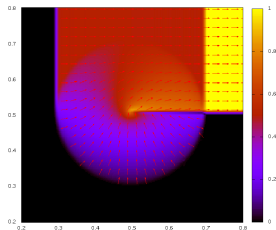
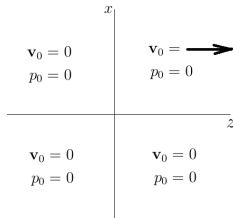


Characteristic cones do not only transport values:
derivatives of initial data are involved!

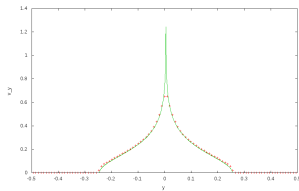
(Discontinuous initial data?)

Stationary vorticity:

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Amadori & Gosse 2015; WB & Klingenberg 2017, subm.

Euler equations

The acoustic equations are contained in the Euler equations:

$$\begin{aligned}\partial_t \rho + \mathbf{v} \cdot \nabla \rho + \rho \nabla \cdot \mathbf{v} &= 0 \\ \partial_t \mathbf{v} + \nabla p &= 0 \\ \partial_t p + c^2 \nabla \cdot \mathbf{v} &= 0\end{aligned}\qquad\qquad\begin{aligned}\partial_t \rho + \mathbf{v} \cdot \nabla \rho + \rho \nabla \cdot \mathbf{v} &= 0 \\ \partial_t \mathbf{v} + (\mathbf{v} \cdot \nabla) \mathbf{v} + \frac{\nabla p}{\rho} &= 0 \\ \partial_t p + \mathbf{v} \cdot \nabla p + \rho c^2 \nabla \cdot \mathbf{v} &= 0\end{aligned}$$

They capture the behaviour of **acoustics** and leave aside **advection**.

They also govern the (**Lagrangian**) evolution of a fluid element.

Numerics:

$$q := (\mathbf{v}, p) \in \mathbb{R}^{d+1} \quad \text{E.g. in 2-d} \quad \partial_t q_{ij} + \frac{f^x_{i+\frac{1}{2},j} - f^x_{i-\frac{1}{2},j}}{\Delta x} + \frac{f^y_{i,j+\frac{1}{2}} - f^y_{i,j-\frac{1}{2}}}{\Delta y} = 0$$

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Linear systems: $f^x = J^x q$, $f^y = J^y q$.

$$J^x = \begin{pmatrix} & & 1 \\ c^2 & 0 & \end{pmatrix} \quad J^y = \begin{pmatrix} 0 & & \\ & c^2 & \\ & & 1 \end{pmatrix}$$

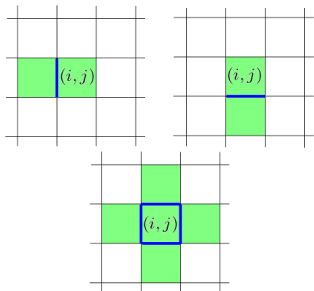
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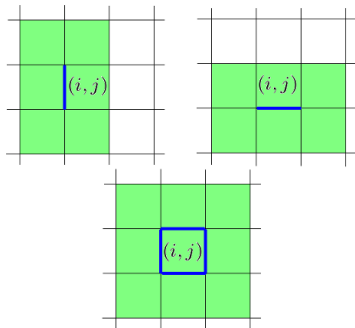
Upwind scheme (directionally split)



$$f_{i+\frac{1}{2},j}^x = \frac{1}{2} J^x (q_{i+1,j} + q_{ij}) - \frac{1}{2} D^x (q_{i+1,j} - q_{ij})$$
$$D_x = |J_x| \quad D_y = |J_y|$$

$$D^x = \begin{pmatrix} c & & \\ & 0 & \\ & & c \end{pmatrix} \quad D^y = \begin{pmatrix} 0 & & \\ & c & \\ & & c \end{pmatrix} \quad (1)$$

You can also use multi-d information:



Multi-dimensional schemes can appear very naturally!

E.g. Morton, Roe 2001: Define discrete derivative operator δ_x, δ_y . Rough idea:

$$\partial_t u + \partial_x p = 0$$

$$\partial_t v + \partial_y p = 0$$

$$\partial_t p + c^2 \nabla \cdot \mathbf{v} = 0$$

$$\partial_t u + \delta_x p = \delta_x^2 u + \delta_x \delta_y v$$

$$\partial_t v + \delta_y p = \delta_x \delta_y u + \delta_y^2 v$$

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Note the gradient on the right hand side!

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With the discrete averaging operator μ_x, μ_y one can actually show that for

$$\partial_t u + \delta_x \mu_x \mu_y^2 p = \delta_x^2 \mu_y^2 u + \delta_x \mu_x \delta_y \mu_y v$$

$$\partial_t v + \delta_y \mu_y \mu_x^2 p = \delta_x \mu_x \delta_y \mu_y u + \delta_y^2 \mu_x^2 v$$

$$\partial_t p + c^2 (\delta_x \mu_x \mu_y^2 u + \delta_y \mu_y \mu_x^2 v) = \delta_x^2 \mu_y^2 p + \delta_y^2 \mu_x^2 p$$

the vorticity is stationary: $\partial_t (\delta_x \mu_y v - \delta_y \mu_x u) = 0$.

Jeltsch & Torrilhon 2006; Mishra & Tadmor 2009; Lung & Roe 2014; ...

Actually, you don't need to be multi-d for vorticity preservation:

Directionally split Roe-type scheme (central + diffusion)

$$\partial_t q_{ij} + \frac{f_{i+\frac{1}{2},j}^x - f_{i-\frac{1}{2},j}^x}{\Delta x} + \frac{f_{i,j+\frac{1}{2}}^y - f_{i,j-\frac{1}{2}}^y}{\Delta y} = 0 \quad (2)$$

$$f_{i+\frac{1}{2},j}^x = \frac{1}{2} J^x (q_{i+1,j} + q_{ij}) - \frac{1}{2} D^x (q_{i+1,j} - q_{ij}) \quad (3)$$

$$D_x = \begin{pmatrix} a_1 & 0 & a_2 \\ 0 & 0 & 0 \\ a_3 & 0 & a_4 \end{pmatrix}$$

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You can show that, iff $a_1 = 0$,

$$\underbrace{\frac{v_{i+1,j} - v_{i-1,j}}{2\Delta x}}_{\simeq \partial_x v} - \underbrace{\frac{u_{i,j+1} - u_{i,j-1}}{2\Delta y}}_{\simeq \partial_y u} + \underbrace{\frac{a_3}{c^2} \left(\frac{u_{i,j+1} - 2u_{ij} + u_{i,j-1}}{2\Delta y} - \frac{v_{i+1,j} - 2v_{ij} + v_{i-1,j}}{2\Delta x} \right)}_{\simeq \mathcal{O}(\Delta x, \Delta y)} \quad (4)$$

is **stationary** (and a discretization of $\partial_x v - \partial_y u$)!

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Good luck with trying to prove this directly.

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2. There is no discrete stationary vorticity for the upwind scheme (1).
3. Vorticity preserving schemes for acoustics are just the ones that behave well in the limit of low Mach number.

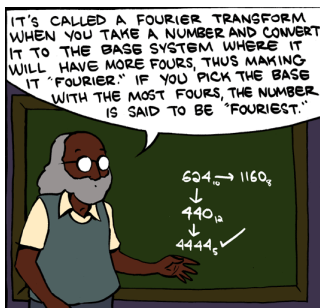
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Continuous case:

Consider the following $n \times n$ hyperbolic system of PDEs

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$$\text{i.e. } \partial_t q + \mathbf{J} \cdot \nabla q = 0 \quad (6)$$

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Insert a **Fourier mode**

$$q(t, \mathbf{x}) = \underbrace{\begin{pmatrix} \hat{u}(t, \mathbf{k}) \\ \hat{v}(t, \mathbf{k}) \\ \hat{p}(t, \mathbf{k}) \end{pmatrix}}_{=:\hat{q}(t, \mathbf{k})} \exp(i\mathbf{k} \cdot \mathbf{x}) \quad (7)$$

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$$\partial_t \hat{q} + \mathbf{i}J_x k_x \hat{q} + \mathbf{i}J_y k_y \hat{q} = 0 \quad (8)$$

$$\text{i.e. } \partial_t \hat{q} + \mathbf{i}(\mathbf{J} \cdot \mathbf{k}) \hat{q} = 0 \quad (9)$$

Derivatives become algebraic factors.

$$\partial_t \hat{q} + \mathfrak{i}(\mathbf{J} \cdot \mathbf{k}) \hat{q} = 0 \quad (10)$$

Recall that **hyperbolicity** guarantees that $\mathbf{J} \cdot \mathbf{k}$ is real diagonalizable:

$$(\mathbf{J} \cdot \mathbf{k}) e_j = \lambda_j e_j \quad j = 1, \dots, n \quad (11)$$

The solution to $\partial_t \hat{q} + \mathfrak{i}(\mathbf{J} \cdot \mathbf{k}) \hat{q} = 0$ can thus be constructed out of these eigenvectors:

$$\hat{q}(t, \mathbf{k}) = \sum_{j=1}^n (\hat{q}_0 \cdot e_j) e_j \exp(-\mathfrak{i} \lambda_j t) \quad (12)$$

Every mode evolves with its own eigenvalue.

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Definition (Nontrivial stationary states)

$\partial_t q + \mathbf{J} \cdot \nabla q = 0$ possesses *nontrivial stationary states* if $\det \mathbf{J} \cdot \mathbf{k} = 0 \quad \forall \mathbf{k} \in \mathbb{R}^d$.

For example, linear advection has stationary states, but no nontrivial ones...

For linear acoustics in 2-d

$$\mathbf{J} \cdot \mathbf{k} = \begin{pmatrix} 0 & 0 & k_x \\ 0 & 0 & k_y \\ k_x c^2 & k_y c^2 & 0 \end{pmatrix} \quad (13)$$

has eigenvalues $(0, \pm c|\mathbf{k}|)$ such that

$$\hat{q}(t, \mathbf{k}) = (\hat{q}_0 \cdot \mathbf{e}_1) \mathbf{e}_1 + (\hat{q}_0 \cdot \mathbf{e}_2) \mathbf{e}_2 \cdot \exp(i c t |\mathbf{k}|) + (\hat{q}_0 \cdot \mathbf{e}_3) \mathbf{e}_3 \cdot \exp(-i c t |\mathbf{k}|) \quad (14)$$

with

$$\mathbf{e}_1 = \begin{pmatrix} -k_y \\ k_x \\ 0 \end{pmatrix} \quad (15)$$

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Solution is stationary, if initial data only contain modes parallel to \mathbf{e}_1 !

$$\hat{q}(t, \mathbf{k}) = (\hat{q}_0 \cdot e_1)e_1 + (\hat{q}_0 \cdot e_2)e_2 \cdot \exp(i\mathbf{c}t|\mathbf{k}|) + (\hat{q}_0 \cdot e_3)e_3 \cdot \exp(-i\mathbf{c}t|\mathbf{k}|) \quad (16)$$

$$e_1 = \begin{pmatrix} -k_y \\ k_x \\ 0 \end{pmatrix} \quad (17)$$

Define projector P_1 onto e_1

$$P_1 e_2 = P_1 e_3 = 0 \qquad P_1 e_1 = e_1 \quad (18)$$

Then **for all initial data**

$$P_1 \hat{q}(t, \mathbf{k}) = P_1 \hat{q}(0, \mathbf{k}) \quad (19)$$

i.e.

$$\partial_t (P_1 \hat{q}) = 0 \quad (20)$$

$$\hat{q}(t, \mathbf{k}) = (\hat{q}_0 \cdot e_1)e_1 + (\hat{q}_0 \cdot e_2)e_2 \cdot \exp(i\mathbf{ct}|\mathbf{k}|) + (\hat{q}_0 \cdot e_3)e_3 \cdot \exp(-i\mathbf{ct}|\mathbf{k}|) \quad (16)$$

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For acoustics in 2-d: $P_1 \hat{q} = -k_y \hat{u} + k_x \hat{v} = \mathbb{F}[\nabla \times \mathbf{v}] \rightarrow$ **stationary vorticity (involution, or constant of motion).**

Existence of nontrivial stationary states is equivalent to the existence of involutions.

Discrete Fourier modes, e.g. in 2-d

$$q_{ij}(t) = \hat{q}(t, \mathbf{k}) \exp\left(\mathbb{i}k_x(i\Delta x) + \mathbb{i}k_y(j\Delta y)\right) \quad \text{instead of} \quad q(t, \mathbf{x}) = \hat{q}(t, \mathbf{k}) \exp(\mathbb{i}k_x \mathbf{x} + \mathbb{i}k_y \mathbf{y})$$

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Consider a finite difference, e.g.

$$\frac{1}{2\Delta x} (q_{i+1,j} - q_{i-1,j}) \quad (21)$$

Inserting the Fourier mode

$$\exp\left(\mathfrak{i}k_x(i\Delta x) + \mathfrak{i}k_y(j\Delta y)\right) \cdot \frac{1}{2\Delta x} \left(\exp(\mathfrak{i}k_x \Delta x) - \exp(-\mathfrak{i}k_x \Delta x)\right) \hat{q} \quad (22)$$

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Obviously, the factor $\exp(\mathfrak{i}k_x \Delta x) =: t_x$ acts as shift by one cell to the right. The above can be rewritten as

$$\exp\left(\mathfrak{i}k_x(i\Delta x) + \mathfrak{i}k_y(j\Delta y)\right) \cdot \frac{1}{2\Delta x} (t_x - t_x^{-1}) \hat{q} \quad (23)$$

Discrete Fourier modes, e.g. in 2-d

$$q_{ij}(t) = \hat{q}(t, \mathbf{k}) \exp\left(\mathfrak{i}k_x(i\Delta x) + \mathfrak{i}k_y(j\Delta y)\right) \quad \text{instead of} \quad q(t, \mathbf{x}) = \hat{q}(t, \mathbf{k}) \exp(\mathfrak{i}k_x \mathbf{x} + \mathfrak{i}k_y \mathbf{y})$$

Consider a finite difference, e.g.

$$\frac{1}{2\Delta x} (q_{i+1,j} - q_{i-1,j}) \quad (21)$$

Inserting the Fourier mode

$$\exp\left(\mathfrak{i}k_x(i\Delta x) + \mathfrak{i}k_y(j\Delta y)\right) \cdot \frac{1}{2\Delta x} \left(\exp(\mathfrak{i}k_x \Delta x) - \exp(-\mathfrak{i}k_x \Delta x)\right) \hat{q} \quad (22)$$

Obviously, the factor $\exp(\mathfrak{i}k_x \Delta x) =: t_x$ acts as shift by one cell to the right. The above can be rewritten as

$$\exp\left(\mathfrak{i}k_x(i\Delta x) + \mathfrak{i}k_y(j\Delta y)\right) \cdot \frac{1}{2\Delta x} (t_x - t_x^{-1}) \hat{q} \quad (23)$$

There is a bijection between linear finite difference formulae and Laurent polynomials in t_x, t_y . E.g.

$$\frac{1}{8\Delta x} \left((q_{i+1,j+1} + 2q_{i+1,j} + q_{i+1,j-1}) - (q_{i-1,j+1} + 2q_{i-1,j} + q_{i-1,j-1}) \right) \quad (24)$$

$$\simeq \frac{1}{8\Delta x} \left((t_x t_y + 2t_x + t_x t_y^{-1}) - (t_x^{-1} t_y + 2t_x^{-1} + t_x^{-1} t_y^{-1}) \right) \quad (25)$$

$$= \frac{(t_x + 1)(t_x - 1)}{2t_x \Delta x} \cdot \frac{(t_y + 1)^2}{4t_y} \hat{q} \quad (26)$$

$\frac{t_x - 1}{\Delta x}$	discretization of ∂_x
$\frac{t_x + 1}{2} \frac{t_x - 1}{\Delta x}$	central difference $q_{i+1} - q_{i-1}$
$\frac{(t_x - 1)^2}{\Delta x^2}$	second derivative $\frac{q_{i+1} - 2q_i + q_{i-1}}{\Delta x^2}$

Lemma

The Fourier transform of a finite difference that approximates ∂_x^n contains precisely n factors $t_x - 1$.

Lemma (Normalization)

Consider the Fourier transform $\frac{1}{t_x^k} \prod_{j=1}^{2k} (t_x - s_j)$ with $s_j = 1$ for $j = 1, 2, \dots, n$ and

otherwise $s_j \neq 1$. Then, as $\Delta x \rightarrow 0$, it approximates $A \Delta x^n \partial_x^n$ with $A = \prod_{j=1, s_j \neq 1}^{2k} (1 - s_j)$.

Example:

Consider a **central scheme** for linear acoustics (**unstable with explicit time integration!**)

$$\partial_t q + J_x \partial_x q + J_y \partial_y q = 0 \quad (27)$$

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In the discrete, the evolution matrix \mathcal{E} plays the role of $\mathfrak{i}(\mathbf{J} \cdot \mathbf{k})$.

Example (cont.):

$$\partial_t \hat{q}(t) + \mathcal{E} \hat{q} = 0 \quad (30)$$

Observe that, as before, with $\mathcal{E}e_j = \lambda_j e_j$

$$\hat{q}(t) = \sum_{j=1}^n (q_0 \cdot e_j) e_j \exp(-i\lambda_j t) \quad (31)$$

For example, for the acoustic equations the evolution of the discrete mode is

$$\hat{q}(t) = (\hat{q}_0 \cdot e_1) e_1 + (\hat{q}_0 \cdot e_2) e_2 \exp(i\alpha t) + (\hat{q}_0 \cdot e_3) e_3 \exp(-i\alpha t) \quad (32)$$

with

$$\alpha = \sqrt{\frac{\sin^2(k_y \Delta y)}{\Delta y^2} + \frac{\sin^2(k_x \Delta x)}{\Delta x^2}} \xrightarrow{\Delta x, \Delta y \rightarrow 0} |\mathbf{k}| \quad (33)$$

$$e_1 = \begin{pmatrix} -\frac{t_y^2 - 1}{t_y \Delta y} \\ \frac{t_x^2 - 1}{t_x \Delta x} \\ 0 \end{pmatrix} \simeq \frac{u_{i+1,j} - u_{i-1,j}}{2\Delta x} + \frac{v_{i,j+1} - v_{i,j-1}}{2\Delta y} = 0 \text{ and } p = \text{const} \quad (34)$$

$$\partial_t \hat{q} + \mathcal{E} \hat{q} = 0 \quad (35)$$

The projector onto the zero-eigenvector of \mathcal{E} yields a **numerical constant of motion**, or **discrete involution** (discretization of vorticity in the case of acoustic equations).

For stability under explicit time integration, one needs to add numerical diffusion:

$$f_{i+\frac{1}{2},j}^x = \frac{1}{2}J^x(q_{i+1,j} + q_{ij}) - \frac{1}{2}D^x(q_{i+1,j} - q_{ij}) \quad (36)$$

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Evolution matrix:

$$\mathcal{E} = J_x \frac{t_x^2 - 1}{2\Delta x t_x} + J_y \frac{t_y^2 - 1}{2\Delta y t_y} - D_x \frac{(t_x - 1)^2}{2\Delta x t_x} - D_y \frac{(t_y - 1)^2}{2\Delta y t_y} \quad (40)$$

What are its eigenvalues?

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What are its eigenvalues? Actually, for the **upwind scheme** ($D_x = |J_x|$, $D_y = |J_y|$)

$$\det \mathcal{E} \neq 0$$

All modes instationary!

(von Neumann stability tells you what happens with them)

Definition

Stationarity preserving scheme: $\dim \ker \mathcal{E} = \dim \ker(\mathbf{J} \cdot \mathbf{k}) \forall \mathbf{k}$.

I. e. the scheme's stationary states discretize all the analytic stationary states.

Theorem

The upwind scheme is not stationarity preserving.

WB 2017; WB 2018, subm.

Consider the initial example

$$D_x = \begin{pmatrix} a_1 & 0 & a_2 \\ 0 & 0 & 0 \\ a_3 & 0 & a_4 \end{pmatrix}$$

$$D_y = \begin{pmatrix} 0 & 0 & 0 \\ 0 & a_1 & a_2 \\ 0 & a_3 & a_4 \end{pmatrix}$$

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You can show that, iff $a_1 = 0$, then $\det \mathcal{E} = 0$, and the numerical scheme becomes stationarity preserving.

The (right) eigenvector corresponding to eigenvalue zero is

$$e_1 = \begin{pmatrix} \frac{a_3(t_y - 2 + t_y^{-1})}{2\Delta y} - \frac{c^2(t_y - t_y^{-1})}{2\Delta y} \\ -\frac{a_3(t_x - 2 + t_x^{-1})}{2\Delta x} + \frac{c^2(t_x - t_x^{-1})}{2\Delta x} \\ 0 \end{pmatrix} \quad (41)$$

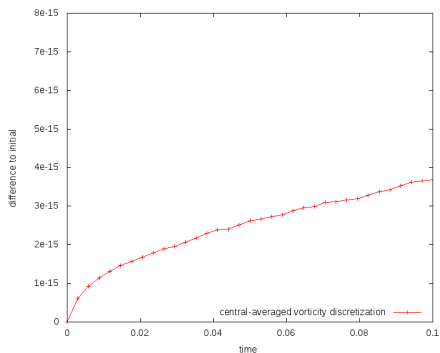
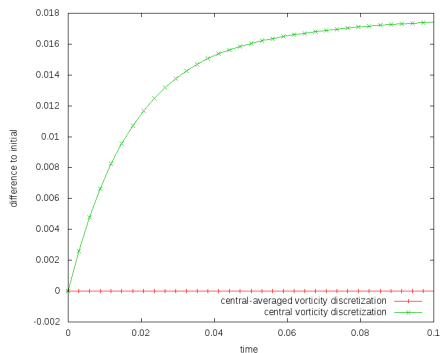
and therefore the vorticity discretization

$$\frac{v_{i+1,j} - v_{i-1,j}}{2\Delta x} - \frac{u_{i,j+1} - u_{i,j-1}}{2\Delta y} + \frac{a_3}{c^2} \left(\frac{u_{i,j+1} - 2u_{ij} + u_{i,j-1}}{2\Delta y} - \frac{v_{i+1,j} - 2v_{ij} + v_{i-1,j}}{2\Delta x} \right) \quad (42)$$

is stationary (as claimed in the introduction).

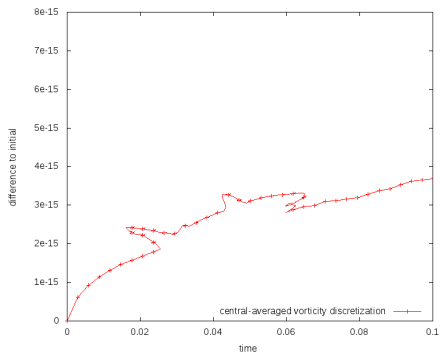
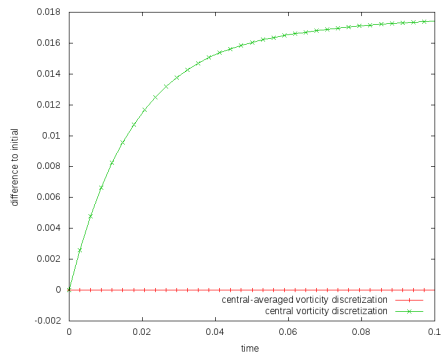
So far:

- Classification of all **vorticity preserving** schemes for linear acoustics
- Vorticity preservation is equivalent to the existence of nontrivial stationary states (**stationarity preserving**)



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Low Mach number limit

Take $\epsilon > 0$, and choose $M_{\text{loc}} = \frac{|\mathbf{v}|}{\sqrt{\gamma p/\rho}} \in \mathcal{O}(\epsilon)$ as $\epsilon \rightarrow 0$. Then this corresponds to solving

$$\begin{aligned} \partial_t \rho + \mathbf{v} \cdot \nabla \rho + \rho \nabla \cdot \mathbf{v} &= 0 \\ \partial_t \mathbf{v} + \frac{\nabla p}{\epsilon^2} &= 0 & \partial_t \rho + \mathbf{v} \cdot \nabla \rho + \rho \nabla \cdot \mathbf{v} &= 0 \\ \partial_t p + c^2 \nabla \cdot \mathbf{v} &= 0 & \partial_t \mathbf{v} + (\mathbf{v} \cdot \nabla) \mathbf{v} + \frac{\nabla p}{\rho \epsilon^2} &= 0 \\ & & \partial_t p + \mathbf{v} \cdot \nabla p + \rho c^2 \nabla \cdot \mathbf{v} &= 0 \end{aligned}$$

Acoustics “inherits” a low Mach number limit.

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$$\mathbf{J} \cdot \mathbf{k} = \begin{pmatrix} 0 & 0 & k_x/\epsilon^2 \\ 0 & 0 & k_y/\epsilon^2 \\ k_x c^2 & k_y c^2 & 0 \end{pmatrix} \quad (43)$$

has eigenvalues $\left(0, \pm \frac{c|\mathbf{k}|}{\epsilon}\right)$ such that

$$\hat{q}(t, \mathbf{k}) = (\hat{q}_0 \cdot \mathbf{e}_1) \mathbf{e}_1 + (\hat{q}_0 \cdot \mathbf{e}_2) \mathbf{e}_2 \exp\left(\mathbf{i} \frac{ct}{\epsilon} |\mathbf{k}|\right) + (\hat{q}_0 \cdot \mathbf{e}_3) \mathbf{e}_3 \exp\left(-\mathbf{i} \frac{ct}{\epsilon} |\mathbf{k}|\right) \quad (44)$$

Formal asymptotic analysis:

$$\mathbf{v} = \mathbf{v}^{(0)} + \mathbf{v}^{(1)}\epsilon + \mathbf{v}^{(2)}\epsilon^2 + \dots \quad (45)$$

$$p = p^{(0)} + p^{(1)}\epsilon + p^{(2)}\epsilon^2 + \dots \quad (46)$$

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Limit equations:

$$\nabla p^{(0)} = 0 \quad \nabla \cdot \mathbf{v}^{(0)} = 0 \quad (49)$$

$$\nabla p^{(1)} = 0 \quad (50)$$

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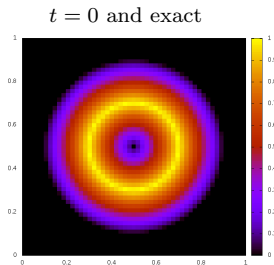
$$\hat{q}(t, \mathbf{k}) = (\hat{q}_0 \cdot \mathbf{e}_1) \mathbf{e}_1 + (\hat{q}_0 \cdot \mathbf{e}_2) \mathbf{e}_2 \exp\left(\mathrm{i} \frac{ct}{\epsilon} |\mathbf{k}|\right) + (\hat{q}_0 \cdot \mathbf{e}_3) \mathbf{e}_3 \exp\left(-\mathrm{i} \frac{ct}{\epsilon} |\mathbf{k}|\right) \quad (52)$$

This means that

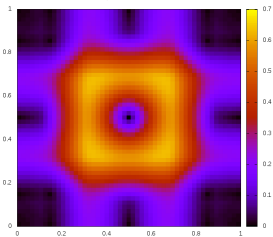
$$\boxed{\lim_{\epsilon \rightarrow 0, t \text{ fixed}} q = \lim_{\epsilon \text{ fixed}, t \rightarrow \infty} q} \quad (53)$$

Low Mach number limit is the same as the long time limit.

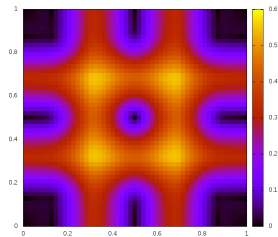
E.g. for the upwind/Roe scheme



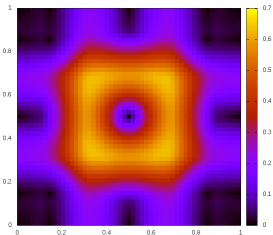
$\epsilon = 1, t = 2$



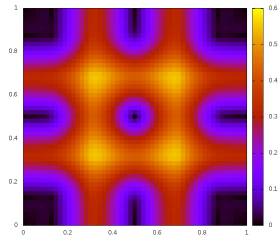
$\epsilon = 1, t = 5$



$\epsilon = \frac{1}{2}, t = 1$



$\epsilon = \frac{1}{5}, t = 1$



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Vorticity preserving \Leftrightarrow Stationarity preserving \Leftrightarrow Low Mach compliant

WB 2017; WB 2018, subm.

Summary:

- acoustics is an atomic system of equations in multi-d
- classification of all vorticity preserving schemes
- new understanding of the performance of schemes in the low Mach number limit
- complete understanding of the behaviour of linear schemes for acoustics

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- acoustics is an atomic system of equations in multi-d
- classification of all vorticity preserving schemes
- new understanding of the performance of schemes in the low Mach number limit
- complete understanding of the behaviour of linear schemes for acoustics

Vorticity preserving \Leftrightarrow Stationarity preserving \Leftrightarrow Low Mach compliant

- you can treat similarly all kinds of linear systems (e.g. taking into account source terms)
- truly multi-dimensional discretizations (uniqueness)
- stationarity preserving limiters for higher order schemes?
- there exist nonlinear versions of stationarity preservation / vorticity “preservation”: relation to low Mach number?

Challenge:

Generalization to non-Cartesian grids?