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# HIGHER-ORDER FV-MLS METHOD FOR THE INCOMPRESSIBLE NAVIER-STOKES EQUATIONS

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# Outline

- Introduction
- The FV-MLS method
- A high-order formulation for incompressible flows
- High-order Fluid-Structure-Interaction techniques
- Conclusions





# Introduction

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# FV-MLS Applications

## ► FV-MLS Applications

- All-speed Navier-Stokes
- Incompressible Navier-Stokes
- Linearized Euler Equations (acoustics)
- Navier-Stokes Korteweg equations
- Turbulence (ILES)
- High-order Sliding mesh applications
- Cavitating flows
- ...

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Wednesday @11:00 by Xesús Nogueira





# The FV-MLS method

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- A high-order formulation for incompressible flows
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- Conclusions





# The Finite Volume Method

- ▶ Let us consider a generic conservation law for the 2D domain  $\Omega_T$

$$\frac{\partial U}{\partial t} + \nabla \cdot \mathcal{F} = S$$

- ▶ Finite Volume discretization over  $\Omega_I$ :

$$\Omega_I \int_{\Omega_I} \frac{\partial U_I}{\partial t} d\Omega + \int_{\Gamma_I} (\mathcal{F}^{\mathcal{H}} - \mathcal{F}^{\mathcal{V}}) \cdot \mathbf{n} d\Gamma = 0$$

- $\mathcal{F}^{\mathcal{H}}$  → Hyperbolic-like term
- $\mathcal{F}^{\mathcal{V}}$  → Elliptic-like term

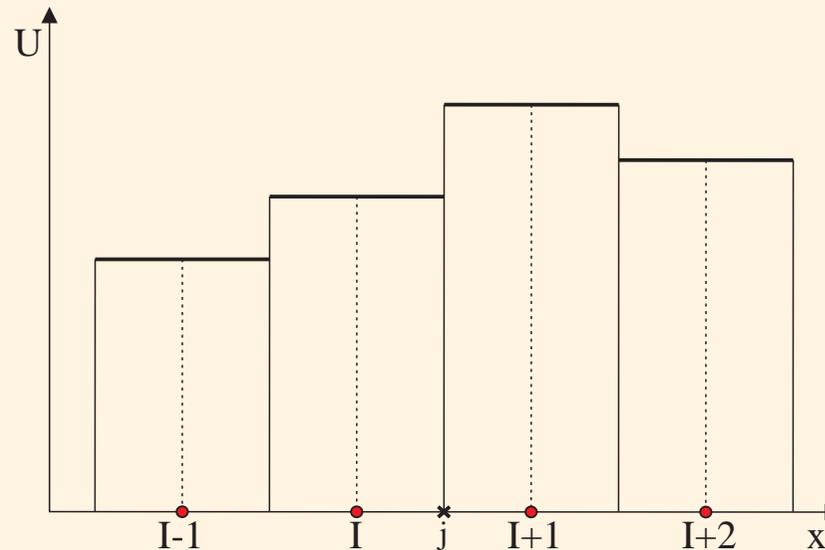




# The Finite Volume Method

## ► Hyperbolic term:

- Godunov approach



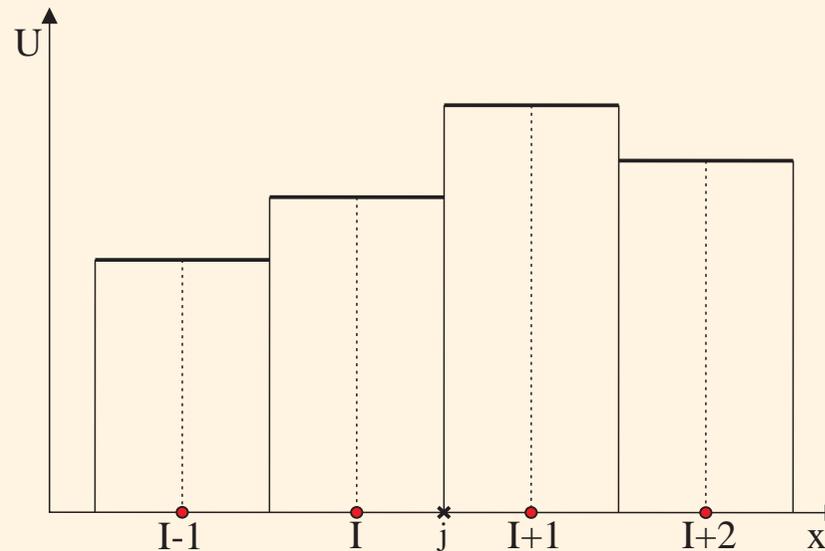
- $\mathcal{F}^H$  is the solution of a Riemann problem
- Initial values  $\rightarrow$  variables at both sides of the interface.



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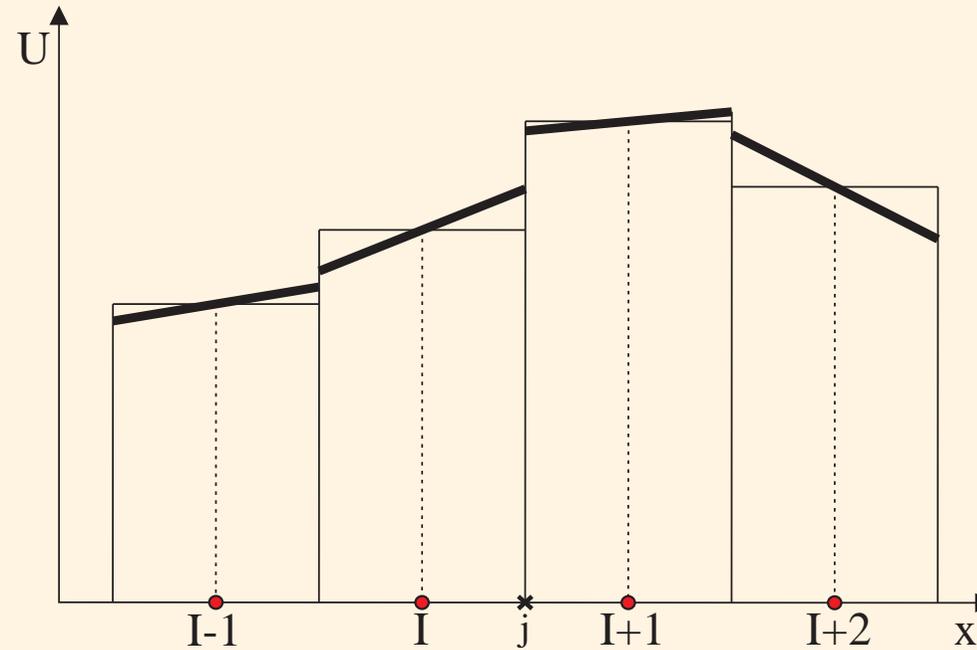
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$$U_L \neq U_R$$



# The Finite Volume Method

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$$U_L = U_I + \nabla U_I \cdot (\mathbf{x}_j - \mathbf{x}_I)$$

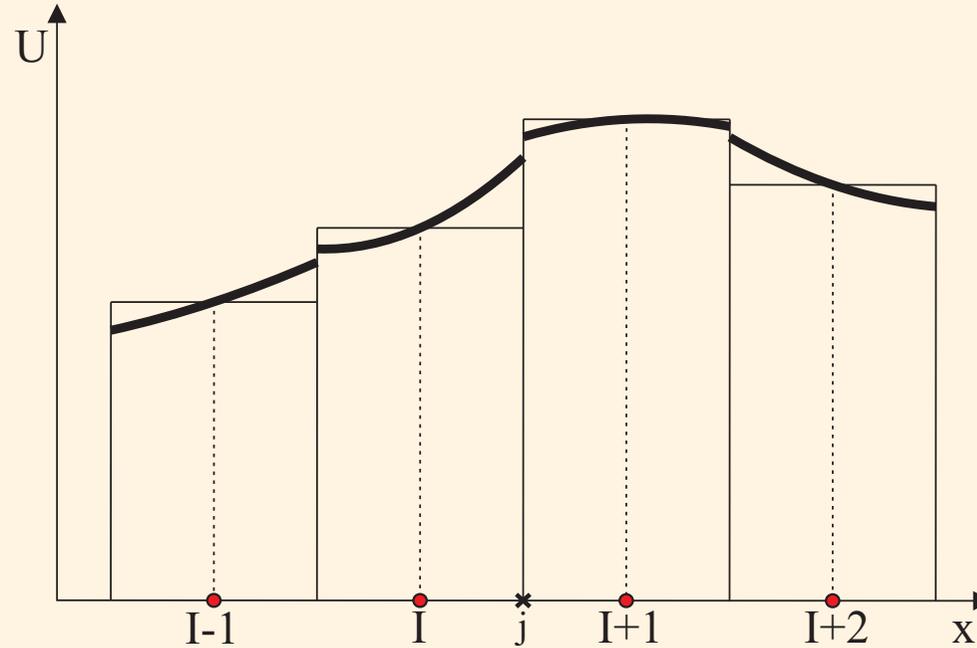
$$U_R = U_{I+1} + \nabla U_{I+1} \cdot (\mathbf{x}_j - \mathbf{x}_{I+1})$$





# The Finite Volume Method

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$$U_L = U_I + \nabla U_I \cdot (\mathbf{x}_j - \mathbf{x}_I) + \frac{1}{2} (\mathbf{x}_j - \mathbf{x}_I)^T \mathbf{H}_I (\mathbf{x}_j - \mathbf{x}_I)$$

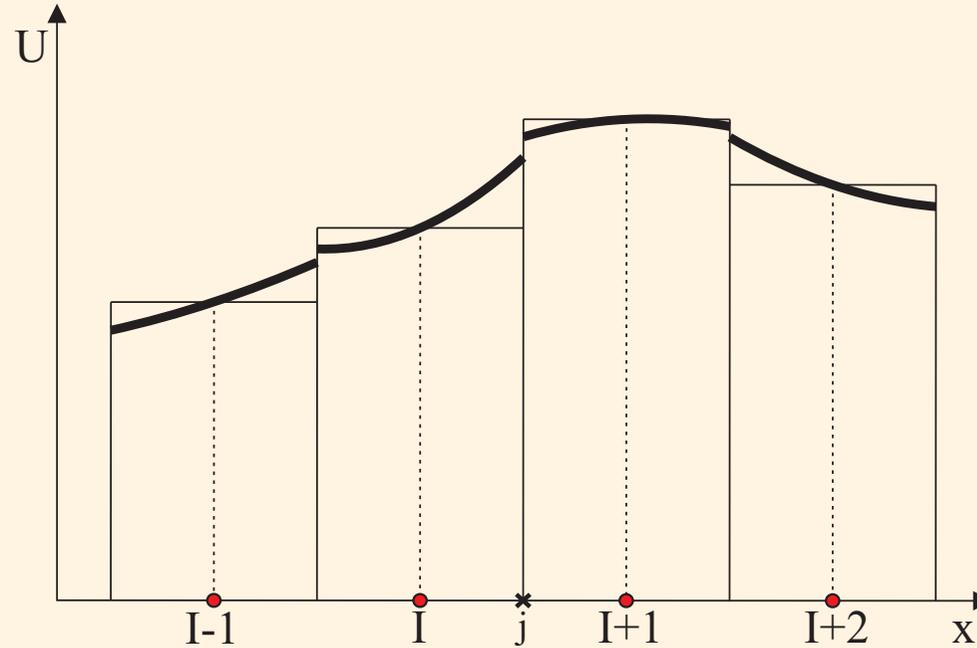
$$U_R = U_{I+1} + \nabla U_{I+1} \cdot (\mathbf{x}_j - \mathbf{x}_{I+1}) + \frac{1}{2} (\mathbf{x}_j - \mathbf{x}_{I+1})^T \mathbf{H}_{I+1} (\mathbf{x}_j - \mathbf{x}_{I+1})$$





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## Finite Volume Method. High-order schemes (II)

### ► Computation of high-order derivatives:

- Easy on structured grids.
- Unstructured grids  $\Rightarrow$  **PROBLEM**.

### ► We propose:

- The use of **Moving Least Squares (MLS)** to obtain an **accurate** and **multidimensional** approximation of derivatives on unstructured grids.





## The basis: Kernel approximations (I)

- ▶ Kernel approximation is based on the properties of **Dirac's Delta** distribution:

$$u(\mathbf{x}) = \int_{\mathbf{y} \in \Omega} u(\mathbf{y}) \delta(\mathbf{x} - \mathbf{y}) d\Omega$$

- ▶ Kernel approximation is defined as:

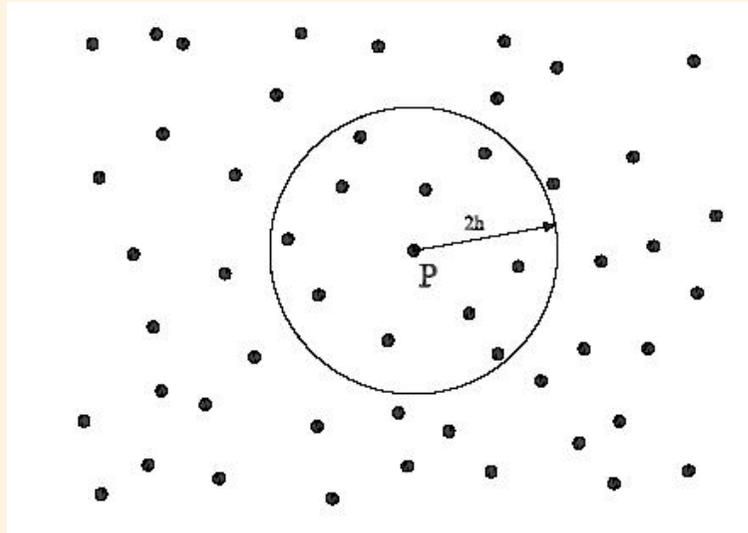
$$u^h(\mathbf{x}) = \int_{\mathbf{y} \in \Omega} u(\mathbf{y}) W(\mathbf{x} - \mathbf{y}, \rho) d\Omega$$





## The basis: Kernel approximations (II)

- ▶ In discrete form:



$$\hat{u}(\mathbf{x}) = \sum_{j=1}^n u_j W(\mathbf{x} - \mathbf{x}_j, h) V_j$$

- ▶  $V_j$  is the statistical volume of a particle  $j$ .
- ▶ Compact support with  $r = 2h$
- ▶  $h$  is the smoothing length.

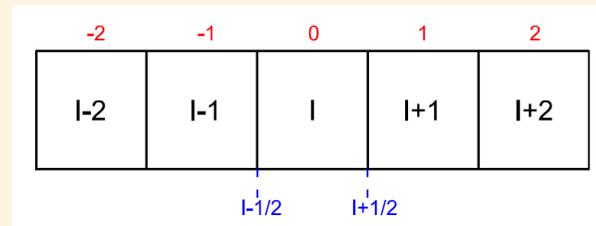
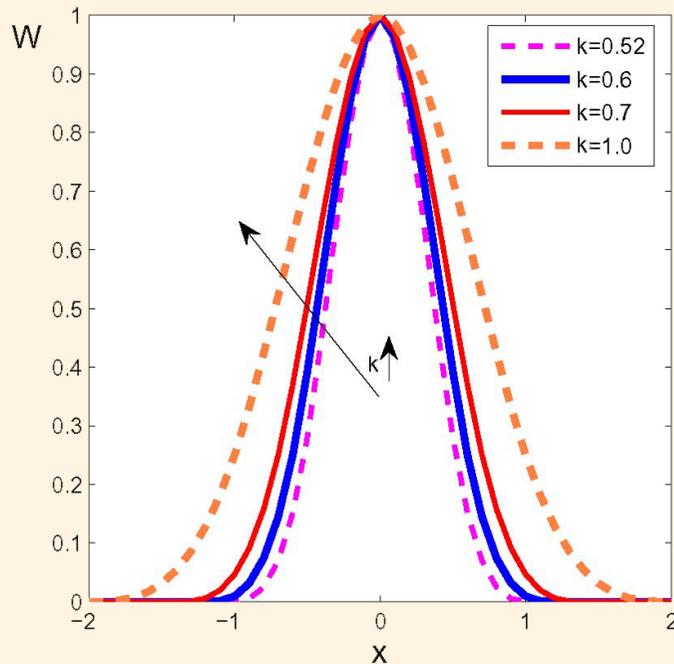


# The basis: Kernel approximations (IV)

- ▶ Many functions used as kernels: splines, gaussians
- ▶ An example, the cubic spline:

$$W_j(\mathbf{x}) = W(\mathbf{x} - \mathbf{x}_j, h) = \frac{\alpha}{h^\nu} \begin{cases} 1 - \frac{3}{2}s^2 + \frac{3}{4}s^3 & s \leq 1 \\ \frac{1}{4}(2 - s)^3 & 1 < s \leq 2 \\ 0 & s > 2 \end{cases}$$

$$s = \frac{\|\mathbf{x} - \mathbf{x}_j\|}{h} \quad h = k \max(\|\mathbf{x} - \mathbf{x}_j\|)$$



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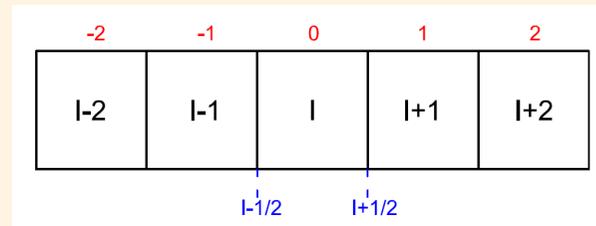
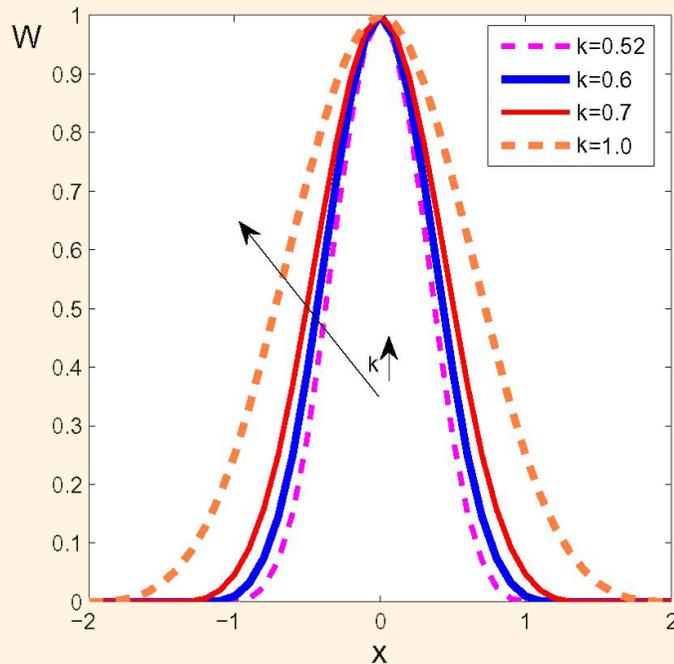


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## The basis: Kernel approximations (V)

- ▶ Another example: Exponential Kernel.

$$W(x, x^*, \kappa) = \frac{e^{-\left(\frac{s}{c}\right)^2} - e^{-\left(\frac{d_m}{c}\right)^2}}{1 - e^{-\left(\frac{d_m}{c}\right)^2}}$$

$$s = |x - x^*|, d_m = 2 \max(|x_j - x^*|), c = \frac{d_m}{2\kappa}$$

- ▶ 2D kernel  $\Rightarrow$  product of two 1D kernels:

$$W_j(\mathbf{x}, \mathbf{x}^*, \kappa_x, \kappa_y) = W_j(x, x^*, \kappa_x) W_j(y, y^*, \kappa_y)$$





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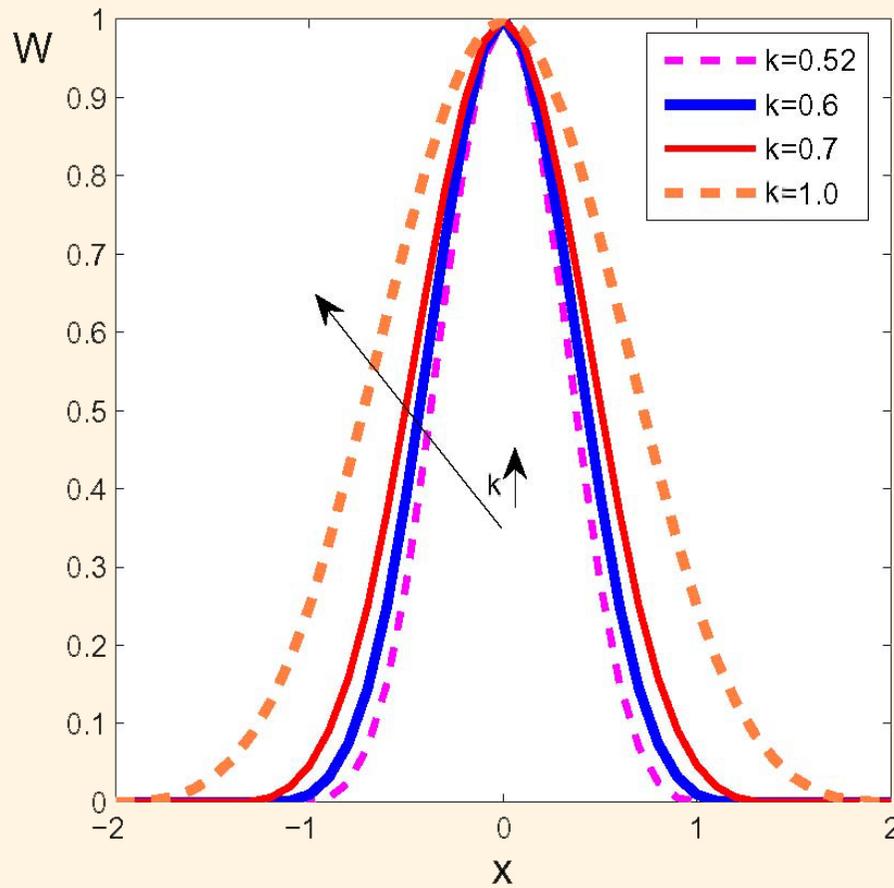
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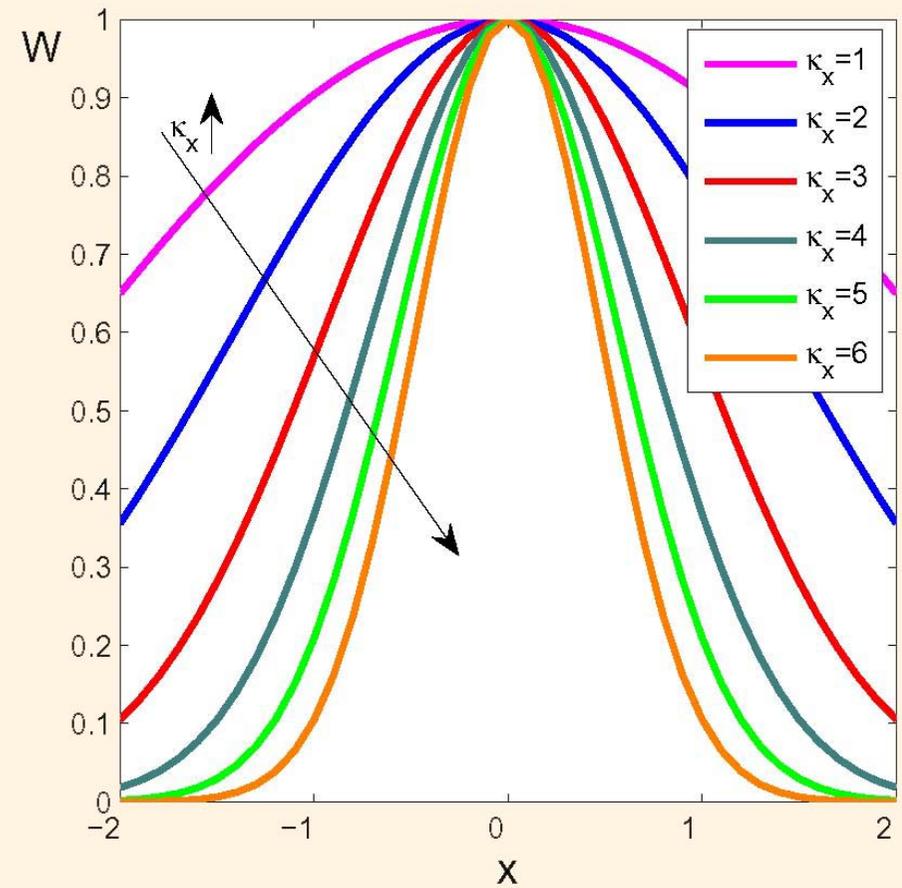


# The basis: Kernel approximations (VI)

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CUBIC SPLINE



EXPONENTIAL KERNEL





## Moving Least Squares (I)

- Reconstruction of  $u(\mathbf{x})$  at a point  $\mathbf{x}$  by using a weighted LS approximation in the vicinity of  $\mathbf{x}$ :

$$u(\mathbf{x}) \approx \hat{u}(\mathbf{x}) = \sum_{i=1}^m p_i(\mathbf{x}) \alpha_i(\mathbf{z}) \big|_{\mathbf{z}=\mathbf{x}} = \mathbf{p}^T(\mathbf{x}) \boldsymbol{\alpha}(\mathbf{z}) \big|_{\mathbf{z}=\mathbf{x}}$$

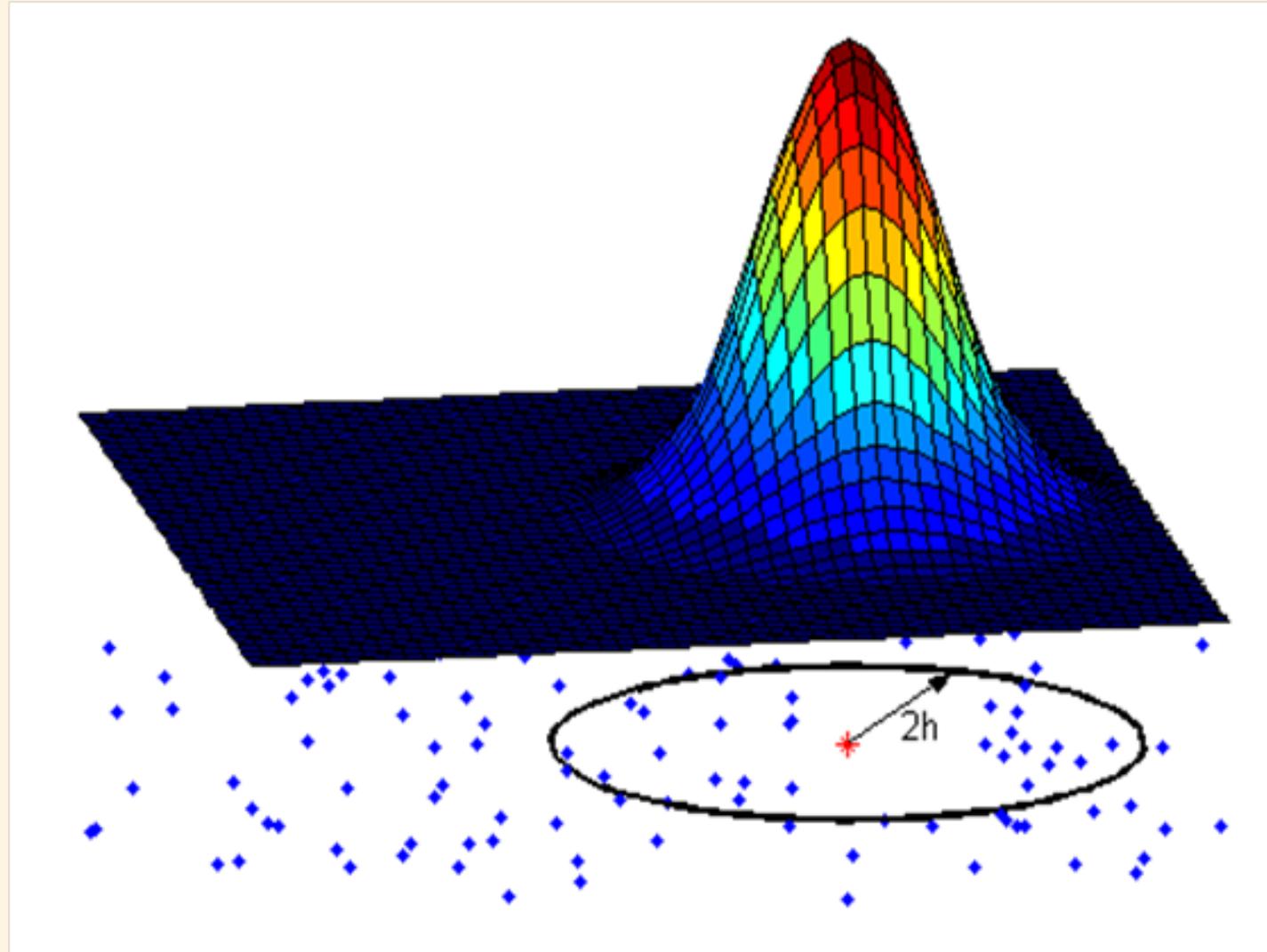
- $\mathbf{p}^T(\mathbf{x})$ : base of functions with dimension  $m$ .
- $\boldsymbol{\alpha}(\mathbf{z}) \big|_{\mathbf{z}=\mathbf{x}}$ : Parameters that minimize the error functional:

$$J(\boldsymbol{\alpha}(\mathbf{z}) \big|_{\mathbf{z}=\mathbf{x}}) = \int_{\mathbf{y} \in \Omega_{\mathbf{x}}} W(\mathbf{z} - \mathbf{y}, h) \big|_{\mathbf{z}=\mathbf{x}} [u(\mathbf{y}) - \mathbf{p}^T(\mathbf{x}) \boldsymbol{\alpha}(\mathbf{z}) \big|_{\mathbf{z}=\mathbf{x}}]^2 d\Omega_{\mathbf{x}}$$

- $W(\mathbf{z} - \mathbf{y}, h) \big|_{\mathbf{z}=\mathbf{x}}$ : **kernel** (smoothing function) with compact support ( $\Omega_{\mathbf{x}}$ ) centered in  $\mathbf{z} = \mathbf{x}$ .
- $h$ : smoothing length.



# Moving Least Squares (III)



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## Moving Least Squares (IV)

- ▶ Minimization of  $J$  leads to:

$$\int_{\mathbf{y} \in \Omega_{\mathbf{x}}} \mathbf{p}(\mathbf{y}) W(\mathbf{z} - \mathbf{y}, h) \Big|_{\mathbf{z}=\mathbf{x}} u(\mathbf{y}) d\Omega_{\mathbf{x}} = \mathbf{M}(\mathbf{x}) \boldsymbol{\alpha}(\mathbf{z}) \Big|_{\mathbf{z}=\mathbf{x}}$$

- ▶  $\mathbf{M}(\mathbf{x})$  is the *moment matrix* defined as:

$$\mathbf{M}(\mathbf{x}) = \int_{\mathbf{y} \in \Omega_{\mathbf{x}}} \mathbf{p}(\mathbf{y}) W(\mathbf{z} - \mathbf{y}, h) \Big|_{\mathbf{z}=\mathbf{x}} \mathbf{p}^T(\mathbf{y})$$





## Moving Least Squares (III)

- In practice,  $\Omega$  is a set of scattered points. Previous integrals are evaluated using points in  $\Omega_{\mathbf{x}}$  as quadrature points:

$$\alpha(z) \Big|_{z=\mathbf{x}} = M^{-1}(\mathbf{x}) P_{\Omega_{\mathbf{x}}} W(\mathbf{x}) u_{\Omega_{\mathbf{x}}}$$

- $u_{\Omega_{\mathbf{x}}}$  contains nodal values of the function  $u_{\mathbf{x}}$  to be approximated, at  $n_{\mathbf{x}}$  nodes in  $\Omega_{\mathbf{x}}$

$$u_{\Omega_{\mathbf{x}}} = (u(\mathbf{x}_1) \ u(\mathbf{x}_2) \ \cdots \ u(\mathbf{x}_{n_{\mathbf{x}}}))^T$$



## Moving Least Squares (IV)

- Discrete expression of the moment matrix is a  $m \times m$  matrix equals to  $\mathbf{M}(\mathbf{x}) = \mathbf{P}_{\Omega\mathbf{x}}\mathbf{W}(\mathbf{x})\mathbf{P}_{\Omega\mathbf{x}}^T$
- $\mathbf{P}_{\Omega\mathbf{x}}$  (dimension  $m \times n_{\mathbf{x}}$ ), and  $\mathbf{W}(\mathbf{x})$  (dimension  $n_{\mathbf{x}} \times n_{\mathbf{x}}$ ) are obtained by

$$\mathbf{P}_{\Omega\mathbf{x}} = (\mathbf{p}(\mathbf{x}_1) \ \mathbf{p}(\mathbf{x}_2) \ \cdots \ \mathbf{p}(\mathbf{x}_{n_{\mathbf{x}}}))$$

$$\mathbf{W}(\mathbf{x}) = \text{diag} \{W_i(\mathbf{x} - \mathbf{x}_i)\} \quad i = 1, \dots, n_{\mathbf{x}} \quad (1)$$

- Finally, MLS approximation is written by:

$$\hat{u}(\mathbf{x}) = \mathbf{p}^T(\mathbf{x})\mathbf{M}^{-1}(\mathbf{x})\mathbf{P}_{\Omega\mathbf{x}}\mathbf{W}(\mathbf{x})\mathbf{u}_{\Omega\mathbf{x}} = \mathbf{N}^T(\mathbf{x})\mathbf{u}_{\Omega\mathbf{x}} = \sum_{j=1}^{n_{\mathbf{x}}} N_j(\mathbf{x})u_j$$



## Moving Least Squares (V)

- ▶ Interpolation can be written as:

$$\hat{u}(\mathbf{x}) = \sum_{j=1}^{n_{\mathbf{x}}} N_j(\mathbf{x}) u_j$$

with

$$\mathbf{N}^T(\mathbf{x}) = \mathbf{p}^T(\mathbf{x}) \mathbf{M}^{-1}(\mathbf{x}) \mathbf{P}_{\Omega_{\mathbf{x}}} \mathbf{W}(\mathbf{x})$$

- ▶  $N_j$  can be considered as “shape functions”.
- ▶  $N_j$  depends on the number of neighbors, the kernel and the base ( $\mathbf{p}^T$ ).
- ▶  $N_j$  is a function of the grid.





## Moving Least Squares (VI)

- ▶ A practical note about the polynomial basis

$$\mathbf{p}(\mathbf{x}) = (1 \quad x \quad y \quad xy \quad x^2 \quad y^2)^T$$

- We define locally and scale the monomials of the basis
- Better conditioning of the momentum matrix
- If MLS shape functions  $\mathbf{N}(\mathbf{x})$  are evaluated at a point  $\mathbf{x}_I$ , the basis is evaluated at  $\frac{\mathbf{x} - \mathbf{x}_I}{h}$
- Then we can write:

$$\mathbf{N}^T(\mathbf{x}_I) = \mathbf{p}^T(\mathbf{0})\mathbf{M}^{-1}(\mathbf{x}_I)\mathbf{P}_{\Omega\mathbf{x}_I}\mathbf{W}(\mathbf{x}_I) = \mathbf{p}^T(\mathbf{0})\mathbf{C}(\mathbf{x}_I)$$

with

$$\mathbf{C}(\mathbf{x}_I) = \mathbf{M}^{-1}(\mathbf{x}_I)\mathbf{P}_{\Omega\mathbf{x}_I}\mathbf{W}(\mathbf{x}_I)$$





## Moving Least Squares (VII)

### ► Computation of derivatives

- First derivatives

$$\frac{\partial \mathbf{N}^T(\mathbf{x})}{\partial x} = \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial x} \mathbf{C}(\mathbf{x}) + \mathbf{p}^T(\mathbf{x}) \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x}$$

- Second derivatives

$$\frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x^2} = \frac{\partial^2 \mathbf{p}^T(\mathbf{x})}{\partial x^2} \mathbf{C}(\mathbf{x}) + 2 \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial x} \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x} + \mathbf{p}(\mathbf{x}) \frac{\partial^2 \mathbf{C}(\mathbf{x})}{\partial x^2}$$

$$\frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x \partial y} = \frac{\partial^2 \mathbf{p}^T(\mathbf{x})}{\partial x \partial y} \mathbf{C}(\mathbf{x}) + \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial x} \frac{\partial \mathbf{C}(\mathbf{x})}{\partial y} + \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial y} \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x}$$

$$+ \mathbf{p}^T(\mathbf{x}) \frac{\partial^2 \mathbf{C}(\mathbf{x})}{\partial x \partial y}$$



## Moving Least Squares (VII)

- where  $\frac{\partial \mathbf{C}(\mathbf{x})}{\partial x}$  is given by

$$\frac{\partial \mathbf{C}(\mathbf{x})}{\partial x} = \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial \mathbf{W}(\mathbf{x})}{\partial x} (\mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x}))$$

- and the second derivatives of  $\mathbf{C}(\mathbf{x})$

$$\begin{aligned} \frac{\partial^2 \mathbf{C}(\mathbf{x})}{\partial x^2} = & \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x} \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial \mathbf{W}}{\partial x} (\mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x})) \\ & + \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial^2 \mathbf{W}(\mathbf{x})}{\partial x^2} (\mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x})) \\ & - \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial \mathbf{W}(\mathbf{x})}{\partial x} \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial \mathbf{W}(\mathbf{x})}{\partial x} (\mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x})) \\ & - \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial \mathbf{W}(\mathbf{x})}{\partial x} \mathbf{p}^T(\mathbf{x}) \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x} \end{aligned}$$



# Moving Least Squares (VII)

$$\begin{aligned}\frac{\partial^2 \mathbf{C}(\mathbf{x})}{\partial x \partial y} &= \frac{\partial \mathbf{C}(\mathbf{x})}{\partial y} \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial W(\mathbf{x})}{\partial x} \left( \mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x}) \right) \\ &+ \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial^2 W(\mathbf{x})}{\partial x \partial y} \left( \mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x}) \right) \\ &- \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial W(\mathbf{x})}{\partial y} \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial W(\mathbf{x})}{\partial x} \left( \mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x}) \right) \\ &- \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial W(\mathbf{x})}{\partial x} \mathbf{p}^T(\mathbf{x}) \frac{\partial \mathbf{C}(\mathbf{x})}{\partial y}\end{aligned}$$



## Moving Least Squares (VII)

### ► Computation of derivatives

- The *diffuse* derivatives are obtained by neglecting all derivatives of  $\mathbf{C}(\mathbf{x})$

$$\frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x^2} \approx \frac{\partial^2 \mathbf{p}^T(\mathbf{x})}{\partial x^2} \mathbf{C}(\mathbf{x})$$

$$\frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x \partial y} \approx \frac{\partial^2 \mathbf{p}^T(\mathbf{x})}{\partial x \partial y} \mathbf{C}(\mathbf{x})$$

$$\frac{\partial^n \mathbf{N}^T(\mathbf{x})}{\partial x^n} \approx \frac{\partial^n \mathbf{p}^T(\mathbf{x})}{\partial x^n} \mathbf{C}(\mathbf{x})$$



Huerta et al., *Pseudo-divergence-free Element Free Galerkin method for incompressible fluid flow*, CMAME, 2004





## Moving Least Squares (VII)

- However, as we already have computed the first derivatives of  $\mathbf{C}(\mathbf{x})$ , it is possible to use a *semi-diffuse* approach without extra effort:

$$\frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x^2} \approx \frac{\partial^2 \mathbf{p}^T(\mathbf{x})}{\partial x^2} \mathbf{C}(\mathbf{x}) + 2 \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial x} \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x}$$

$$\frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x \partial y} \approx \frac{\partial^2 \mathbf{p}^T(\mathbf{x})}{\partial x \partial y} \mathbf{C}(\mathbf{x}) + \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial x} \frac{\partial \mathbf{C}(\mathbf{x})}{\partial y} + \frac{\partial \mathbf{p}^T(\mathbf{x})}{\partial y} \frac{\partial \mathbf{C}(\mathbf{x})}{\partial x}$$

Derivatives	$L_1$ Error	$L_2$ Error
Diffuse	$1.631 \times 10^{-5}$	$4.784 \times 10^{-5}$
Semi-Diffuse	$1.586 \times 10^{-5}$	$4.710 \times 10^{-5}$
Full	$1.288 \times 10^{-5}$	$3.656 \times 10^{-5}$

- ▷ It has been proved that use of diffuse or semi-diffuse derivatives does not decrease the *order of accuracy*.
- ▷ It should be noted that the *accuracy* is affected

*Accuracy assesment of a high-order moving least squares finite volume method for compressible flows, C&F, 2013*



## The FV-MLS method (I)

- Note that due to the local scaling  $\frac{\mathbf{x} - \mathbf{x}_I}{h}$

$$\mathbf{p}^T(\mathbf{0}) = (1 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0)$$

$$\frac{\partial \mathbf{p}^T(\mathbf{0})}{\partial x} = \left( 0 \quad \frac{1}{h} \quad 0 \quad 0 \quad 0 \quad 0 \right)$$

$$\frac{\partial \mathbf{p}^T(\mathbf{0})}{\partial y} = \left( 0 \quad 0 \quad \frac{1}{h} \quad 0 \quad 0 \quad 0 \right)$$



## The FV-MLS method

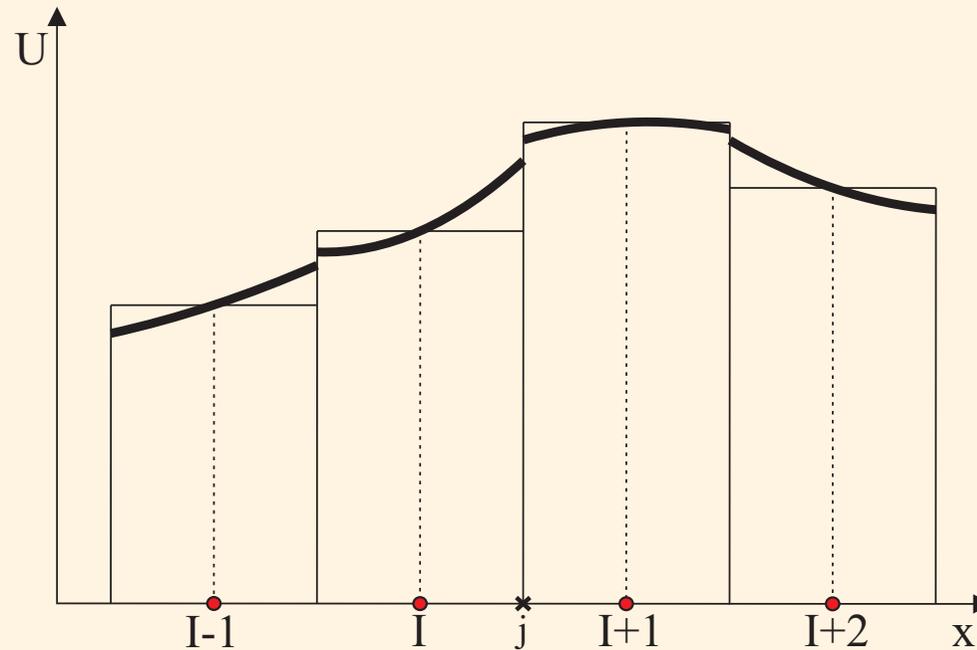
- ▶ This scheme acknowledges the **different nature** of convective and diffusive terms.
- ▶ We start from a **high-order, continuous** MLS approximation of the solution:
- ▶ **Convective** terms discretization:
  - **Breaks** the continuous representation of the MLS approximation.
  - Obtains a continuous representation of the variables **inside each cell**.
- ▶ **Diffusive** terms discretization is:
  - Centered → **Direct interpolation** at Gauss points with MLS.
  - Continuous.
  - Highly accurate.





## The FV-MLS method

- Hyperbolic term → Flux Difference Splitting



$$U_L = U_I + \nabla U_I \cdot (\mathbf{x}_j - \mathbf{x}_I) + \frac{1}{2} (\mathbf{x}_j - \mathbf{x}_I)^T \mathbf{H}_I (\mathbf{x}_j - \mathbf{x}_I)$$

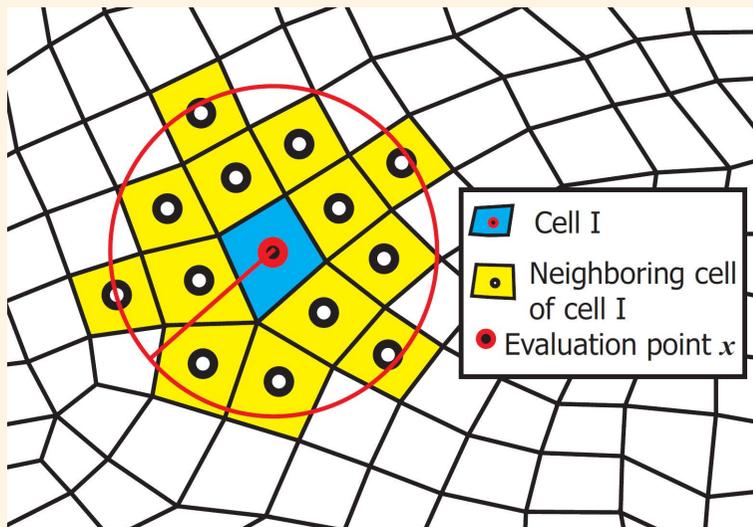
$$U_R = U_{I+1} + \nabla U_{I+1} \cdot (\mathbf{x}_j - \mathbf{x}_{I+1}) + \frac{1}{2} (\mathbf{x}_j - \mathbf{x}_{I+1})^T \mathbf{H}_{I+1} (\mathbf{x}_j - \mathbf{x}_{I+1})$$



# The FV-MLS method

## ► Hyperbolic-like terms:

- MLS is used to **compute the gradients** and high-order derivatives required for the reconstruction of the variable at integration points placed at interface.



$$\nabla U_I = \sum_{j=1}^{n_x} U_j \nabla N_j(\mathbf{x}_I)$$

$$\mathbf{C}(\mathbf{x}_I) = \mathbf{M}^{-1}(\mathbf{x}_I) \mathbf{P}_{\Omega \mathbf{x}_I} \mathbf{W}(\mathbf{x}_I)$$

$$\frac{\partial \mathbf{C}(\mathbf{x})}{\partial x} = \mathbf{C}(\mathbf{x}) \mathbf{W}^{-1}(\mathbf{x}) \frac{\partial \mathbf{W}(\mathbf{x})}{\partial x} (\mathbf{I} - \mathbf{p}^T(\mathbf{x}) \mathbf{C}(\mathbf{x}))$$

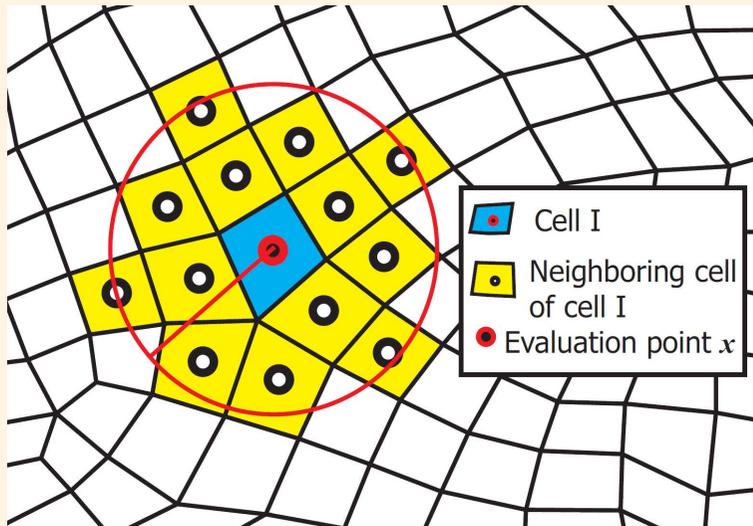
$$\left. \frac{\partial \mathbf{N}^T(\mathbf{x})}{\partial x} \right|_{x=x_I} = \frac{\partial \mathbf{p}^T(\mathbf{0})}{\partial x} \mathbf{C}(\mathbf{x}_I) + \mathbf{p}^T(\mathbf{0}) \frac{\partial \mathbf{C}(\mathbf{x}_I)}{\partial x}$$



# The FV-MLS method

## ► Hyperbolic-like terms:

- MLS is used to **compute the gradients** and high-order derivatives required for the reconstruction of the variable at integration points placed at interface.



$$\frac{\partial^2 \mathbf{U}_I}{\partial x^2} = \sum_{j=1}^{n_x} \mathbf{U}_j \frac{\partial^2 N_j(\mathbf{x}_I)}{\partial x^2}$$

$$\left. \frac{\partial^2 \mathbf{N}^T(\mathbf{x})}{\partial x^2} \right|_{x=x_I} \approx \frac{\partial^2 \mathbf{p}^T(\mathbf{0})}{\partial x^2} \mathbf{C}(\mathbf{x}_I)$$

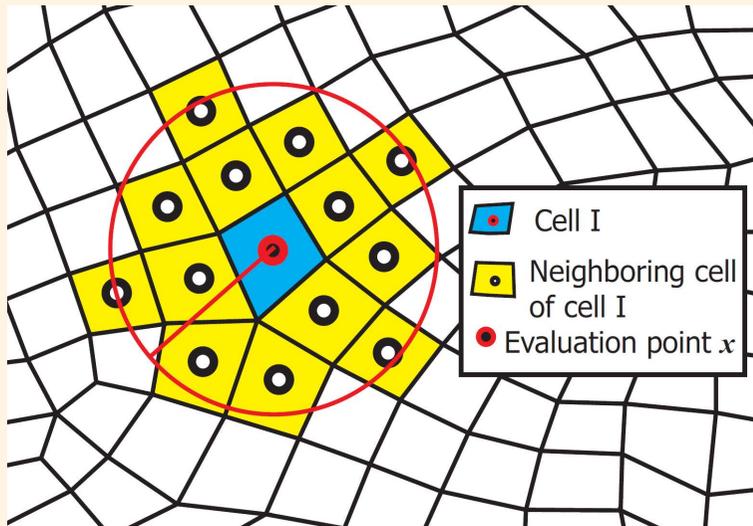
$$\mathbf{C}(\mathbf{x}_I) = \mathbf{M}^{-1}(\mathbf{x}_I) \mathbf{P}_{\Omega \mathbf{x}_I} \mathbf{W}(\mathbf{x}_I)$$



# The FV-MLS method

## ► Hyperbolic-like terms:

- MLS is used to **compute the gradients** and high-order derivatives required for the reconstruction of the variable at integration points placed at interface.



$$\frac{\partial^n \mathbf{U}_I}{\partial x^n} = \sum_{j=1}^{n_x} \mathbf{U}_j \frac{\partial^n N_j(\mathbf{x}_I)}{\partial x^n}$$

$$\left. \frac{\partial^n \mathbf{N}^T(\mathbf{x})}{\partial x^n} \right|_{x=x_I} \approx \frac{\partial^n \mathbf{p}^T(\mathbf{0})}{\partial x^n} \mathbf{C}(\mathbf{x}_I)$$

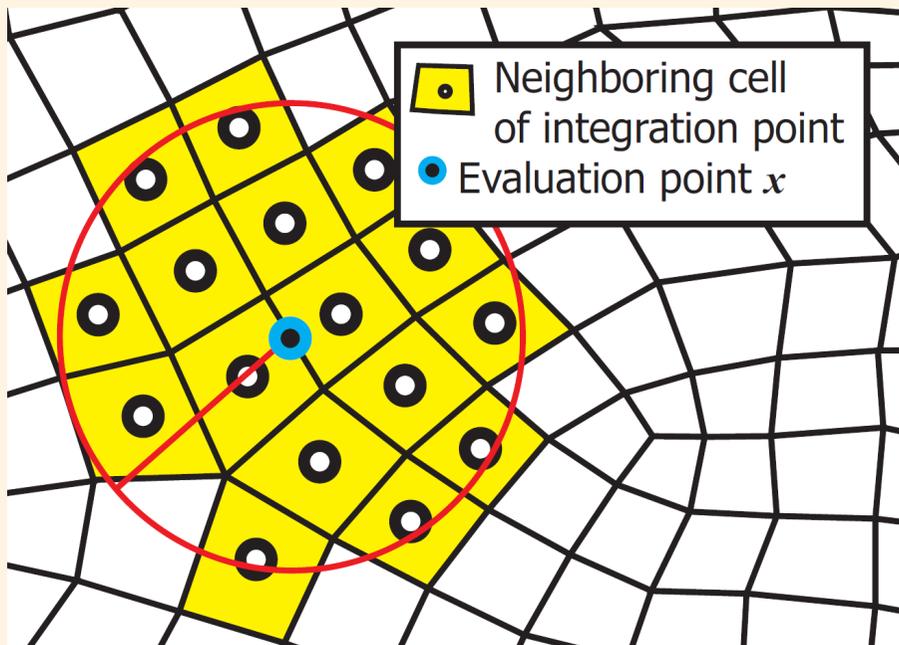
$$\mathbf{C}(\mathbf{x}_I) = \mathbf{M}^{-1}(\mathbf{x}_I) \mathbf{P}_{\Omega \mathbf{x}_I} \mathbf{W}(\mathbf{x}_I)$$



# The FV-MLS method

## ► Elliptic-like terms:

- Direct interpolation at Gauss points with MLS.



$$U_{iq} = \sum_{j=1}^{n_{iq}} U_j N_j(\mathbf{x}_{iq})$$

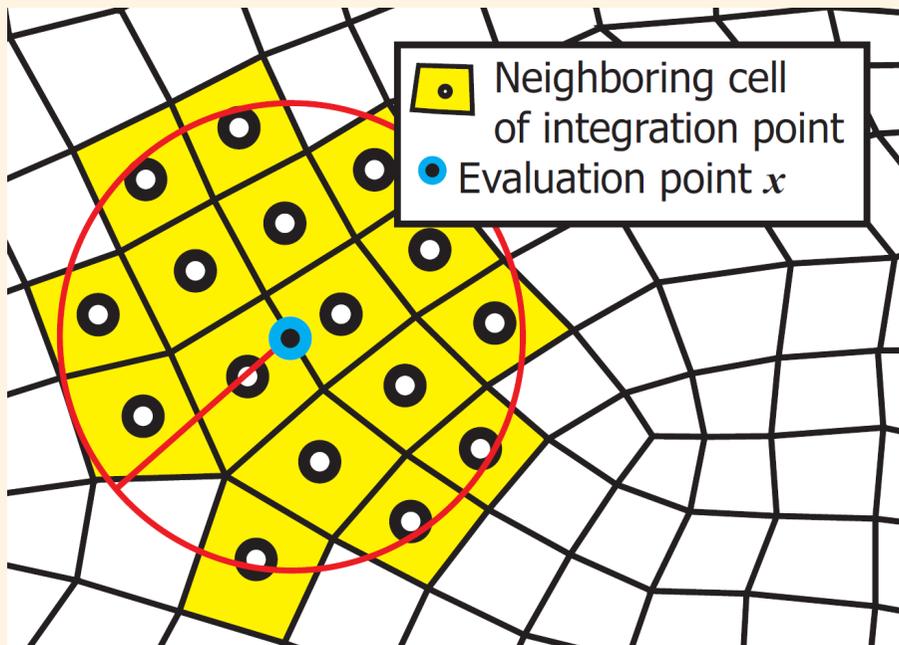
$$N^T(\mathbf{x}_{iq}) = \mathbf{p}^T(\mathbf{0}) \mathbf{C}(\mathbf{x}_{iq})$$



# The FV-MLS method

## ► Elliptic-like terms:

- Direct interpolation at Gauss points with MLS.



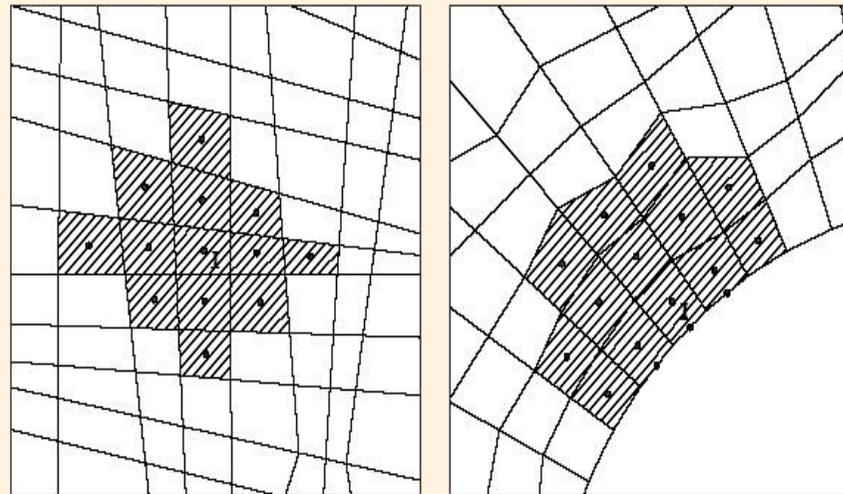
$$\nabla U_{iq} = \sum_{j=1}^{n_{iq}} U_j \nabla N_j(\mathbf{x}_{iq})$$

$$\left. \frac{\partial \mathbf{N}^T(\mathbf{x})}{\partial x} \right|_{\mathbf{x}=\mathbf{x}_{iq}} = \frac{\partial \mathbf{p}^T(\mathbf{0})}{\partial x} \mathbf{C}(\mathbf{x}_{iq}) + \mathbf{p}^T(\mathbf{0}) \frac{\partial \mathbf{C}(\mathbf{x}_{iq})}{\partial x}$$



## The FV-MLS method (V)

- ▶ Vertices and/or centroids of the control cells are the “particles” to perform the MLS approximation.
- ▶ We need to define **stencils** to “mark” the neighbor particles that define the cloud of points.





## The FV-MLS method (VI)

- ▶ How to define stencils?
- ▶ There exists an optimal size  $n_{xI}$  of points in the stencil such as  $N_{min} < n_{xI}$

$$N_{min} = \frac{(d + order)!}{d!order!}$$

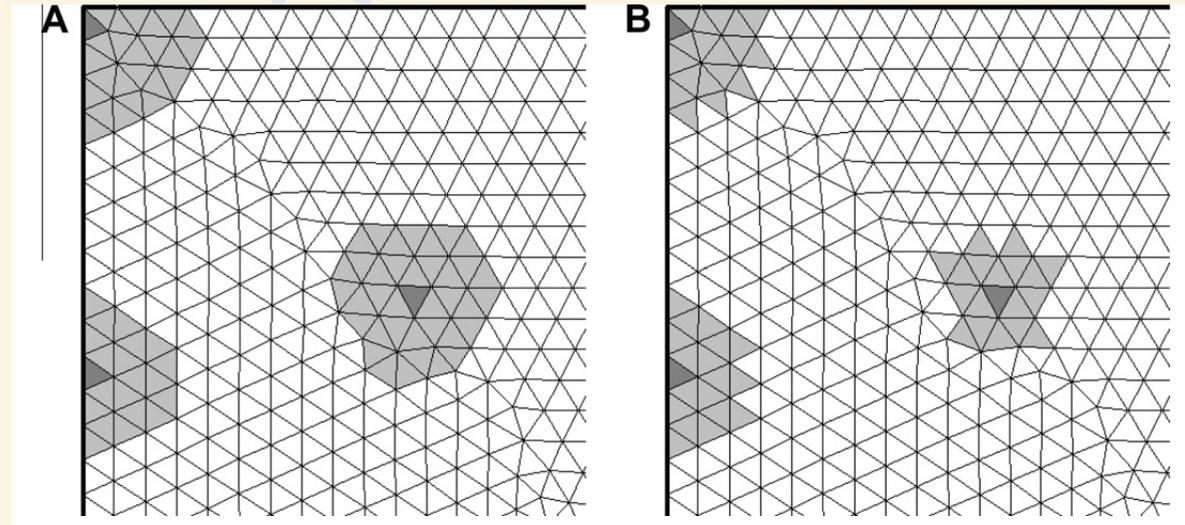
- ▶ If it is large  $\Rightarrow$  excessive dissipation
- ▶ Maybe optimization??





## The FV-MLS method (VII)

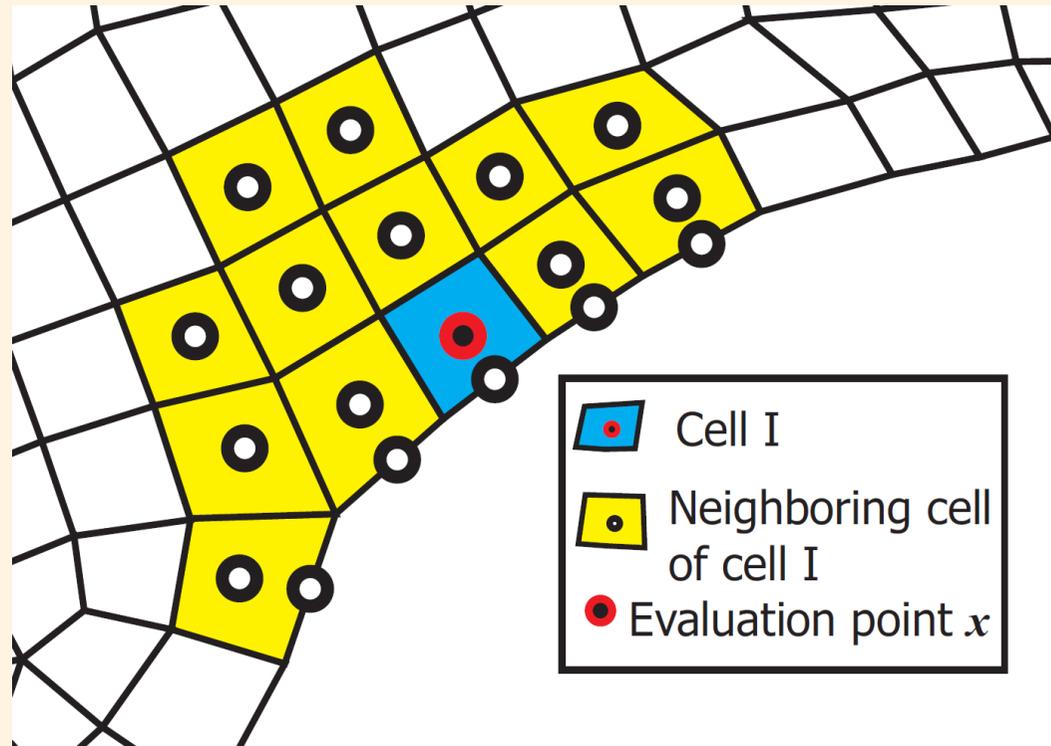
- ▶ We want stencils as compact as possible by using layers of cells around the active cell
- ▶ In practice this requires a high number of points in the stencil
- ▶ To overcome this inconvenient, last particles are placed such as satisfying a barycentric equilibrium





## The FV-MLS method (VII)

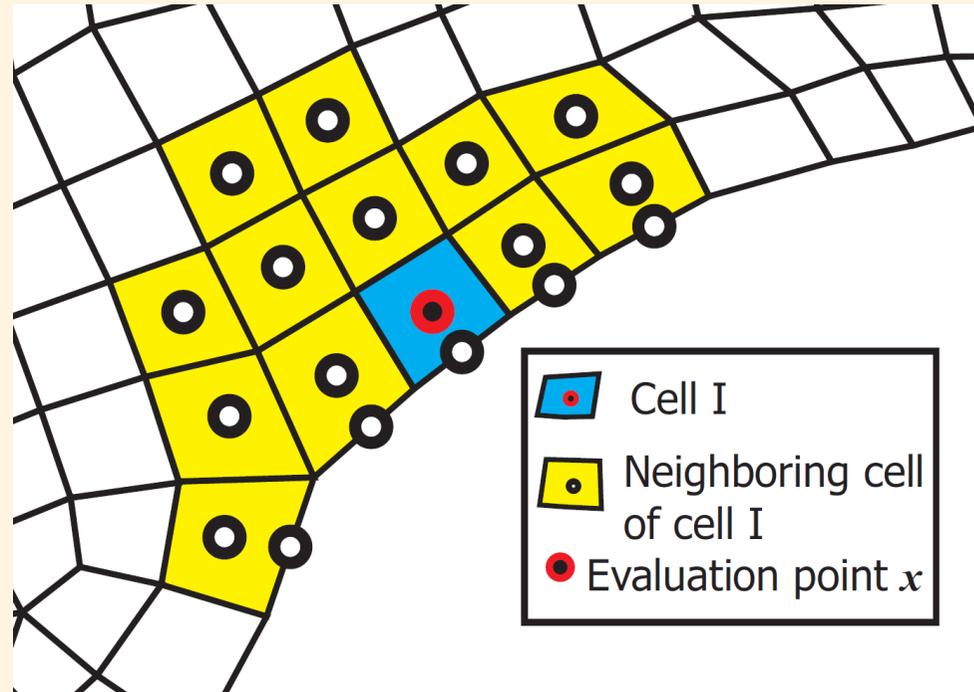
- ▶ Boundary conditions: We impose them on the numerical fluxes





## The FV-MLS method (VII)

- ▶ However, in order to improve the reconstruction we include ghost cells in the stencil

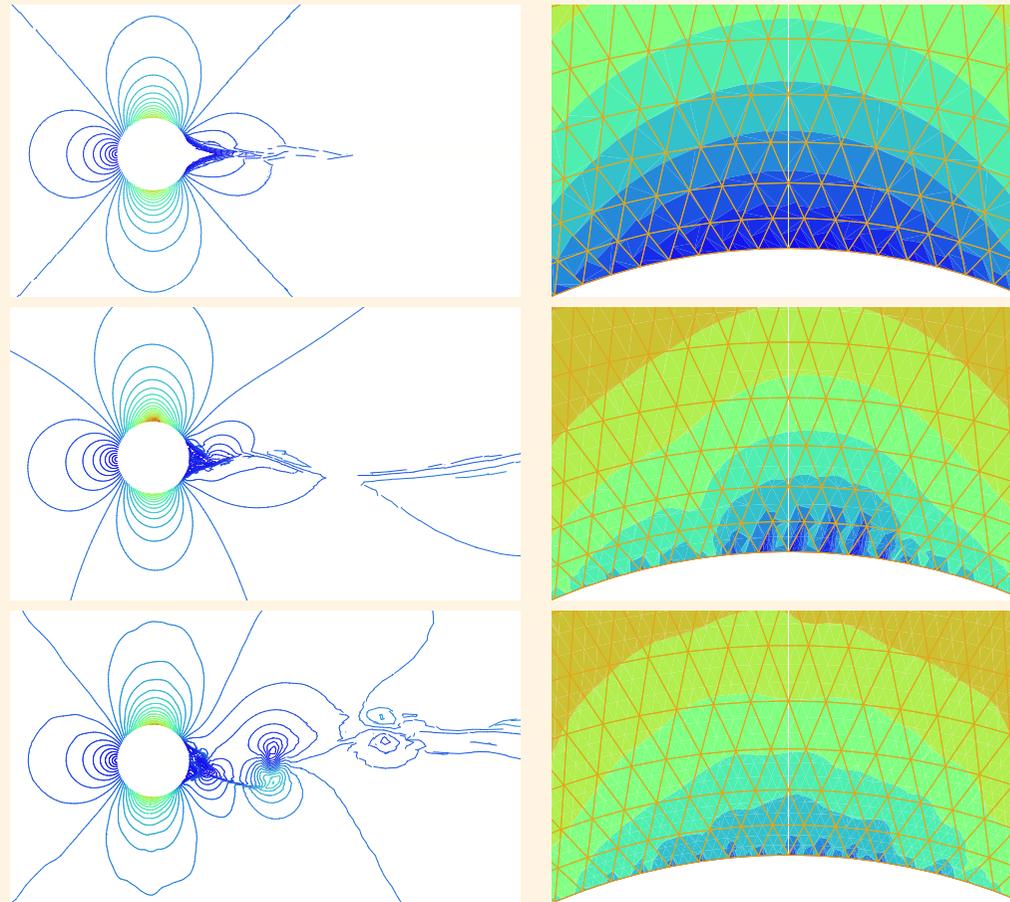


$$\nabla U_I = \sum_{j=1}^{n_x} U_j \nabla N_j(\mathbf{x}_I)$$



# A note on curved boundaries

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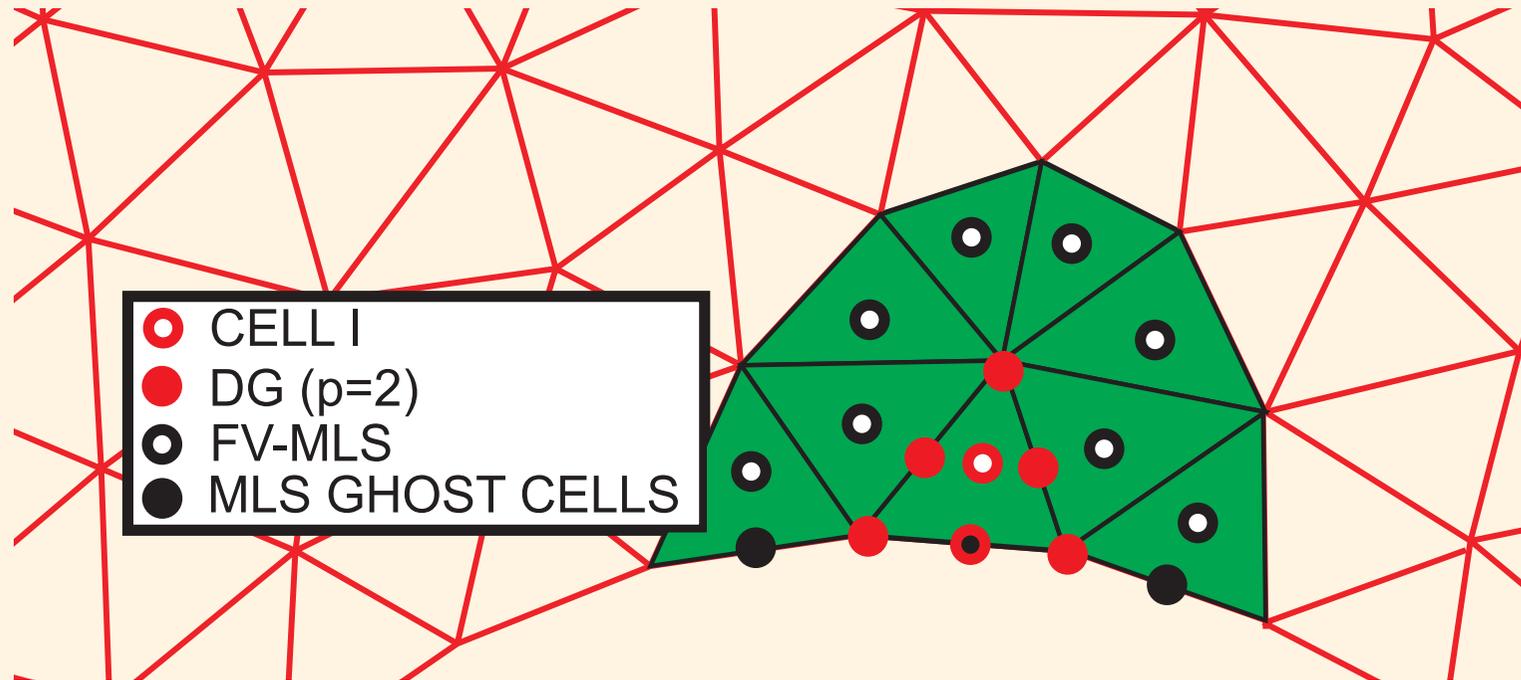


Mach isolines (left) and density at the top of the cylinder (right) with reflecting boundary conditions.  $p = 1, 2, 3$ , from top to bottom. Taken from Krivodonova and Berger, High-Order Accurate Implementation of Solid Wall Boundary Conditions in Curved Geometries, JCP, 2006





# A note on curved boundaries

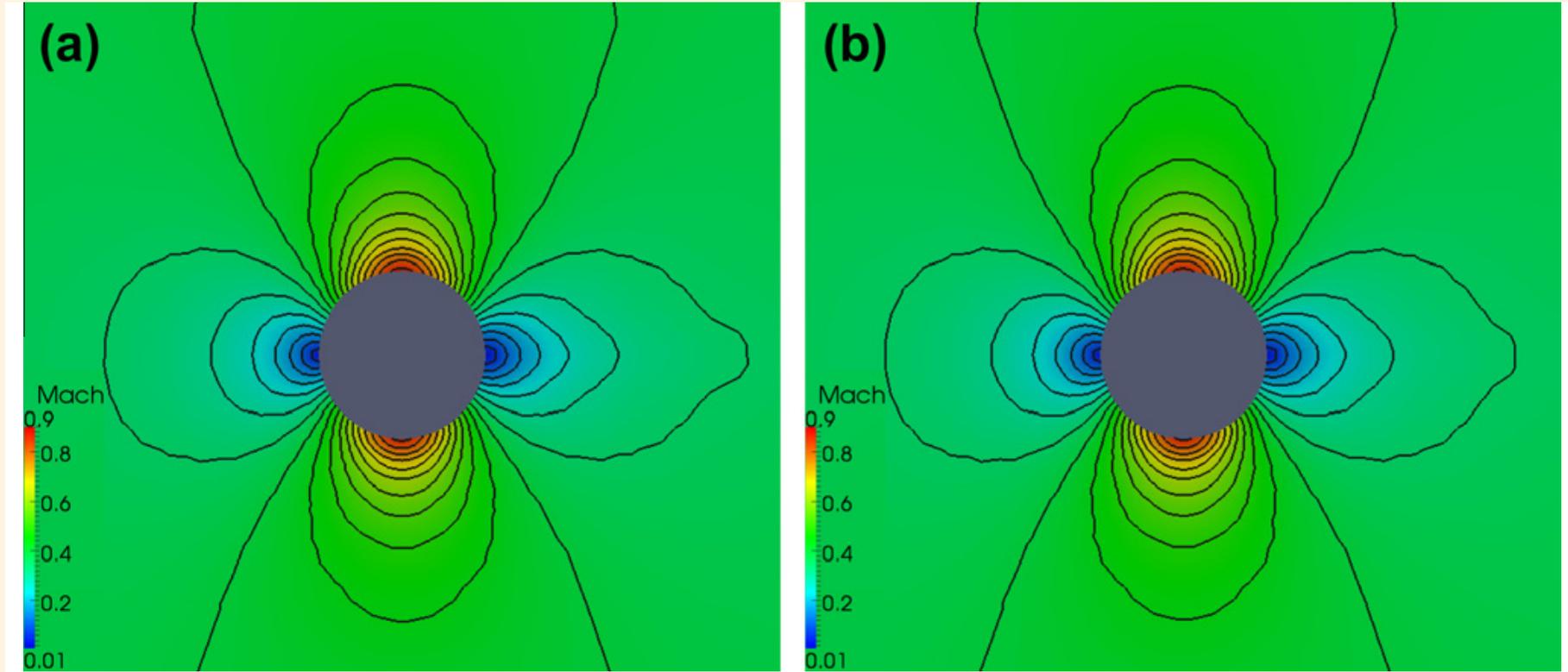


Schematic representation of the differences on curved boundary discretization between FV-MLS and DG. Shaded cells represent the MLS stencil.



# A note on curved boundaries

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Third-order accurate FV-MLS computation on a  $64 \times 16$  grid using boundary normal evaluations based on a straight representation (a) or on a physical representation (b) for curved geometry.

*Accuracy assesment of a high-order moving least squares finite volume method for compressible flows, C&F, 2013*





## Properties of the FV-MLS method

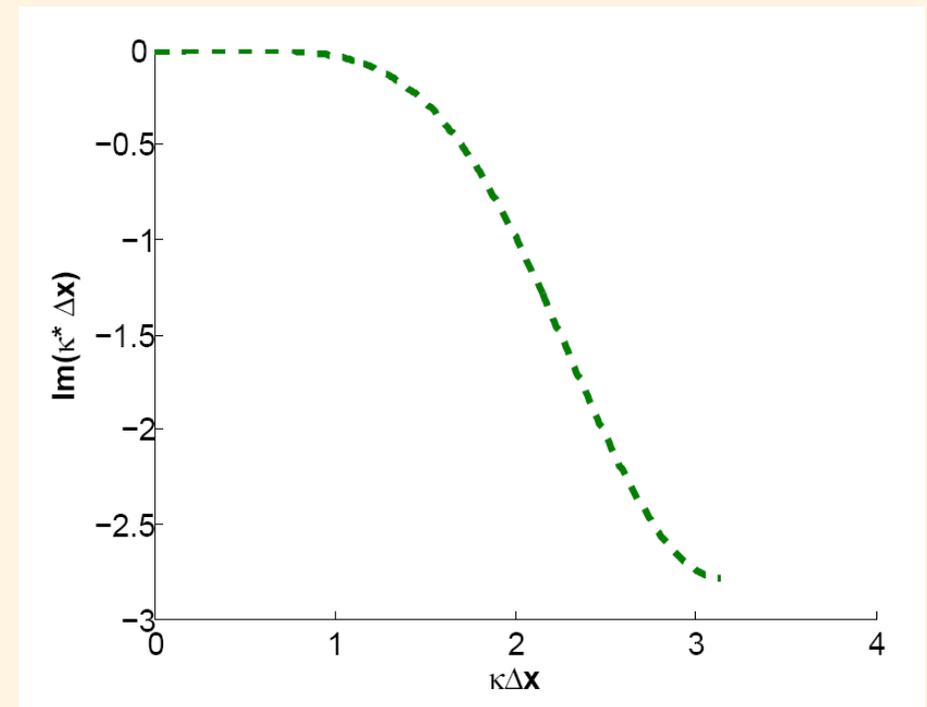
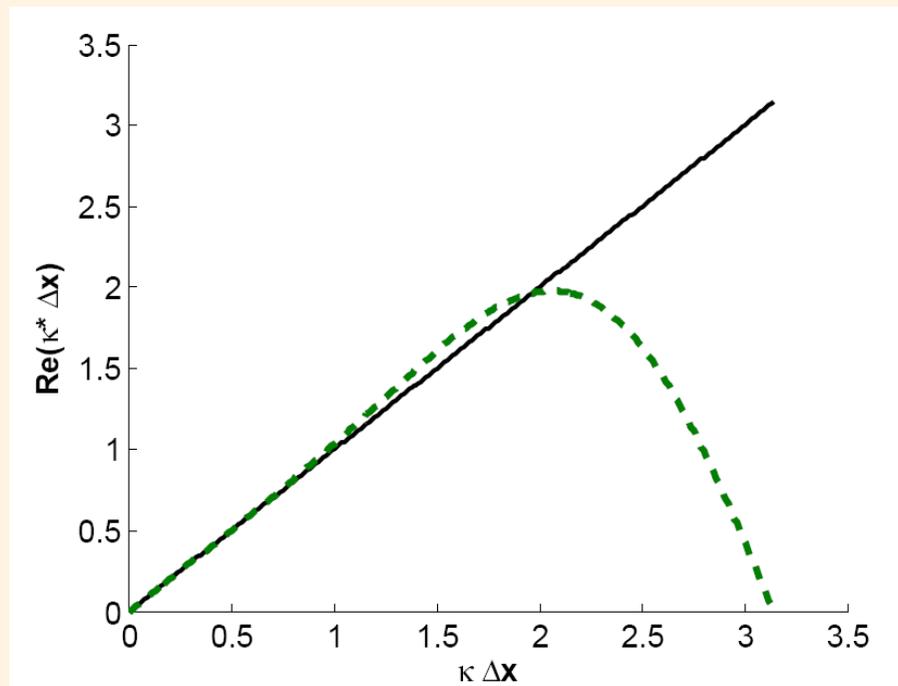
- ▶ We perform a **Fourier Analysis** for the 1D linear advection equation.
- ▶ We obtain the **dispersion-dissipation** properties.
- ▶ We compare **MLS interpolation** with Piecewise Polynomial Interpolation.
- ▶ We check the **order of convergence**.





## Dispersion and dissipation (I)

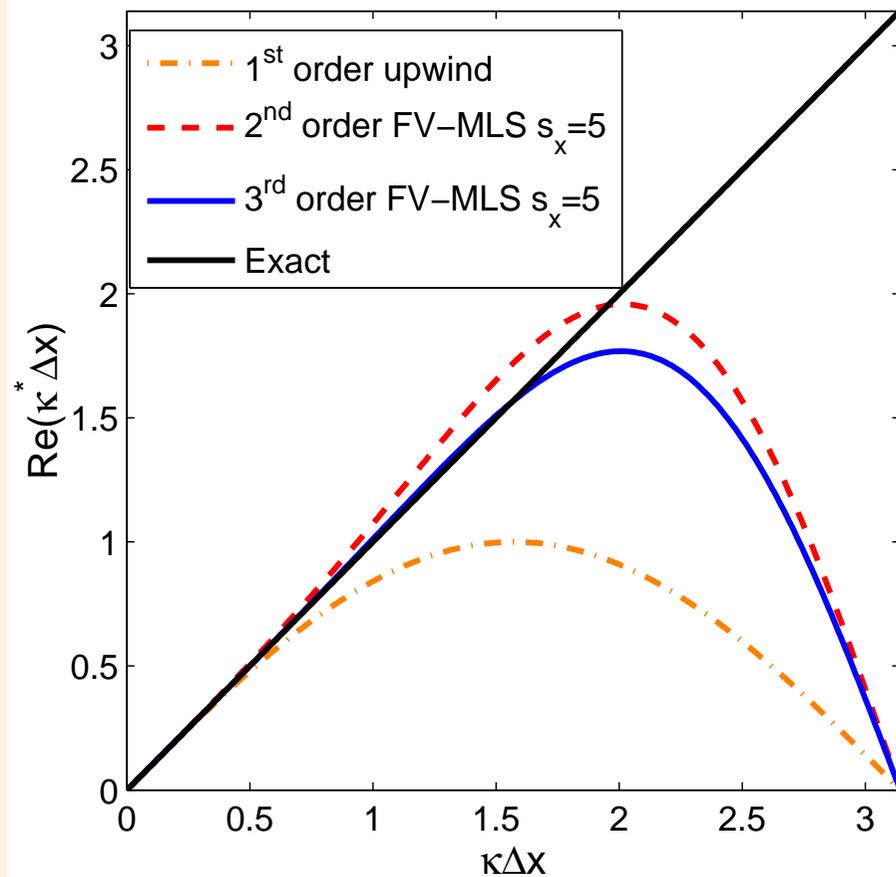
- ▶ Dispersion error: Associated with the error in the speed of the wave propagation
- ▶ Dissipation error: Associated with the error in the wave amplitude



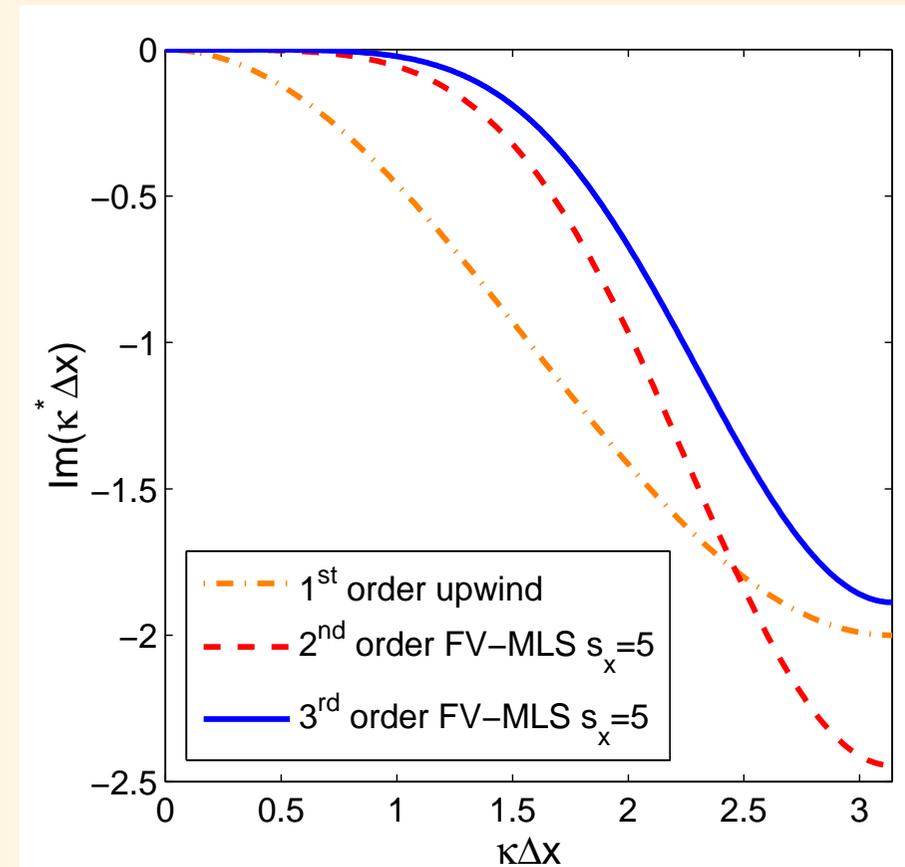


# Dispersion and dissipation

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DISPERSION



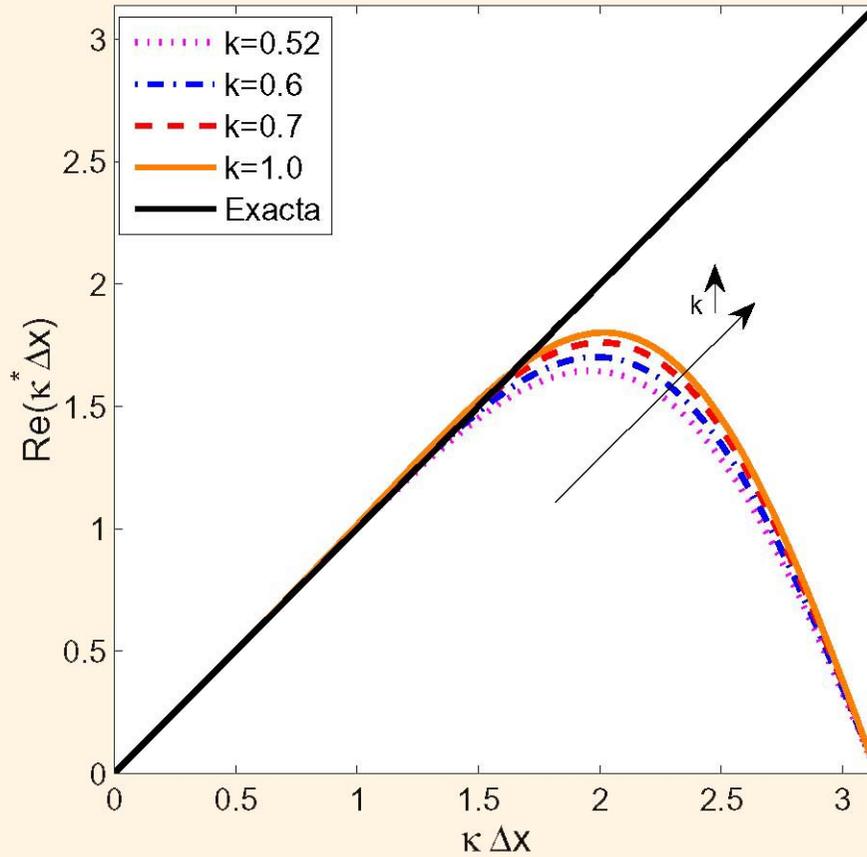
DISSIPATION



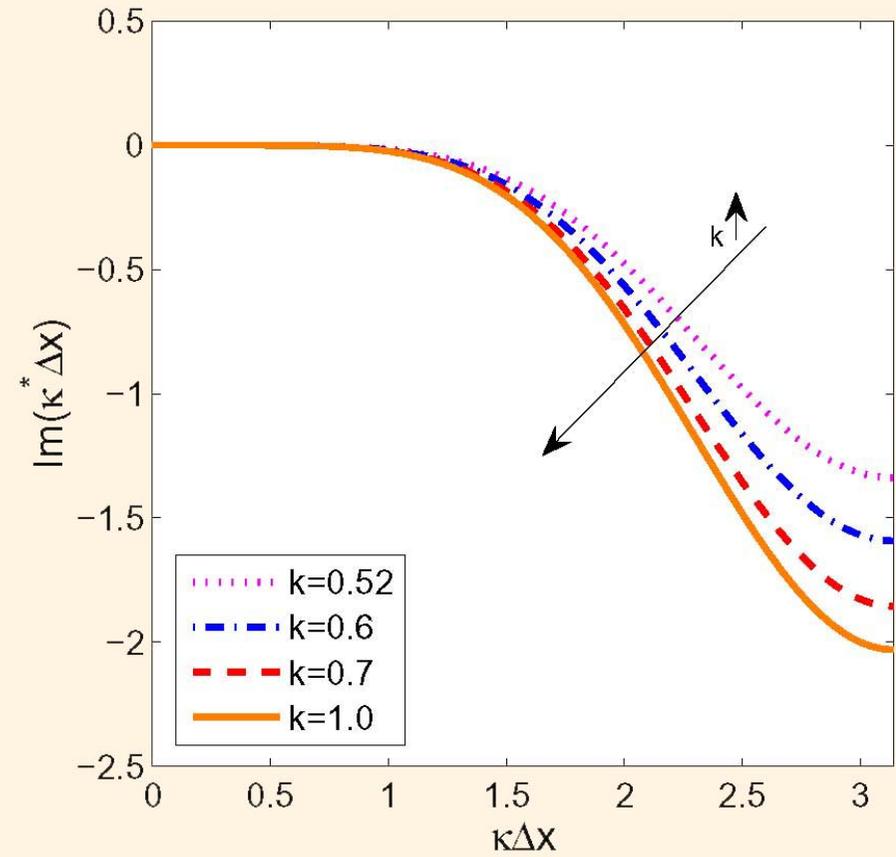


# Dispersion-Dissipation Properties. Cubic spline kernel

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DISPERSION



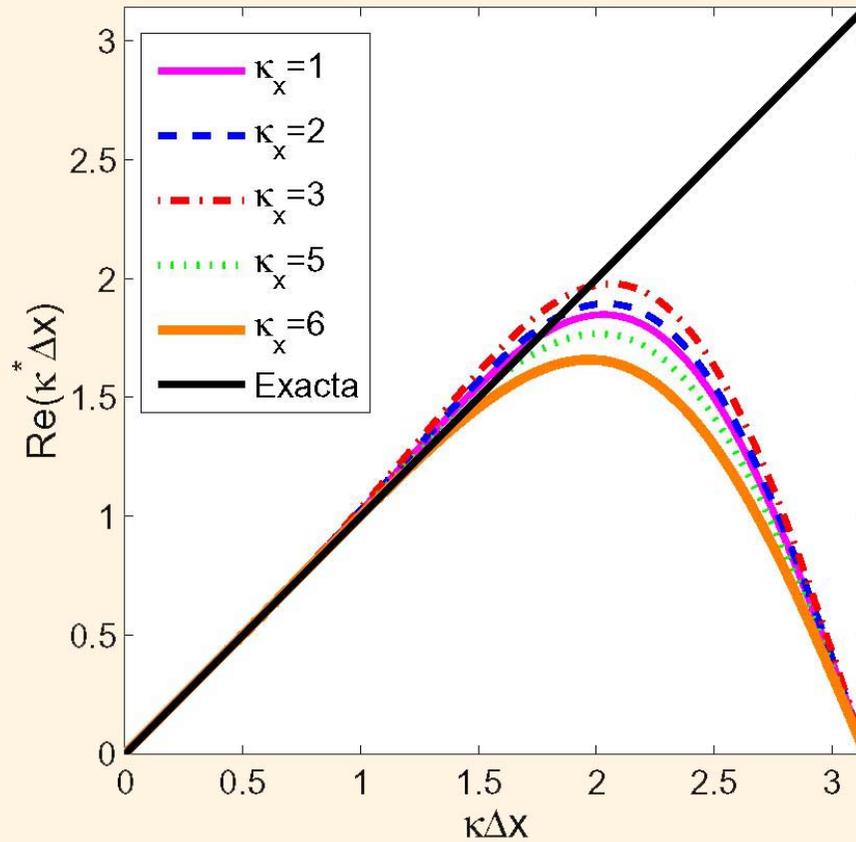
DISSIPATION



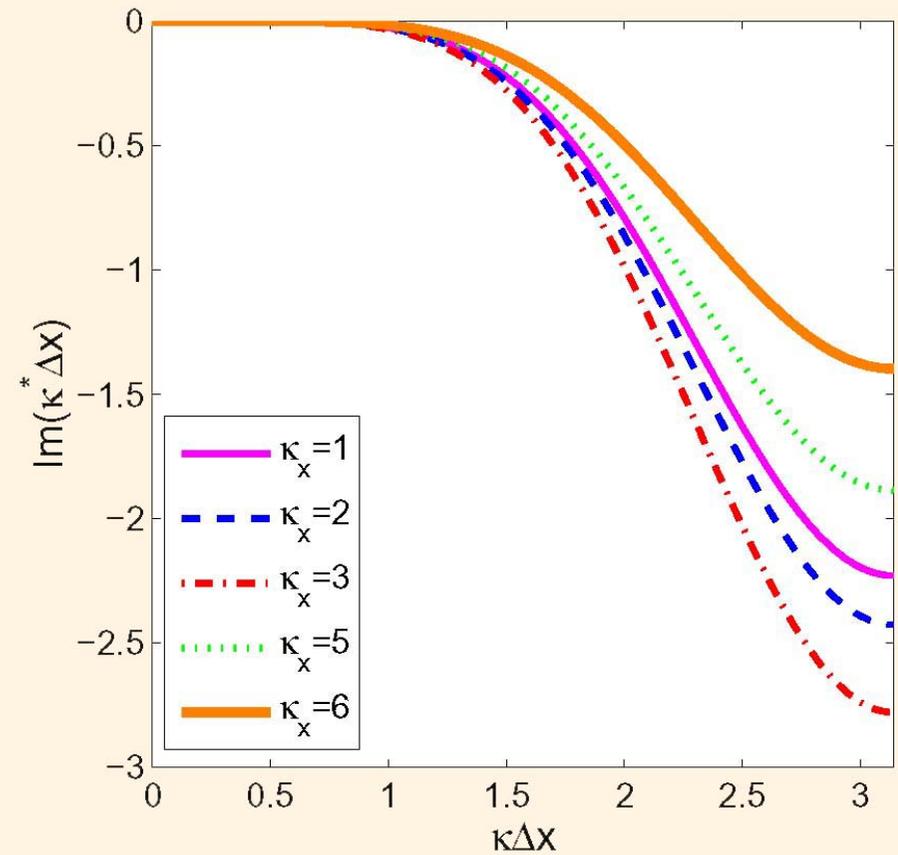


# Dispersion-Dissipation Properties. Exponential kernel

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DISPERSION



DISSIPATION

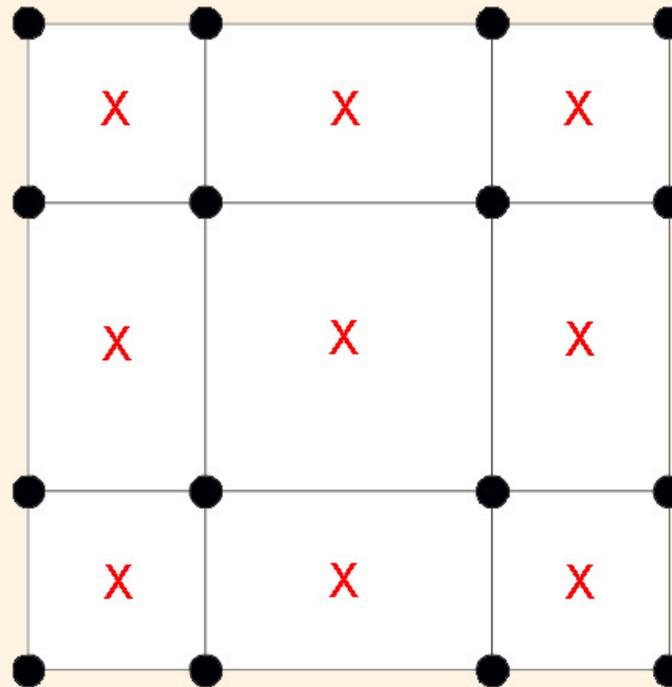




# IS MLS INTERPOLATION ACCURATE?

## A COMPARISON BETWEEN Piecewise Polynomial Interpolation (PPI) AND FV-MLS

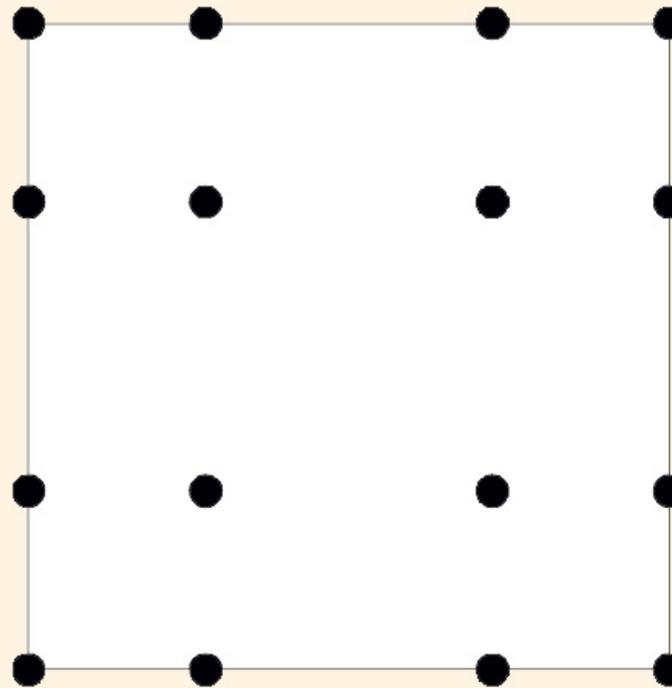
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## IS MLS INTERPOLATION ACCURATE?

- ▶ We compare interpolation with equivalent spatial resolution by using Moving Least Squares (MLS) (cubic basis) and Piecewise Polynomial Interpolation (PPI) ( $p = 3$ ).

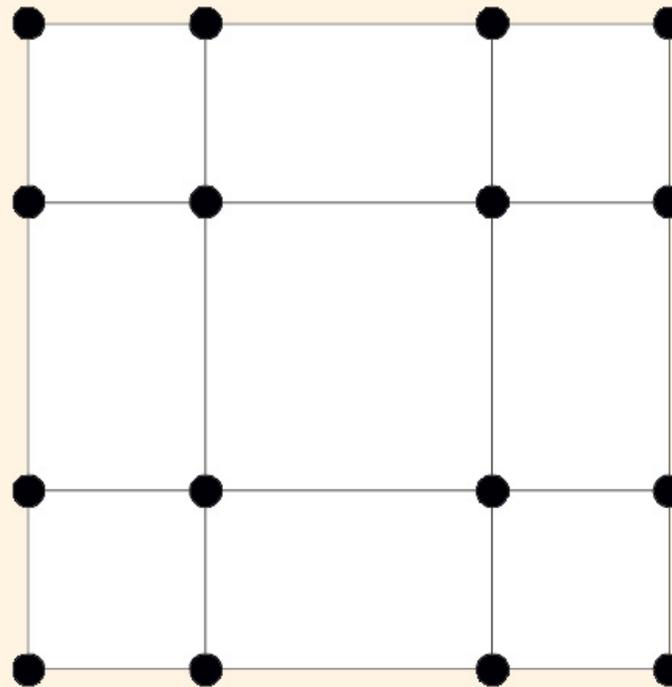


Division of a  $p = 3$  element to obtain a FV-MLS grid with equivalent spatial resolution.



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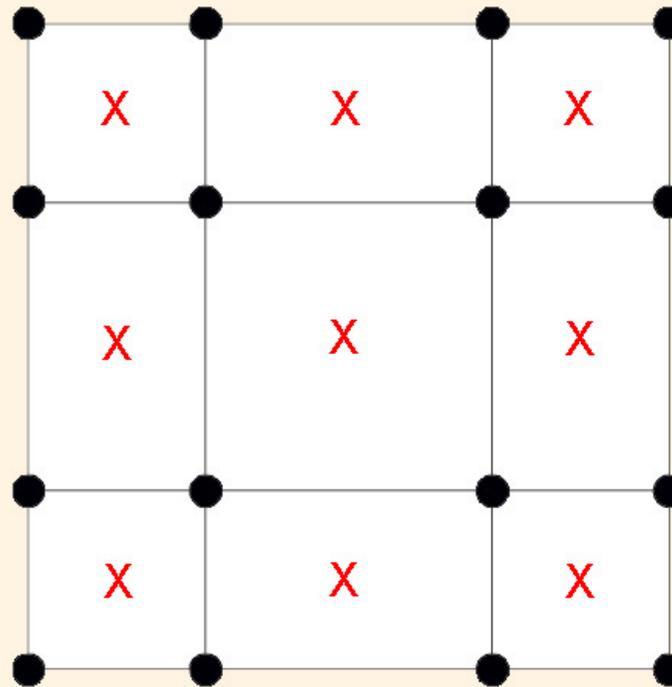


Division of a  $p = 3$  element to obtain a FV-MLS grid with equivalent spatial resolution.



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Division of a  $p = 3$  element to obtain a FV-MLS grid with equivalent spatial resolution.



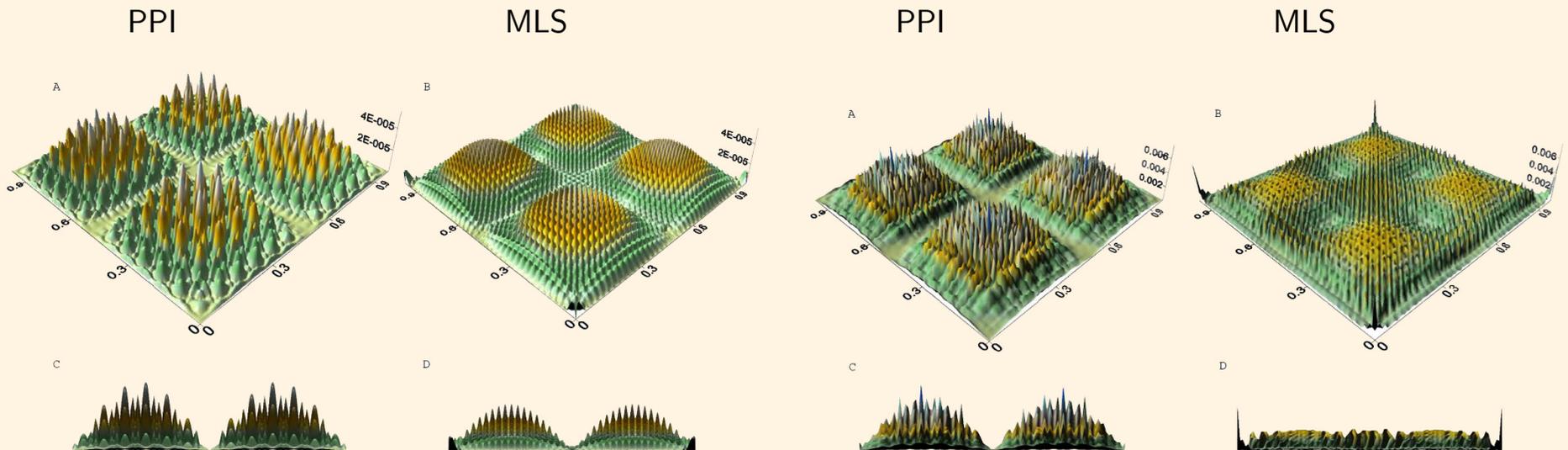
# Interpolation on a structured grid

► Function:  $u(x, y) = \sin(2\pi x) \sin(2\pi y)$

- $13 \times 13$   $p = 3$  elements on a cartesian  $[0, 1] \times [0, 1]$  grid.
- $39 \times 39$  FV-MLS elements on a cartesian  $[0, 1] \times [0, 1]$  grid.
- We interpolate for **both grids** at the **same points** (located at the  $4 \times 4$  Gauss-Legendre points of each FV-MLS element).

Absolute value of the error in the **variable**

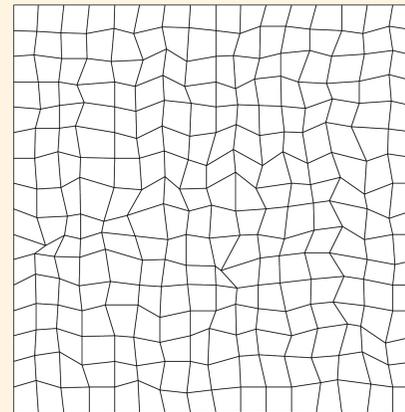
Absolute value of the error in the **derivative**





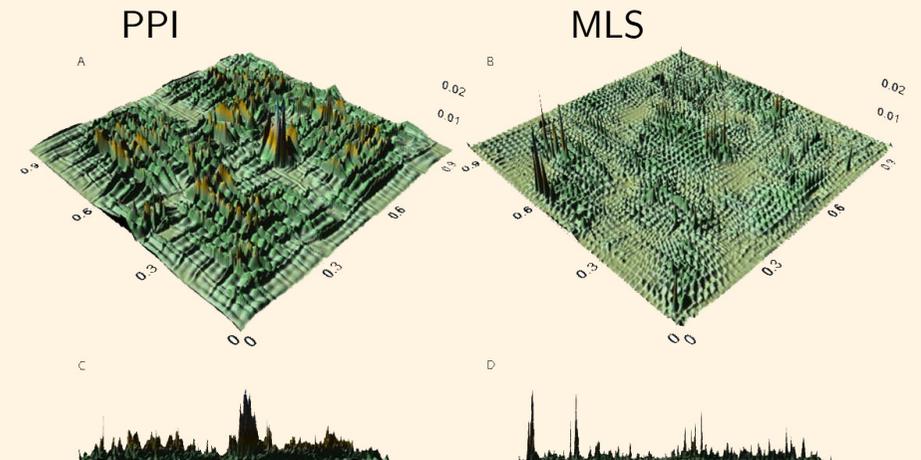
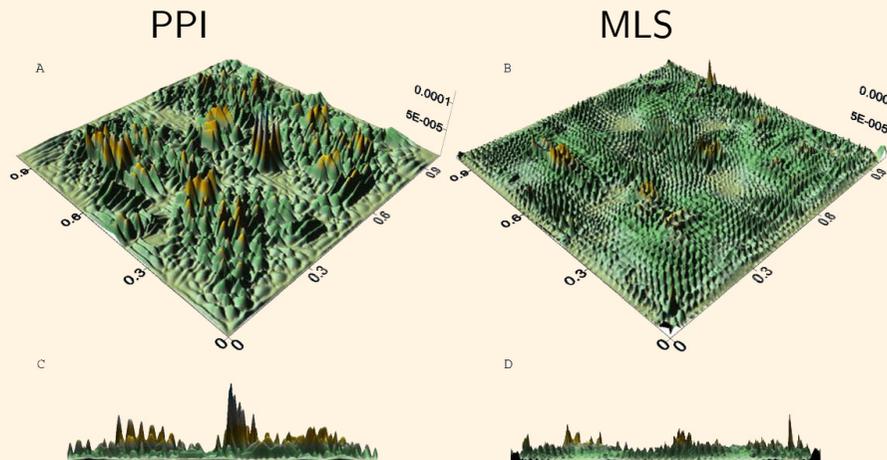
# Interpolation on a distorted grid

► Function:  $u(x, y) = \sin(2\pi x) \sin(2\pi y)$



Absolute value of the error in the **variable**

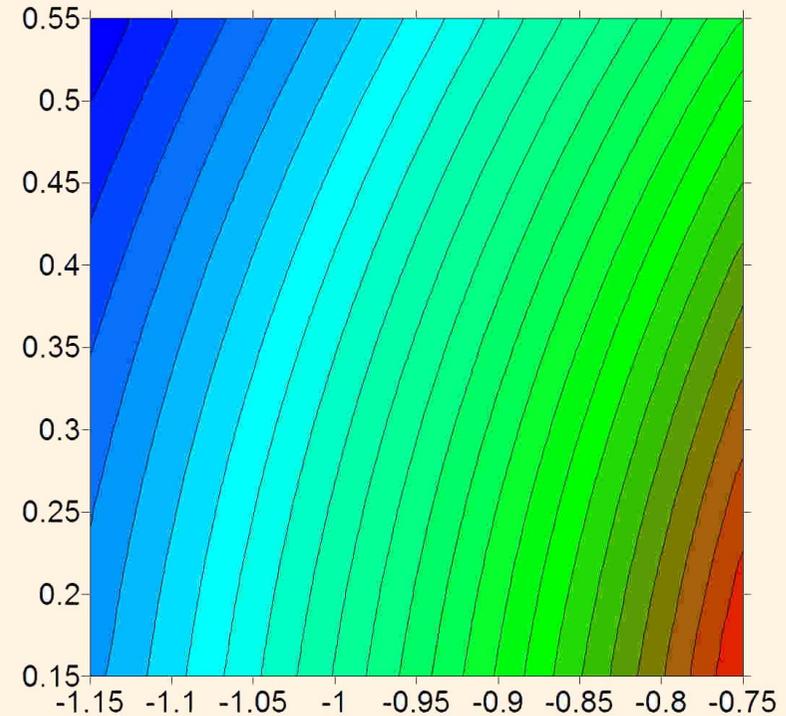
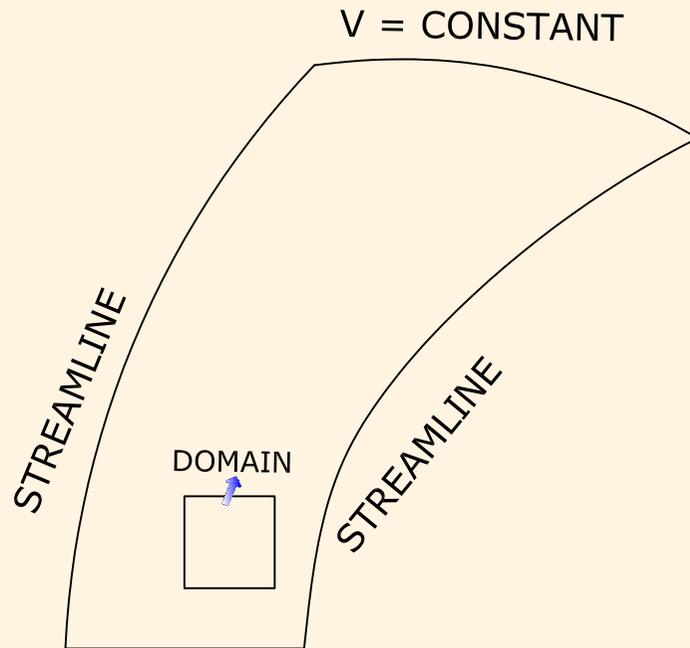
Absolute value of the error in the **derivative**





# Order of Convergence. Ringleb Flow (I)

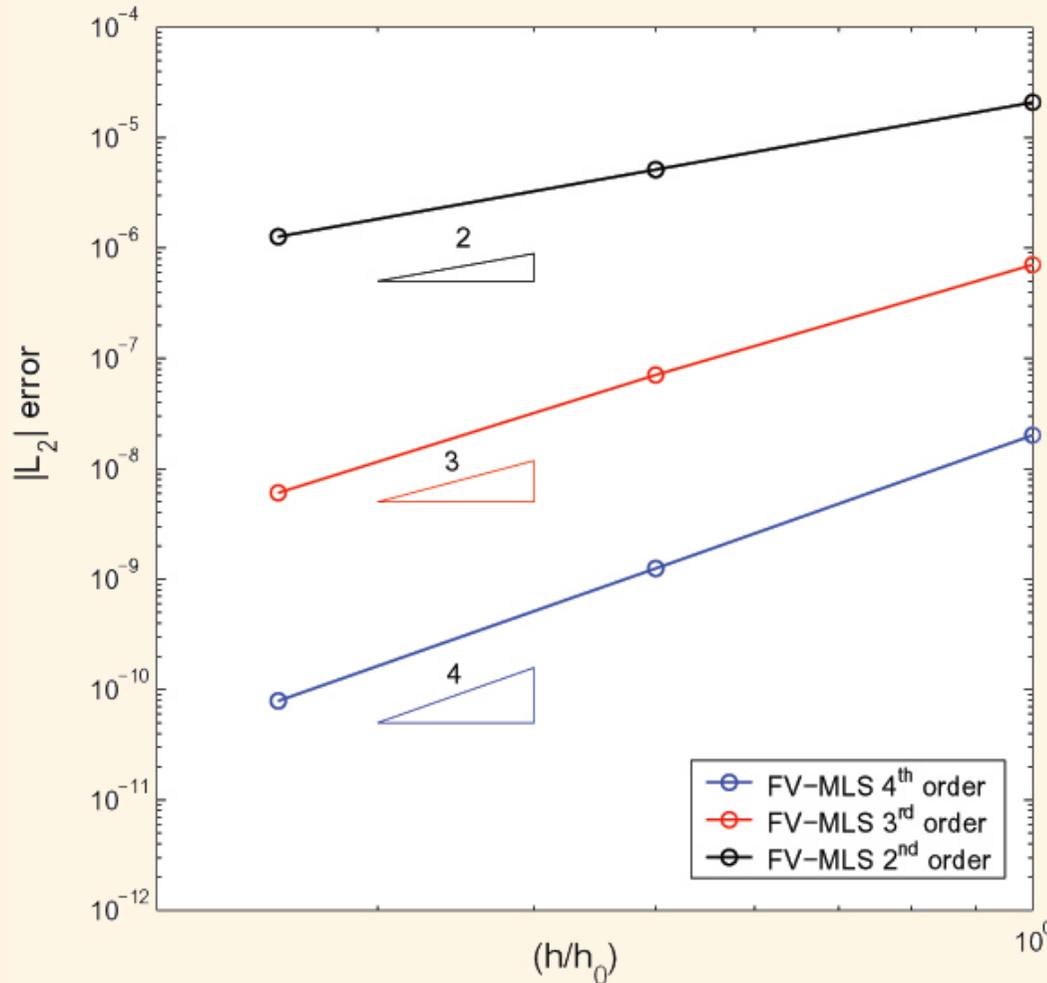
► Domain:  $-1.15 \leq x \leq -0.75$  ,  $0.15 \leq y \leq 0.55$



Mach isolines.



# Order of Convergence. Ringleb Flow (II)



## Grids

FV-MLS
15 × 15
30 × 30
60 × 60

► The order of convergence is the expected one.

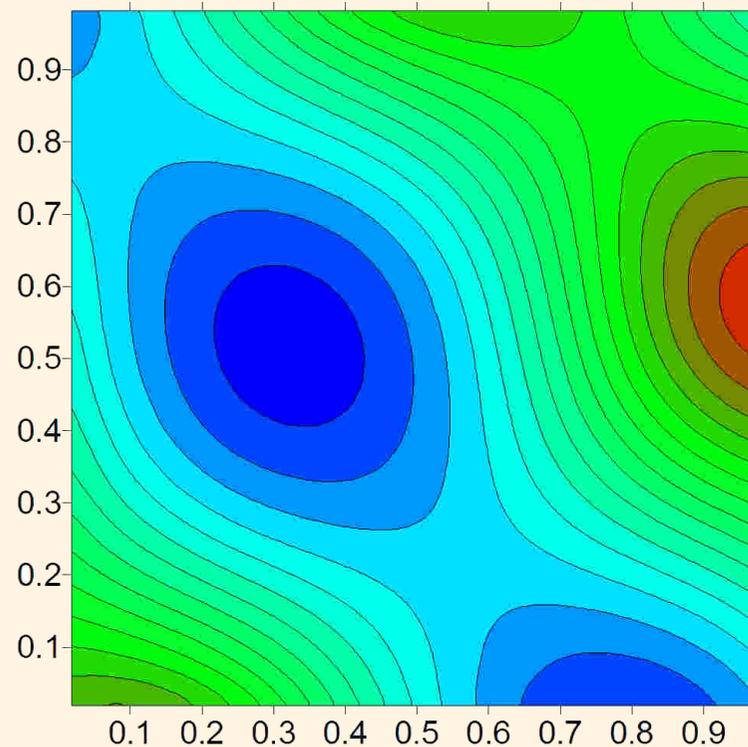




# Order of Convergence. Poisson (I)

$$\begin{aligned} -\Delta u &= f & \text{in } \Omega \\ u &= g_D & \text{on } \Gamma_D \end{aligned}$$

$$u(x, y) = \exp(\alpha \sin(Ax + By)) + \beta \cos(Cx + Dy)$$



Isolines of the exact solution for  $u$ .



# Order of Convergence. Poisson (II)

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$h/h_0$	DG $p = 3$ Error $L_2 u$	FV-MLS Error $L_2 u$
1	$2.50 E - 04$	$8.34 E - 05$
0.5	$1.20 E - 05$	$5.60 E - 06$
0.25	$6.05 E - 07$	$3.75 E - 07$
0.125	$3.16 E - 08$	$2.52 E - 08$

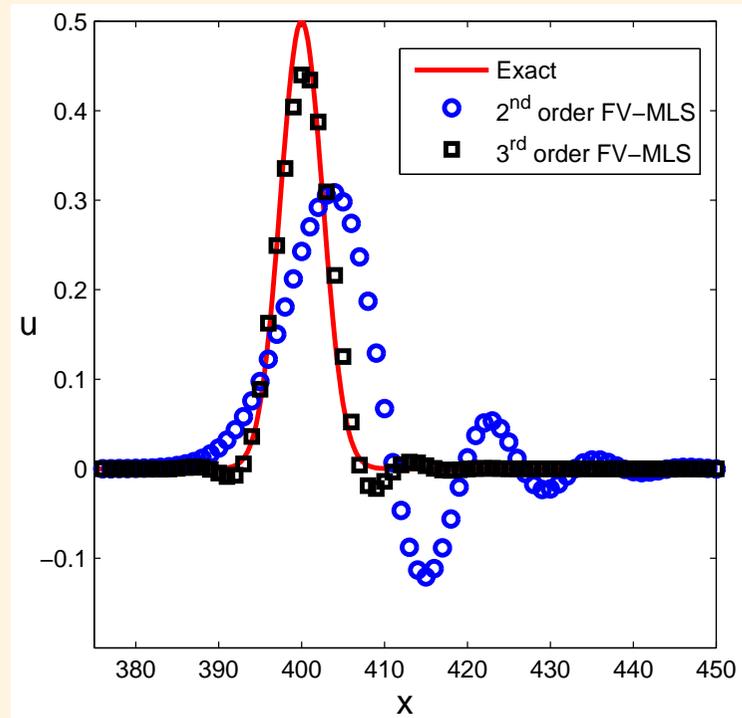
$h/h_0$	DG $p = 3$ Order of Convergence $u$	FV-MLS Order of Convergence $u$	DG $p = 3$ Order of Convergence $s$	FV-MLS Order of Convergence $s$
1	—	—	—	—
0.5	4.38	3.86	3.83	3.54
0.25	4.31	3.99	3.69	3.52
0.125	4.26	3.89	3.60	3.46





# A first motivating example. 1D Linear advection equation

- First ICASE/LaRC Workshop on Benchmark problems in CAA
- We solve  $\frac{\partial u}{\partial t} + a \frac{\partial u}{\partial x} = 0$  with  $u(x, 0) = 0.5e^{-\ln(2)\left(\frac{x}{3}\right)^2}$



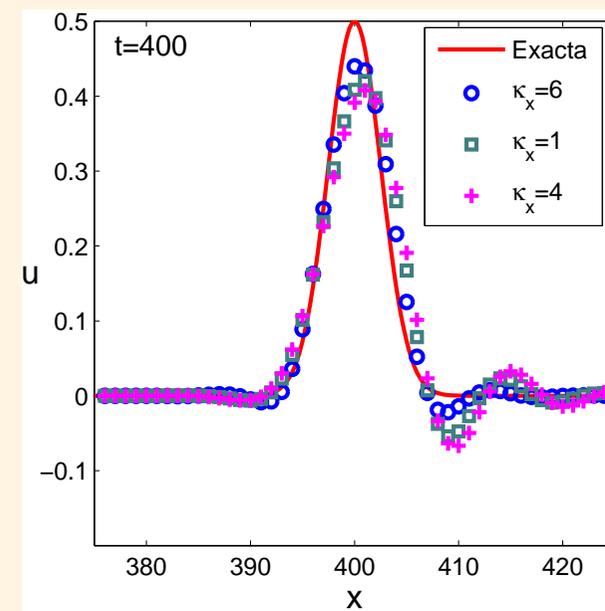
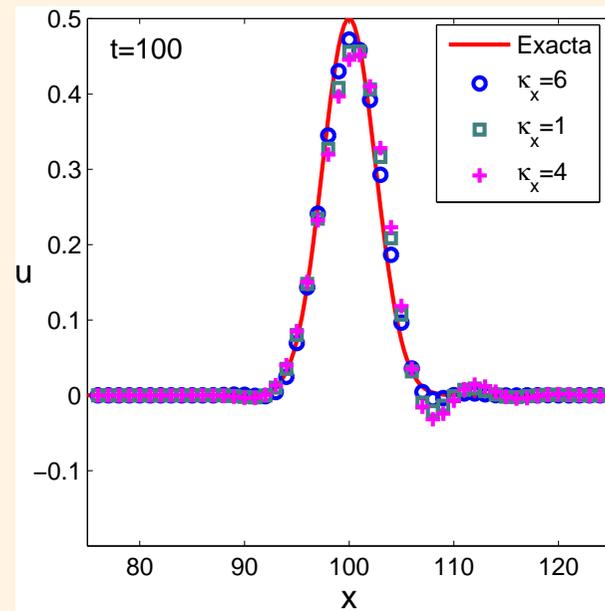
1D Linear advection equation,  $a = 1$ ,  $t = 400$ ,  $\Delta x = 1$ ,  $CFL = 0.6$



# A first motivating example. 1D Linear advection equation

- First ICASE/LaRC Workshop on Benchmark problems in CAA

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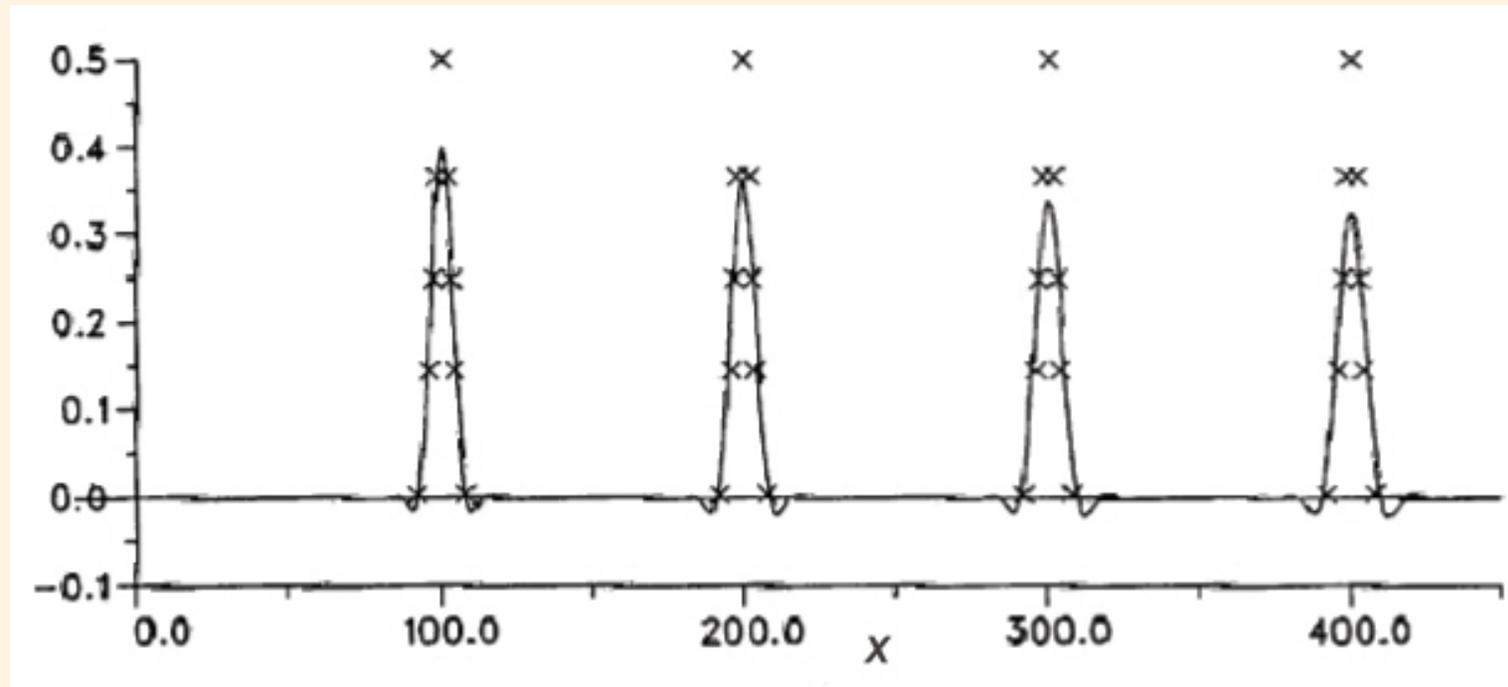


1D Linear advection equation,  $a = 1$ ,  $t = 400$ ,  $\Delta x = 1$ ,  $CFL = 0.6$



# A first motivating example. 1D Linear advection equation

- Solution with a **fourth order** MacCormack scheme,  $\Delta x = 1$ ,  $CFL = 0.2$

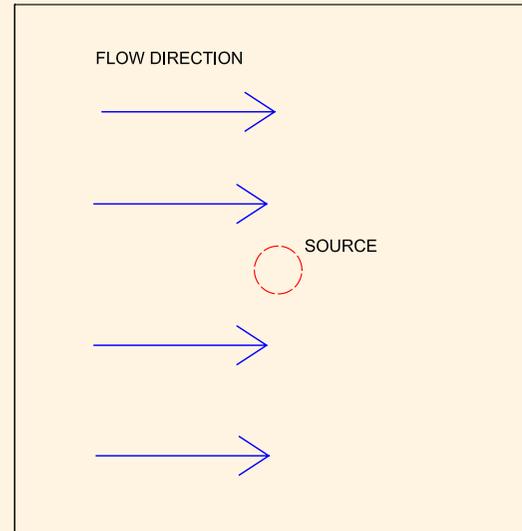


Viswanathan, Sankar, A Comparative Study of Upwind and MacCormac schemes for CAA Benchmark problems, First ICASE/LaRC Workshop on Benchmark problems in CAA, NASA Conference Publication 3300, 185-195, 1995



## A CAA example on an unstructured grid.

- Solve the LEE for the convection of a monopolar source



Sketch of the problem.

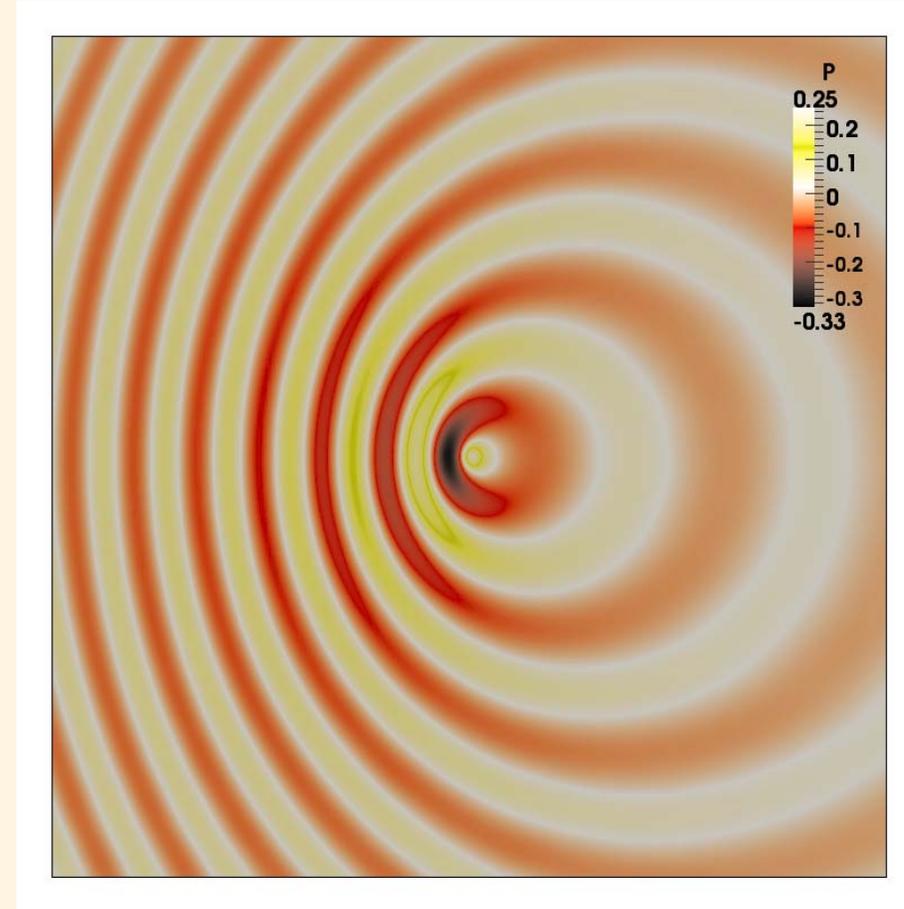
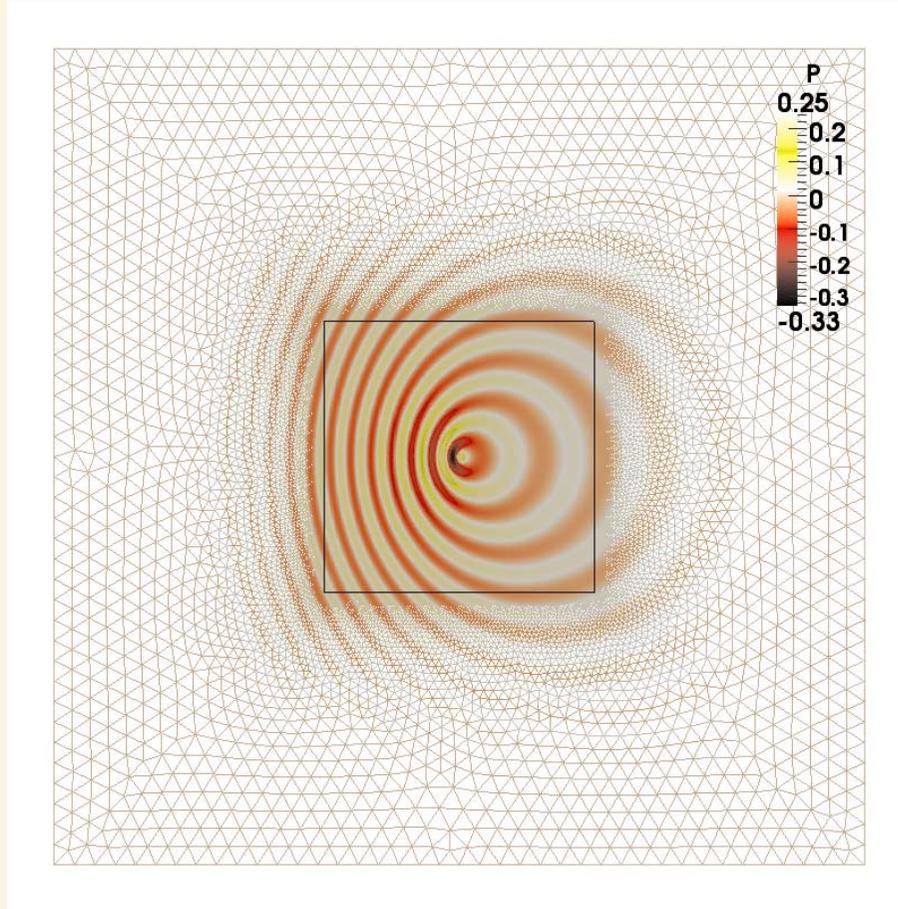
$$S = \epsilon e^{-\alpha[(x-x_s)^2 + (y-y_s)^2]} \sin wt$$





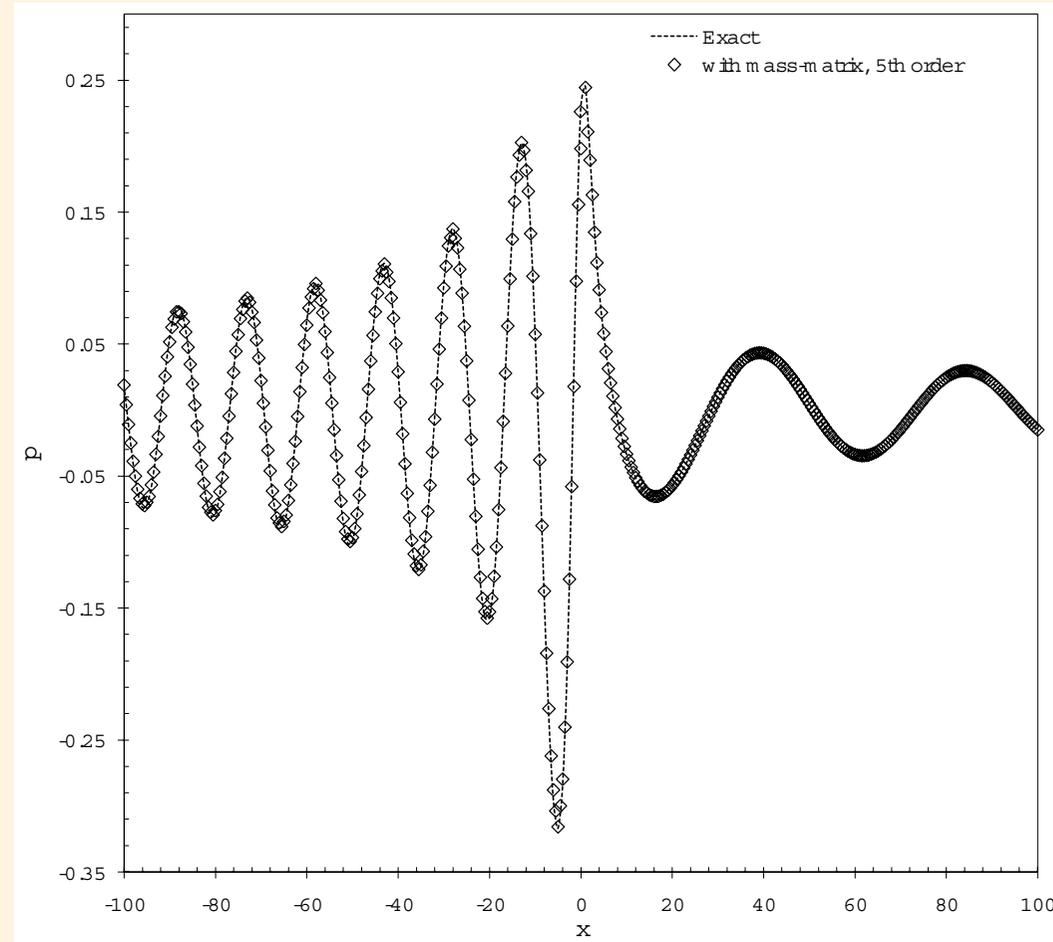
# A CAA example on an unstructured grid.

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# A CAA example on an unstructured grid.



Acoustic pressure profile across  $y = 0$





# A high-order formulation for incompressible flows

- Introduction
- The FV-MLS method
- A high-order formulation for incompressible flows
- High-order Fluid-Structure-Interaction techniques
- Conclusions





# A high-order formulation for incompressible flows

- ▷ Introduction
- ▷ Formulation
- ▷ Numerical Examples

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# A high-order formulation for incompressible flows

- ▷ Introduction
- ▷ Formulation
- ▷ Numerical Examples

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# Introduction

## ► Incompressibility assumption:



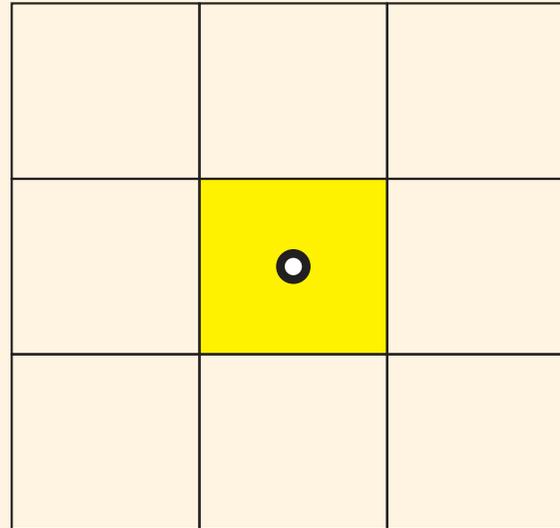
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# Introduction

## ► Checkerboard:



$$\int_{\Omega_I} \frac{\partial p}{\partial x} d\Omega = \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} [p_j \hat{n}_{xj}]_{ig} W_{ig}$$

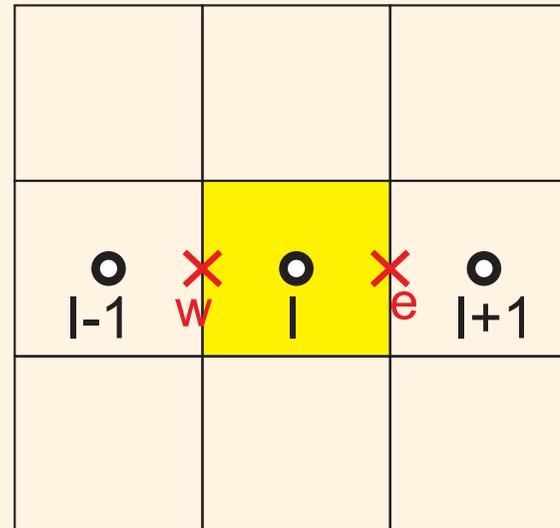
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# Introduction

## ► Checkerboard:

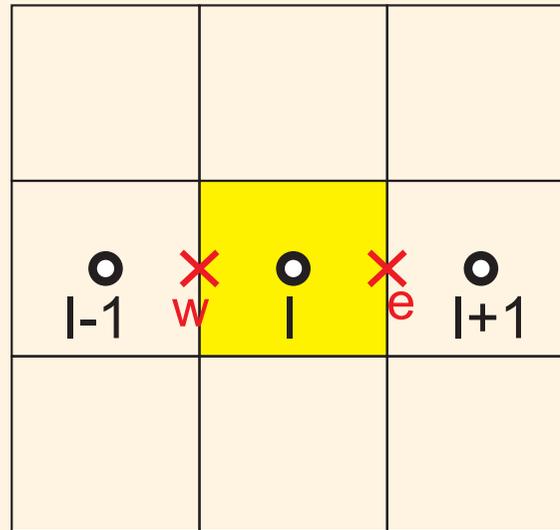


$$\int_{\Omega_I} \frac{\partial p}{\partial x} d\Omega = (p\hat{n}_x)_e + (p\hat{n}_x)_w$$



# Introduction

## ► Checkerboard:



$$\int_{\Omega_I} \frac{\partial p}{\partial x} d\Omega = (p)_e - (p)_w$$

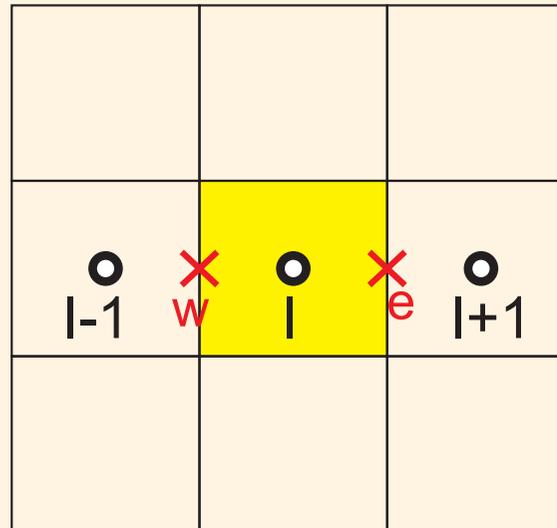
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# Introduction

## ► Checkerboard:



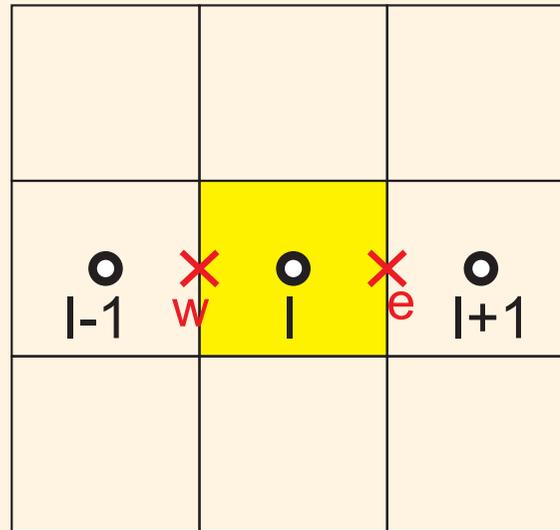
$$(p)_e = \frac{p_{i+1} + p_i}{2} \quad (p)_w = \frac{p_i + p_{i-1}}{2}$$

$$\int_{\Omega_I} \frac{\partial p}{\partial x} d\Omega = (p)_e - (p)_w$$



# Introduction

## ► Checkerboard:



$$(p)_e = \frac{p_{i+1} + p_i}{2} \quad (p)_w = \frac{p_i + p_{i-1}}{2}$$

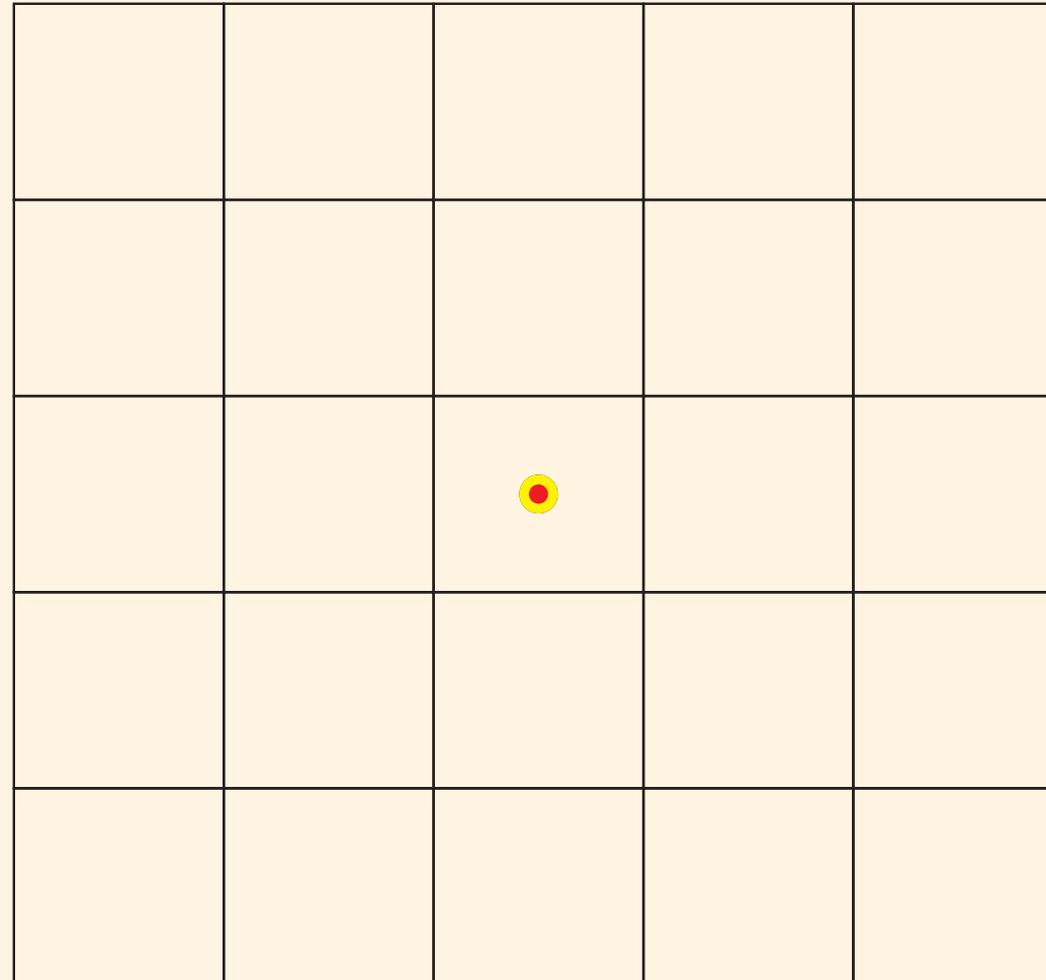
$$\int_{\Omega_I} \frac{\partial p}{\partial x} d\Omega = \frac{p_{i+1} - p_{i-1}}{2}$$





# Introduction

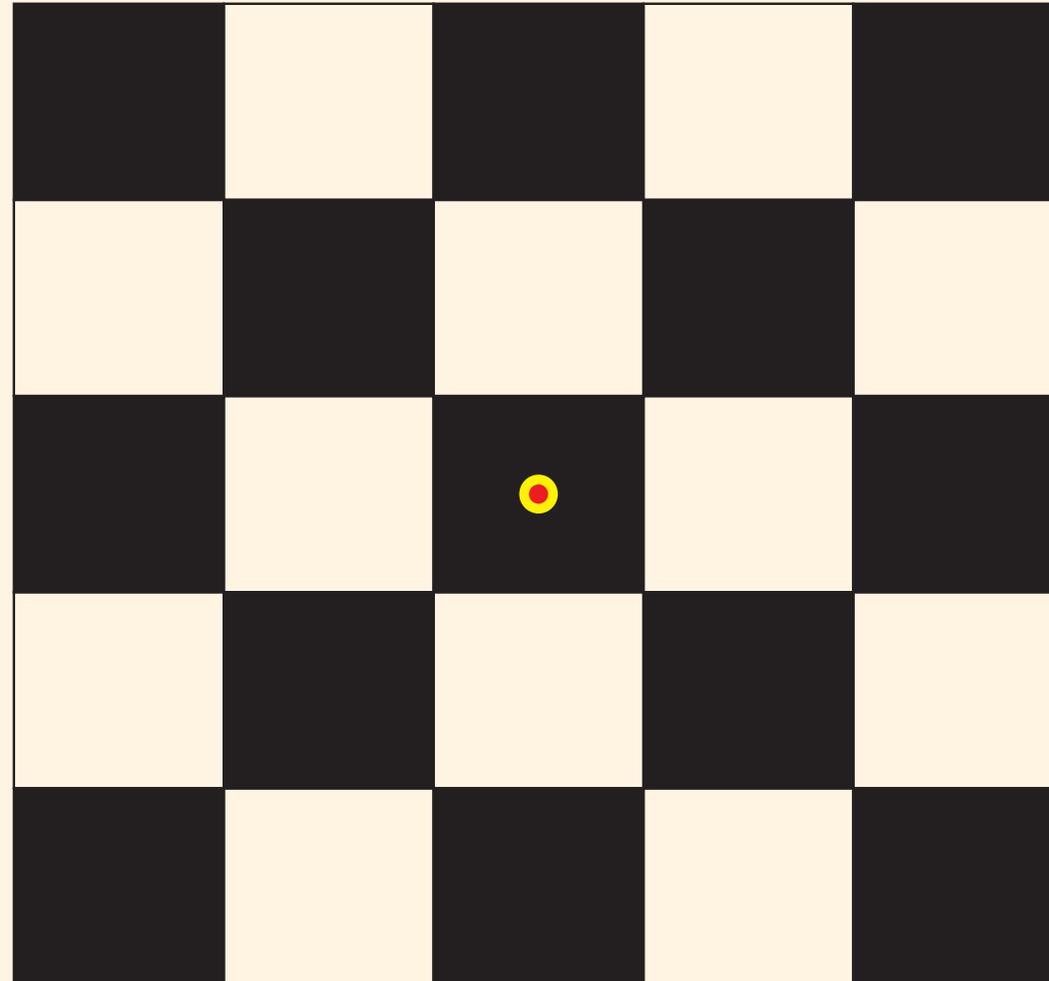
## ► Checkerboard:





# Introduction

## ► Checkerboard:





# Introduction

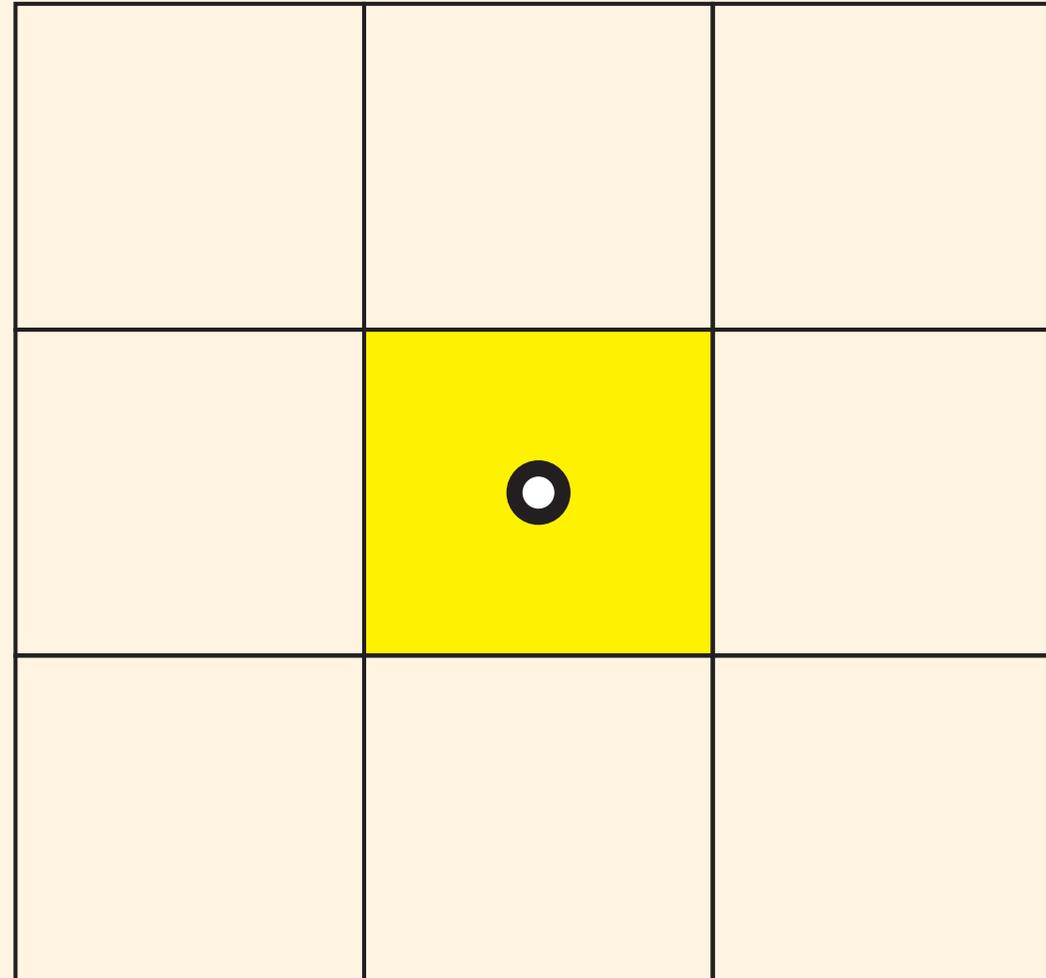
- ▶ In order to solve the checkerboard:
  - Collocated grid arrangement → Special interpolation (MIM)
  - Staggered grid arrangement → Special location of the variables





# Introduction

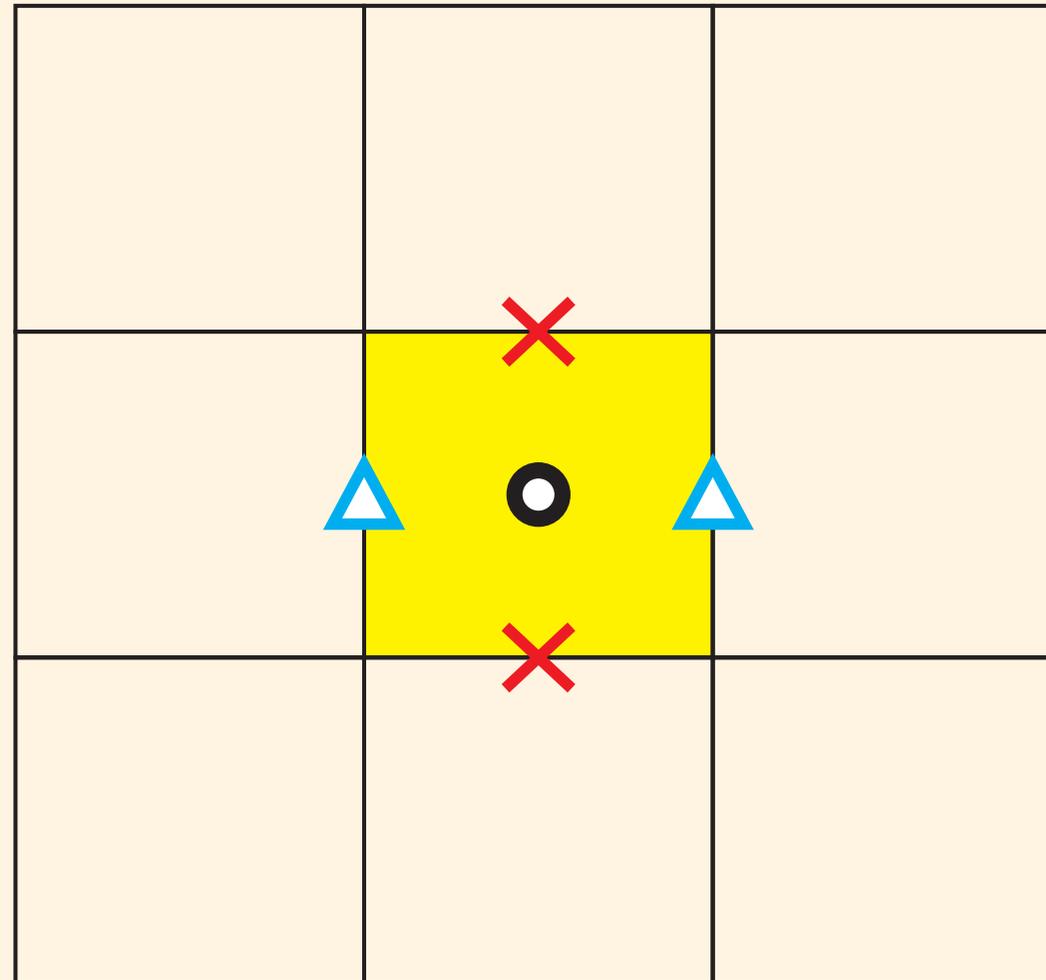
- ▶ Collocated grid arrangement  $\rightarrow u, v, p$  located at cell centroid.





# Introduction

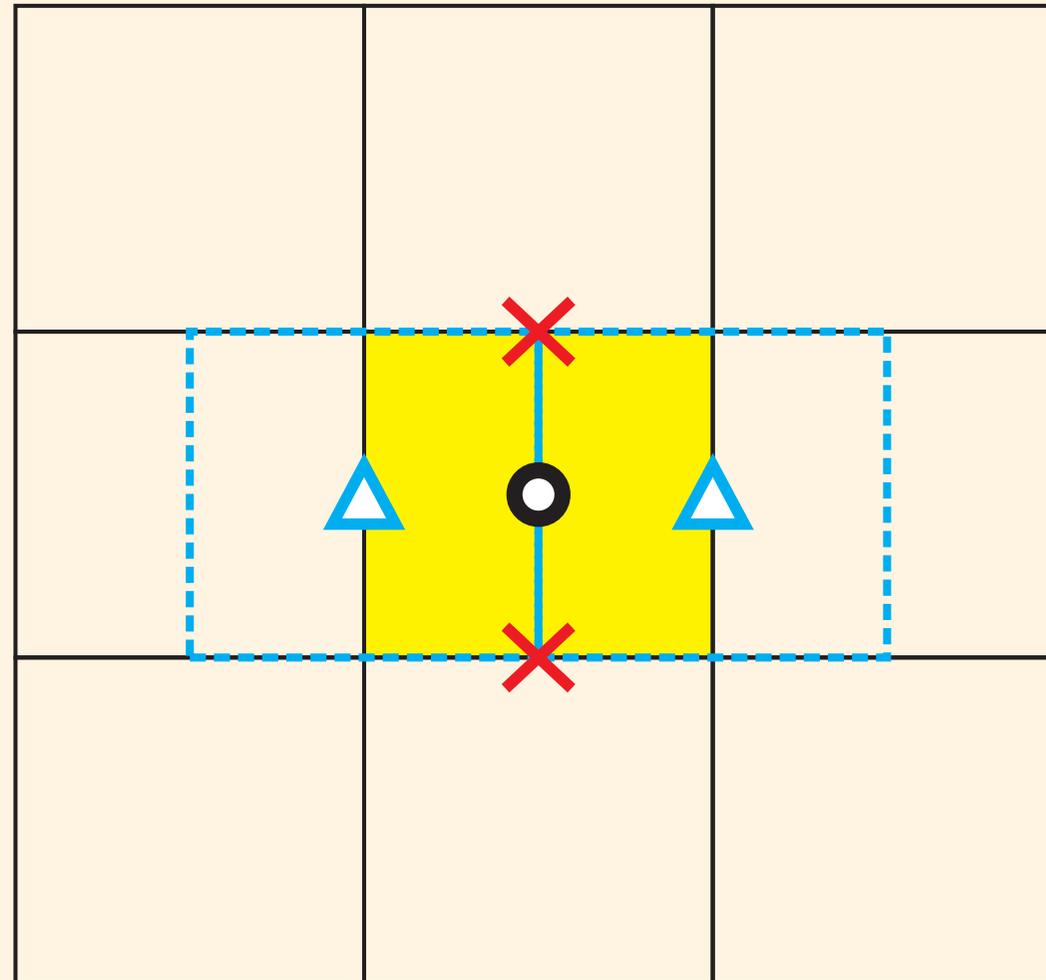
- ▶ Staggered grid  $\rightarrow u, v, p$  located at different locations.





# Introduction

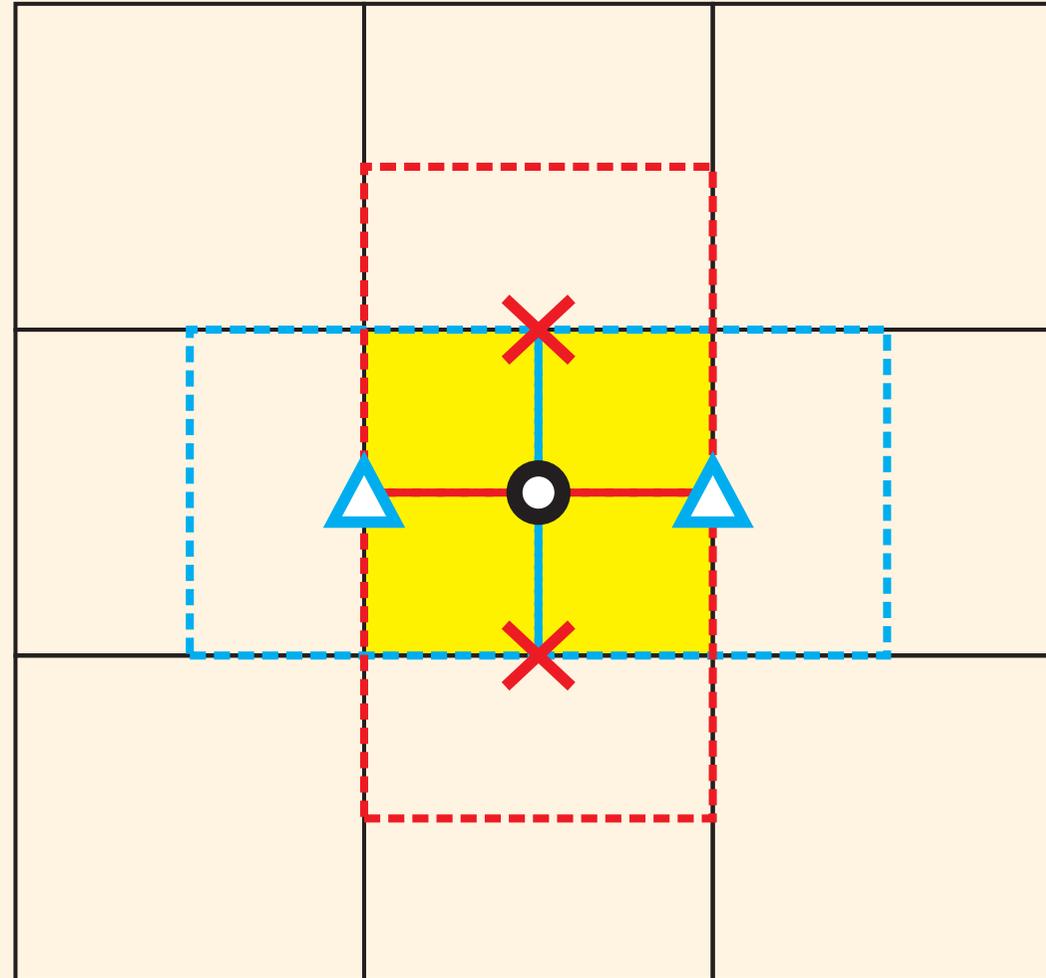
- ▶ Staggered grid  $\rightarrow u$  control volume.





# Introduction

- ▶ Staggered grid  $\rightarrow v$  control volume.





# Introduction

- ▶ Staggered grid arrangement:
  - Variables stored at different locations
  - No interpolations required
  - Drawback → Complex in unstructured and/or 3D grids
- ▶ Collocated grid arrangement:
  - Variables stored at cell centroid
  - Structured and unstructured grid
  - Drawback → Possibility of checkerboard





# Introduction

- ▶ Staggered grid arrangement:
  - Variables stored at different locations
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# Introduction

- ▶ **Staggered grid arrangement:**
  - Variables stored at different locations
  - No interpolations required
  - Drawback → Complex in unstructured and/or 3D grids
- ▶ **Collocated grid arrangement:**
  - Variables stored at cell centroid
  - Structured and unstructured grids
  - Drawback → Possibility of checkerboard
  - Special interpolation is required





# A high-order formulation for incompressible flows

- ▷ Introduction
- ▷ Formulation
- ▷ Numerical Examples

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# Incompressible Navier Stokes

## ► Incompressible Navier-Stokes:

$$\frac{\partial \mathbf{U}}{\partial t} + \mathbf{U} \cdot (\nabla \mathbf{U}) = -\nabla p + \frac{1}{Re} (\Delta \mathbf{U})$$
$$\nabla \cdot \mathbf{U} = 0$$

where  $\mathbf{U} = (u, v)^T$  is the velocity field,  $p(x, y, t)$  is the pressure variable and  $Re$  denotes the Reynolds number.

## ► Resolution procedure:

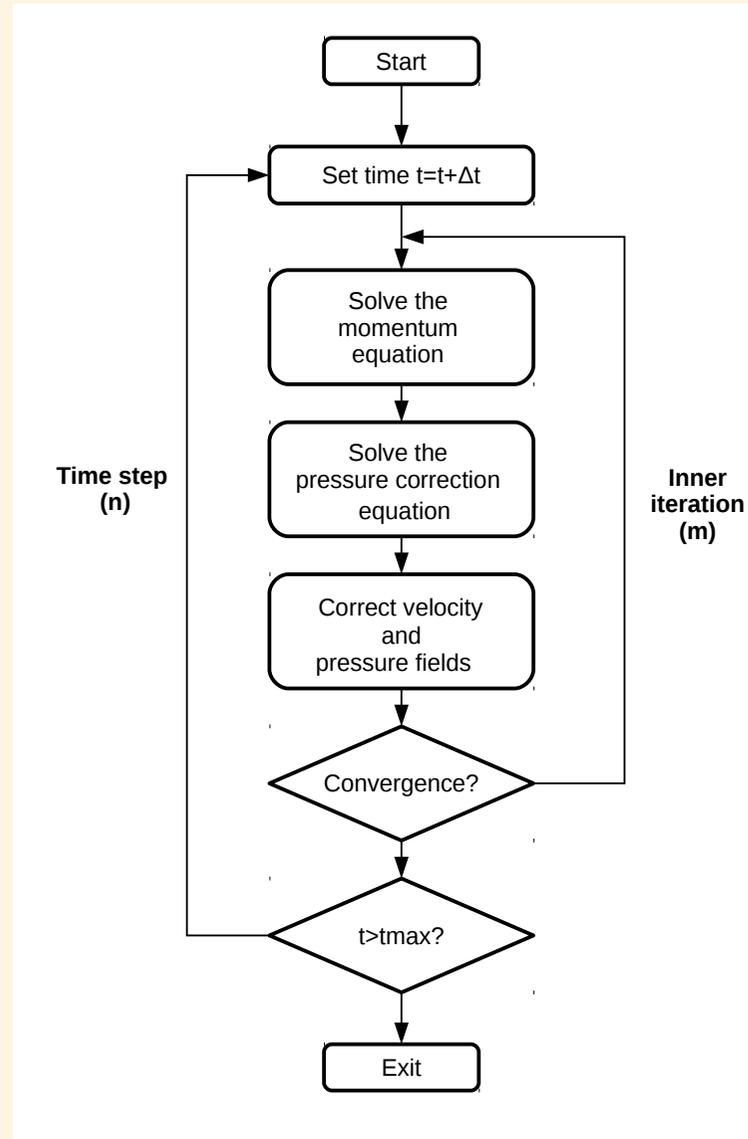
- A collocated Semi-Implicit Method for Pressure Linked Equations (SIMPLE).
- Momentum Interpolation Method to avoid checkerboard oscillations.





# SIMPLE

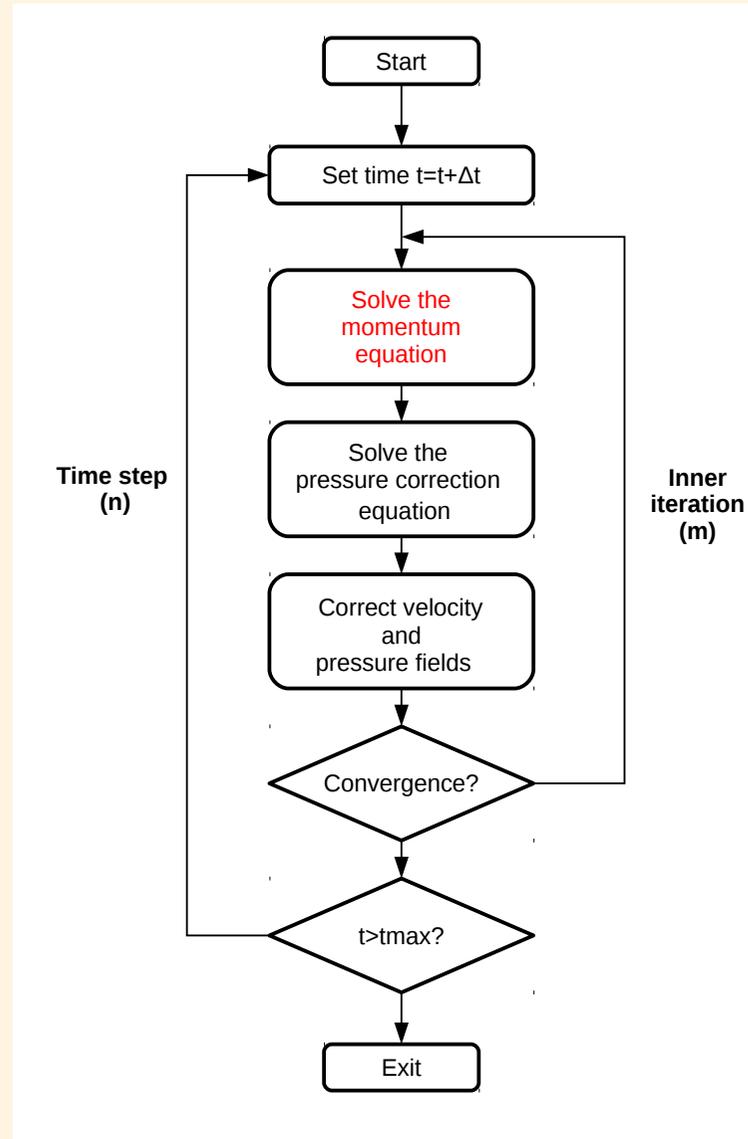
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# SIMPLE

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## ► Momentum equation

- Cell centered finite volume scheme

$$\int_{\Omega_I} \frac{\partial \mathbf{U}}{\partial t} d\Omega + \int_{\Omega_I} \mathbf{U} \cdot (\nabla \mathbf{U}) d\Omega = - \int_{\Omega_I} \nabla p d\Omega + \frac{1}{Re} \left( \int_{\Omega_I} (\Delta \mathbf{U}) d\Omega \right)$$

- Discretized momentum equation

$$V_I \frac{3\mathbf{U}_I^{m+1,n+1} - 4\mathbf{U}_I^n + \mathbf{U}_I^{n-1}}{2\Delta t} + \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} \left[ H_j^{m,n+1} \mathbf{U}_j^{m+1,n+1} \right]_{ig} \mathcal{W}_{ig} =$$

$$= - \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} \left[ p_j^{m,n+1} \cdot \hat{\mathbf{n}}_j \right]_{ig} \mathcal{W}_{ig} + \frac{1}{Re} \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} \left[ \nabla \mathbf{U}_j^{m+1,n+1} \cdot \hat{\mathbf{n}}_j \right]_{ig} \mathcal{W}_{ig}$$



## ► Momentum equation

- Cell centered finite volume scheme

$$\int_{\Omega_I} \frac{\partial \mathbf{U}}{\partial t} d\Omega + \int_{\Omega_I} \mathbf{U} \cdot (\nabla \mathbf{U}) d\Omega = - \int_{\Omega_I} \nabla p d\Omega + \frac{1}{Re} \left( \int_{\Omega_I} (\Delta \mathbf{U}) d\Omega \right)$$

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$$= - \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} \left[ p_j^{m,n+1} \cdot \hat{\mathbf{n}}_j \right]_{ig} \mathcal{W}_{ig} + \frac{1}{Re} \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} \left[ \nabla \mathbf{U}_j^{m+1,n+1} \cdot \hat{\mathbf{n}}_j \right]_{ig} \mathcal{W}_{ig}$$



# SIMPLE

► Higher-order approximations are made using MLS:

- Pressure term 
$$p_j = \sum_{k=1}^{n\mathbf{x}} N_k^g(\mathbf{x}_j) p_k$$

- Diffusive term 
$$\nabla \mathbf{U}_j = \sum_{l=1}^{n\mathbf{x}} \nabla N_l^g(\mathbf{x}_j) \mathbf{U}_l$$

- Convective term  $\Rightarrow$  Deferred correction approach

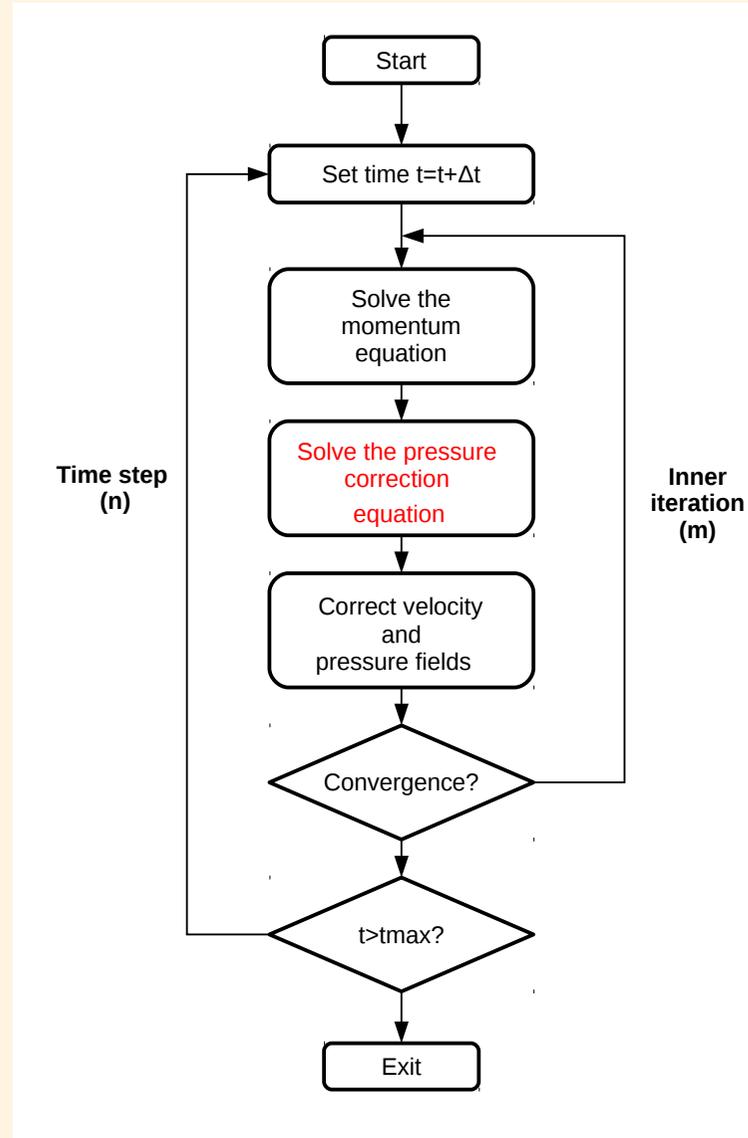
$$\mathbf{U}_j = (\mathbf{U}_j^{LO})^{m+1, n+1} + (\mathbf{U}_j^{HO} - \mathbf{U}_j^{LO})^{m, n+1}$$

$$\mathbf{U}_j^{LO} = \begin{cases} \mathbf{U}_I & , H_j \geq 0 \\ \mathbf{U}_N & , H_j < 0 \end{cases} \quad \mathbf{U}_j^{HO} = \sum_{k=1}^{n\mathbf{x}} N_k^g(\mathbf{x}_j) \mathbf{U}_k$$



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## ► Pressure correction equation

$$\int_{\Omega_I} \nabla \cdot \mathbf{U} d\Omega = \int_{\Gamma_I} \mathbf{U} \cdot \mathbf{n}_j d\Gamma = \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} [\hat{\mathbf{U}}_j \cdot \hat{\mathbf{n}}_j]_{ig} \mathcal{W}_{ig}$$

- In order to avoid checkerboard oscillations  $\Rightarrow$  Momentum Interpolation Method (MIM)

$$\hat{\mathbf{U}}_j = \mathbf{U}_j^* + \left( \frac{V_I}{a_I} \right)_j \left[ (\overline{\nabla p_I})_j - \nabla p_j \right]$$

- The MIM was proposed by Rhie and Chow in 1983.





# SIMPLE

## ► Pressure correction equation

- Checkerboard oscillations  $\Rightarrow$  Momentum Interpolation Method

$$\hat{\mathbf{U}}_j = \mathbf{U}_j^* + \left( \frac{V_I}{a_I} \right)_j \left[ (\overline{\nabla p_I})_j - \nabla p_j \right]$$

- ▷ These terms are usually obtained at integration point  $j$  using linear interpolation.
- ▷ We propose to use higher-order approximations using MLS

$$\mathbf{U}_j^* = \sum_{k=1}^{n\mathbf{x}} N_k^g(\mathbf{x}_j) \mathbf{U}_k^* \quad \left( \frac{V_I}{a_I} \right)_j = \sum_{k=1}^{n\mathbf{x}} N_k^g(\mathbf{x}_j) \left( \frac{V_I}{a_I} \right)_k$$

$$(\overline{\nabla p_I})_j = \sum_{k=1}^{n\mathbf{x}} N_k^g(\mathbf{x}_j) \nabla p_k \quad \nabla p_j = \sum_{l=1}^{n\mathbf{x}} \nabla N_l^g(\mathbf{x}_j) p_l$$



## ► Pressure correction equation

- A pressure correction equation is solved in order to impose the continuity “constraint”.

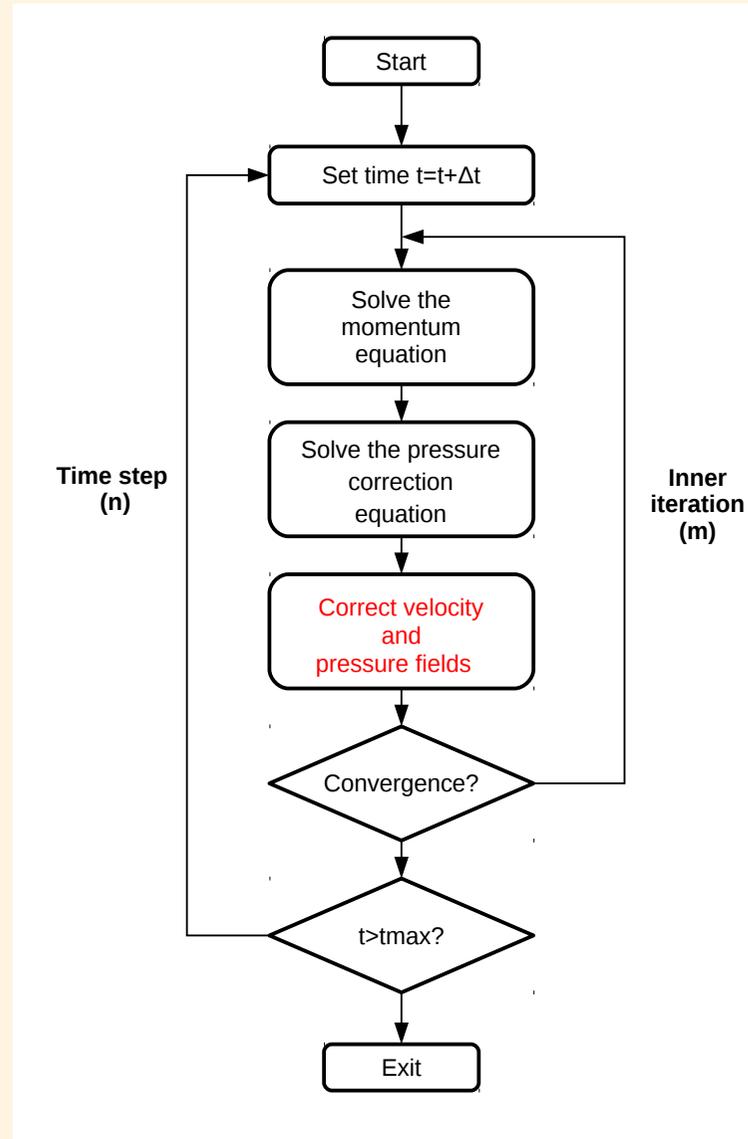
$$\sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} [\hat{\mathbf{U}}_j \cdot \hat{\mathbf{n}}_j]_{ig} \mathcal{W}_{ig} - \sum_{j=1}^{N_f} \sum_{ig=1}^{N_G} \left[ \left( \frac{V_I}{a_I} \right)_j (\nabla p')_j \cdot \hat{\mathbf{n}}_j \right]_{ig} \mathcal{W}_{ig} = 0$$

- The pressure correction,  $p'$ , is the unknown.
- Approximations at integration point are obtained with MLS.



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# SIMPLE

- ▶ Correct velocity and pressure fields at cell centroids as

$$\mathbf{U}^{m+1,n+1} = \mathbf{U}^* + \mathbf{U}' = \mathbf{U}^* - \frac{V_I}{a_I} \left( \nabla p' \right)_I$$

$$p^{m+1,n+1} = p^{m,n+1} + \left( p' \right)^{m+1,n+1}$$

- The value  $\left( \nabla p' \right)_I$  is approximated at cell centroid using MLS

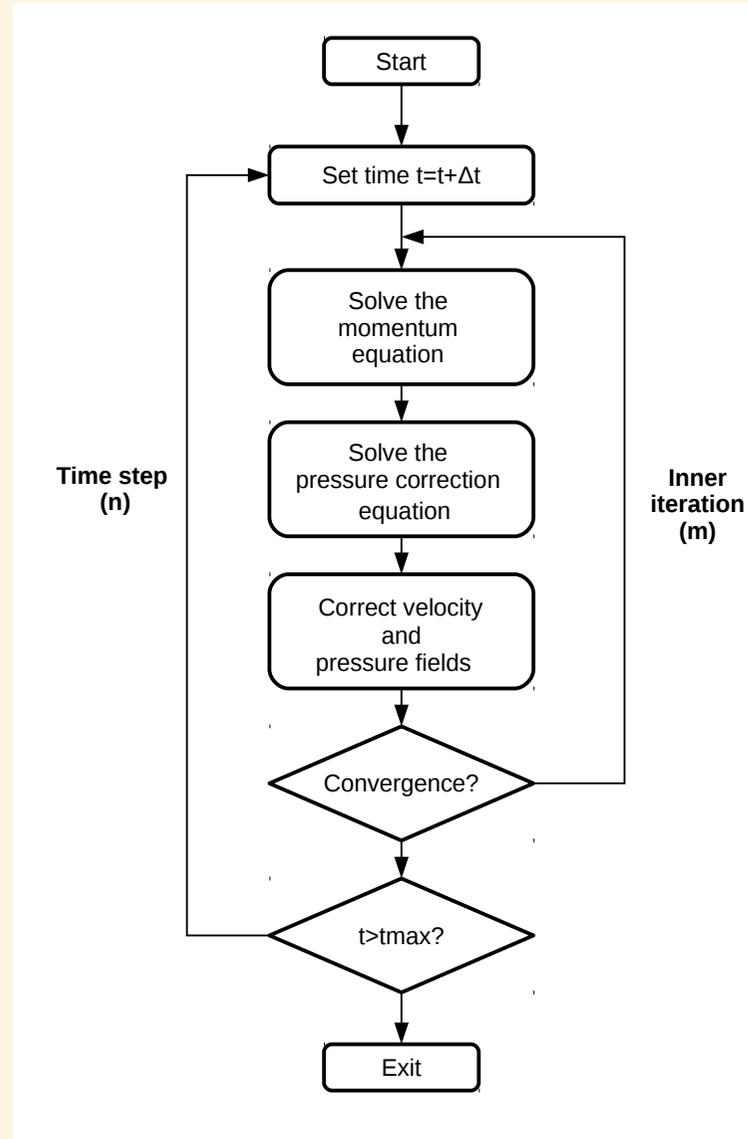
$$\nabla p'_I = \sum_{l=1}^{n_{\mathbf{x}}} \nabla N_l^g(\mathbf{x}_j) p'_l$$





# SIMPLE

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# A high-order formulation for incompressible flows

- ▷ Introduction
- ▷ Formulation
- ▷ Numerical Examples

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# Numerical Examples

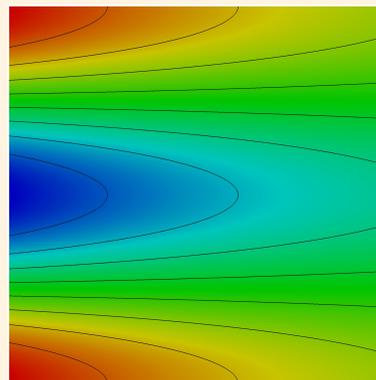
## ► Kovaszny Flow.

$$u(x, y) = 1 - e^{\alpha x} \cos(2\pi y)$$

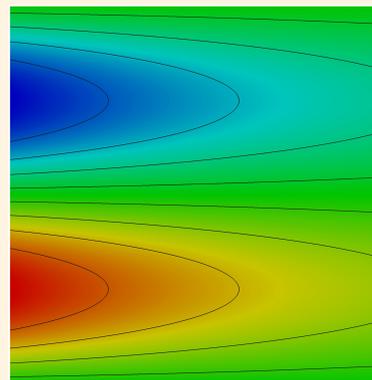
$$v(x, y) = \frac{\alpha}{2\pi} e^{\alpha x} \sin(2\pi y) \quad \alpha = \frac{Re}{2} - \sqrt{\frac{Re^2}{4} + 4\pi^2}$$

$$p(x, y) = \frac{1}{2} (1 - e^{2\alpha x})$$

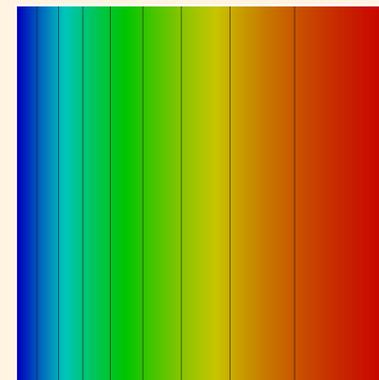
- Domain  $\Omega = [-0.5, 0.5] \times [0.5, 0.5]$ .  $Re=40$



$u$



$v$

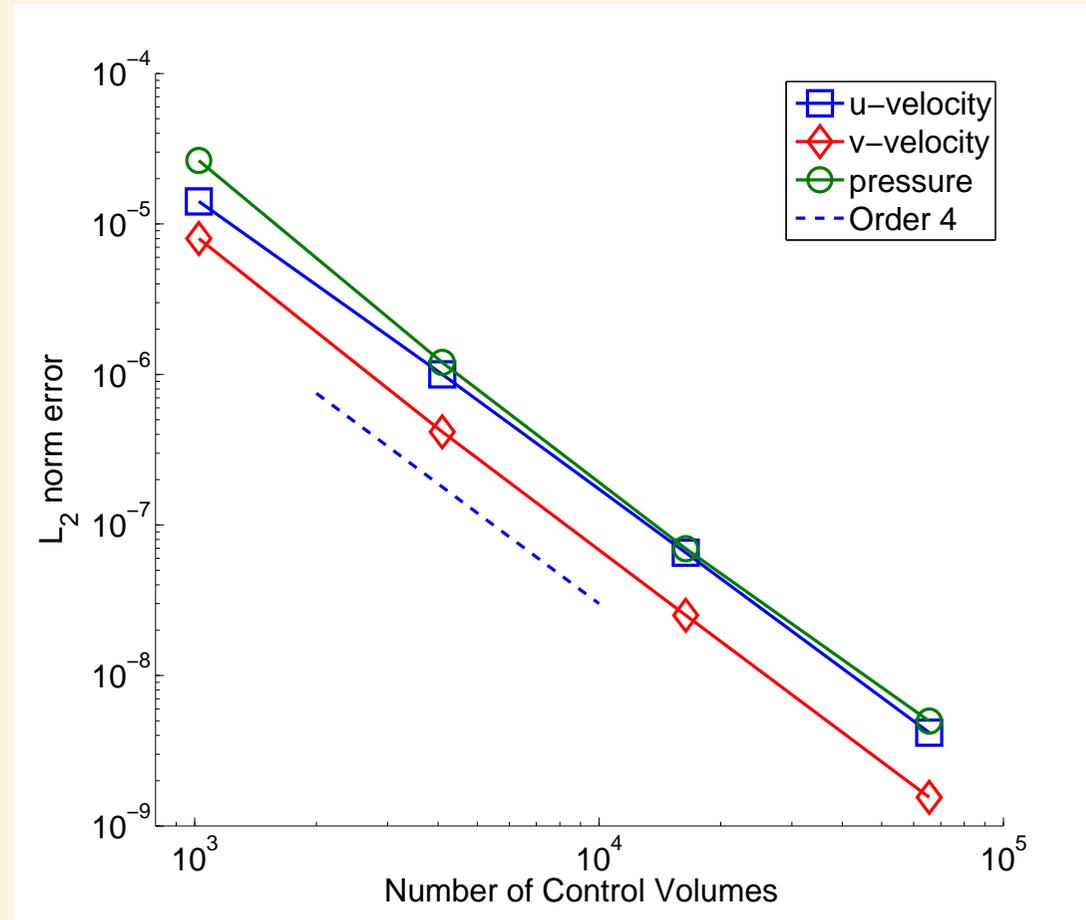


$p$



# Numerical Examples

## ► Kovaszny Flow.



- The formal order of accuracy is recovered





# Numerical Examples

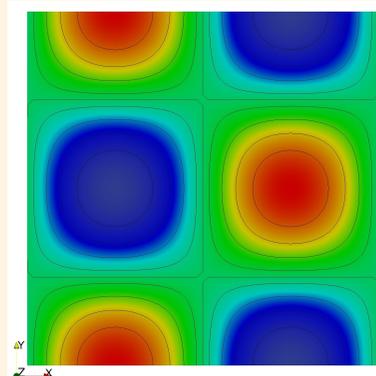
## ► 2D Taylor-Green Flow.

$$u(x, y, t) = e^{\frac{-2t}{Re}} \cos(y) \sin(x)$$

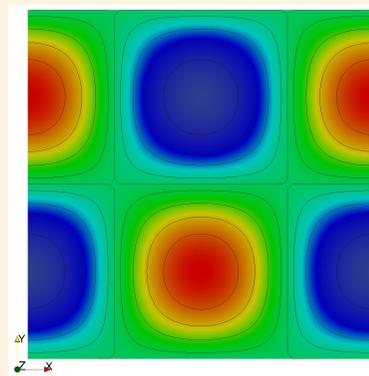
$$v(x, y, t) = -e^{\frac{-2t}{Re}} \cos(x) \sin(y)$$

$$p(x, y, t) = \frac{e^{\frac{-4t}{Re}}}{4} (\cos(2x) + \cos(2y))$$

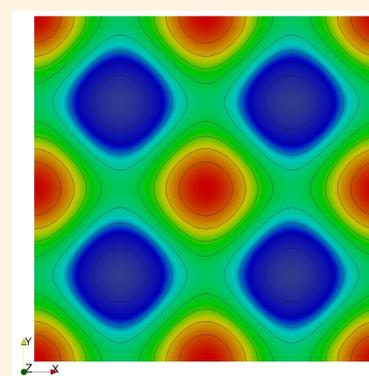
- Domain  $\Omega = [0, 2\pi] \times [0, 2\pi]$ .  $Re=100$



$u$



$v$

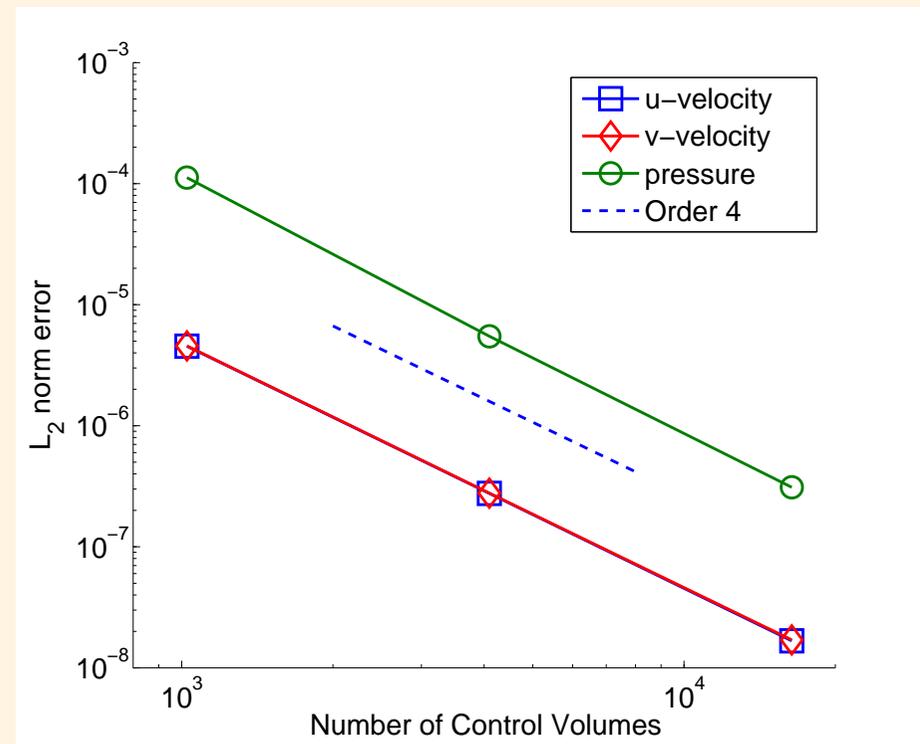
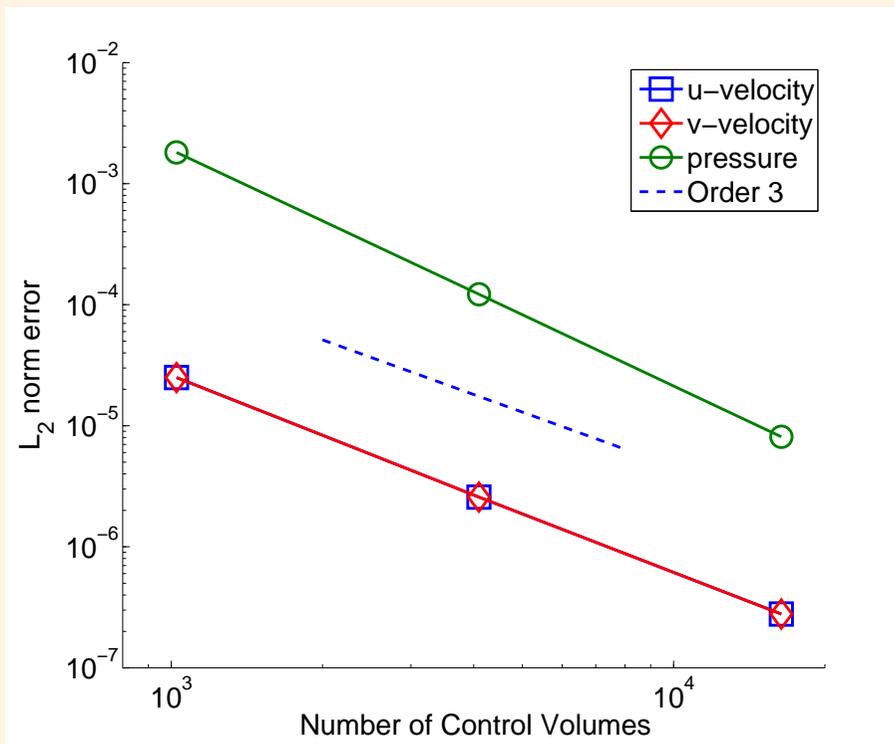


$p$



# Numerical Examples

## ► 2D Taylor-Green Flow.



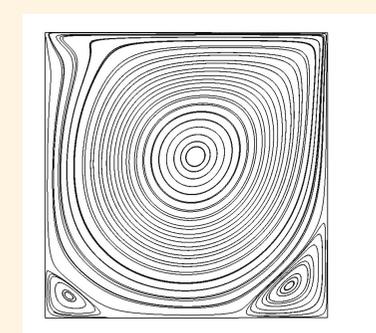
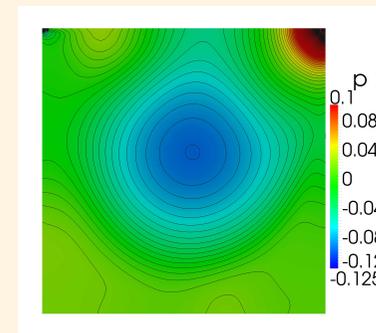
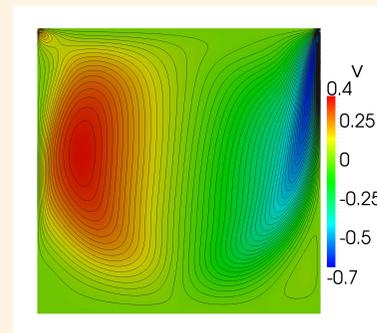
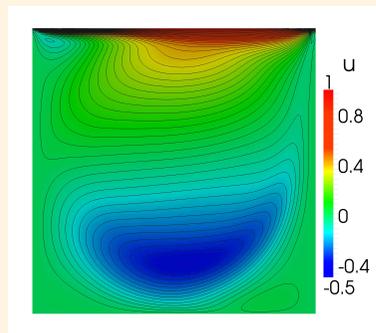
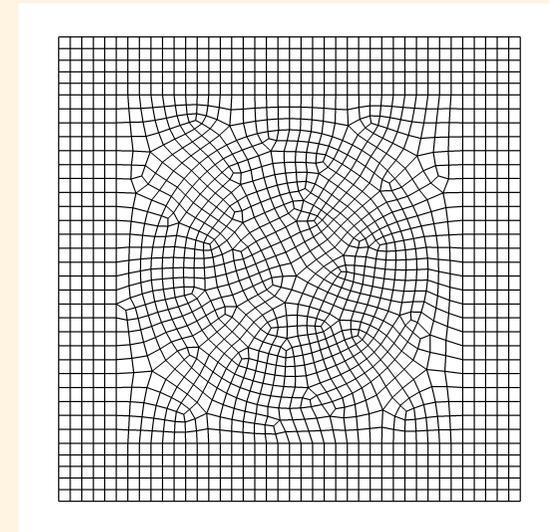
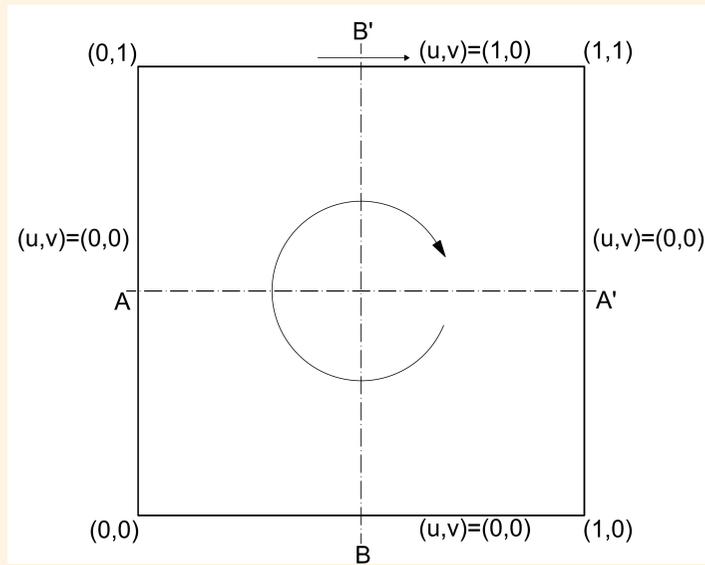
- The formal order of accuracy is recovered



# Numerical Examples

## ► Cavity Flow

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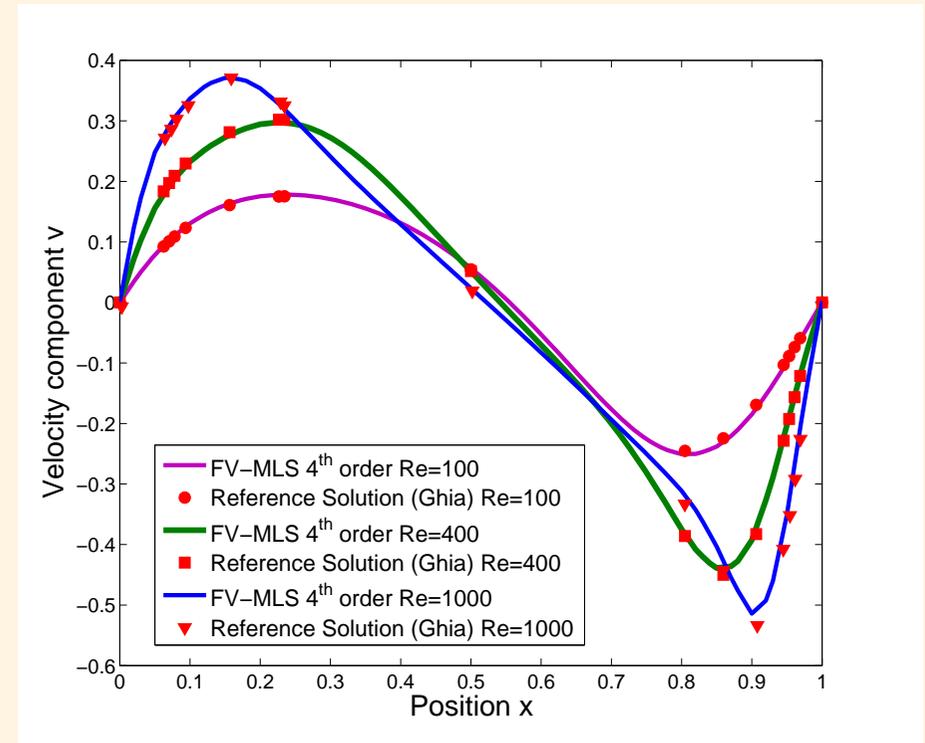
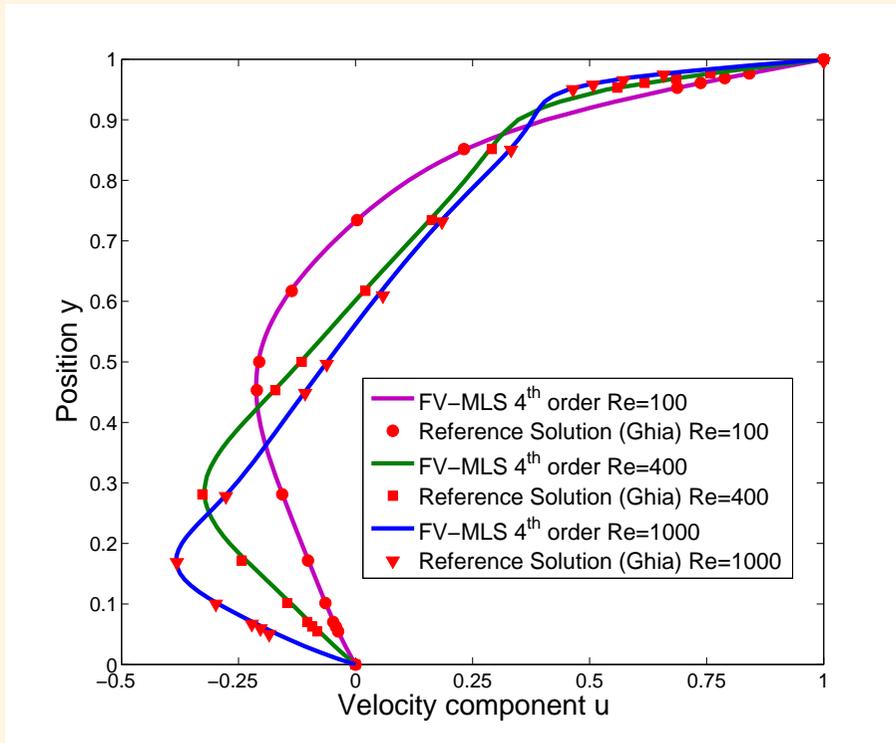
Re=1000





# Numerical Examples

## ► Cavity Flow. 1635 cells.



- Excellent agreement with the reference solution for different Reynolds number. The reference solution is obtained on a 128x128 structured mesh (16384 cells).

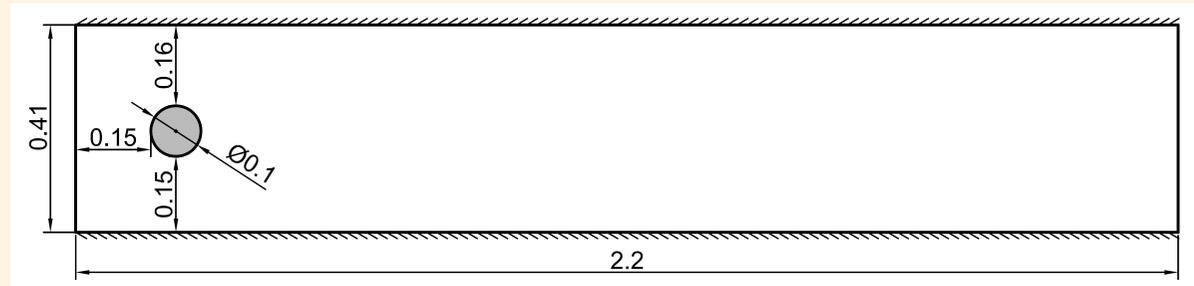




# Numerical Examples

## ► Laminar Flow around a cylinder

- Benchmark proposed by Schäfer and Turek.



- Parabolic velocity profile at inlet

$$u(0, y) = \frac{4U_m y(H - y)}{H^2}, v(0, y) = 0$$

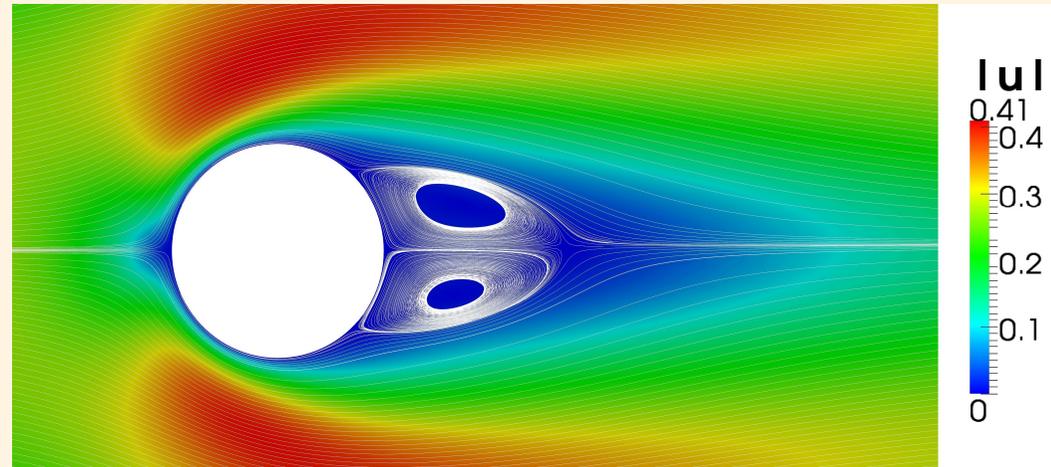
- Two test cases
  - ▷ Reynolds 20
  - ▷ Reynolds 100

Reference Solution: Schäfer, M., Turek, S., *Benchmark Computations of Laminar Flow Around a Cylinder*, Notes on Numerical Fluid Mechanics, Volume 52 , pp. 547-566, 1996.



# Laminar Flow around a cylinder

- Reynolds 20

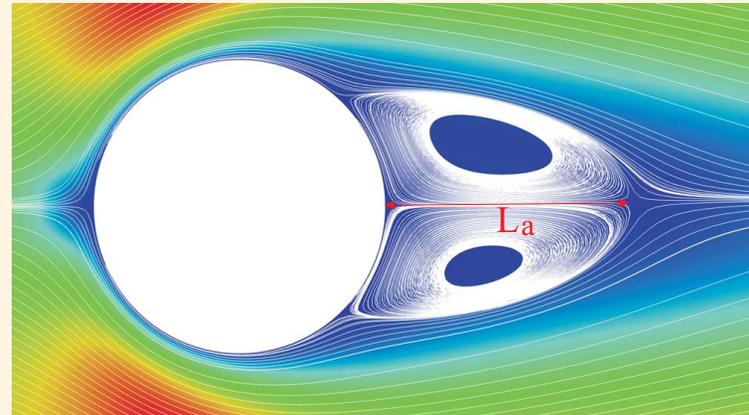


Mesh	Order	$C_D$	$C_L$	$L_a$	$\Delta p$
Mesh A (4968 cells)	2	5.5869	0.0087	0.0881	0.1149
	3	5.5919	0.0108	0.0851	0.1161
Mesh B (19079 cells)	2	5.5817	0.0113	0.0851	0.1168
	3	5.5859	0.0107	0.0845	0.1174
Upper bound	—	5.5900	0.0110	0.0852	0.1176
Lower bound	—	5.5700	0.0104	0.0842	0.1172



# Laminar Flow around a cylinder

- Reynolds 20

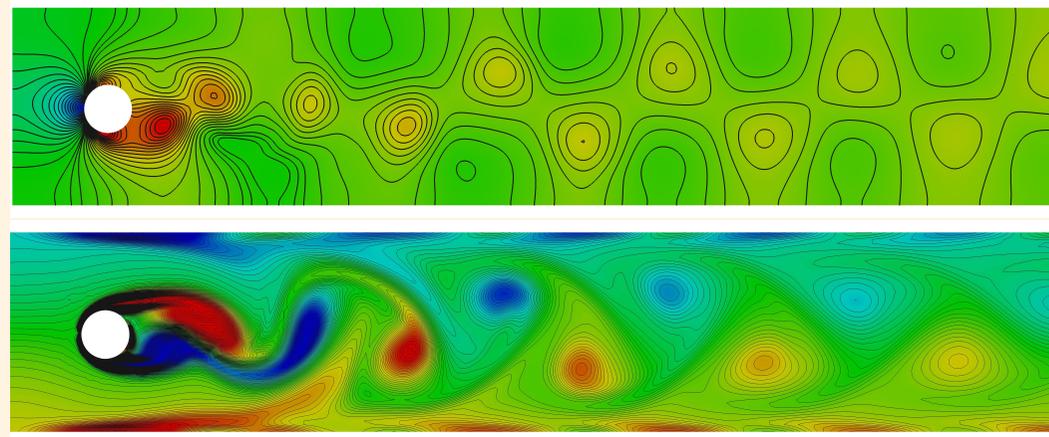


Mesh	Order	$C_D$	$C_L$	$L_a$	$\Delta p$
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Upper bound	—	5.5900	0.0110	0.0852	0.1176
Lower bound	—	5.5700	0.0104	0.0842	0.1172



# Laminar Flow around a cylinder

- Reynolds 100



Mesh	Order	$C_{Dmax}$	$C_{Lmax}$	$St$	$\Delta p$
Mesh A (4968 cells)	2	3.2741	1.2246	0.2825	2.3548
	3	3.2986	1.0451	0.2924	2.3962
Mesh B (19079 cells)	2	3.2702	1.0662	0.2952	2.4731
	3	3.2380	0.9985	0.3008	2.4858
Upper bound	—	3.2400	1.0100	0.3050	2.5000
Lower bound	—	3.2200	0.9900	0.2950	2.4600



# High-order Fluid-Structure-Interaction techniques

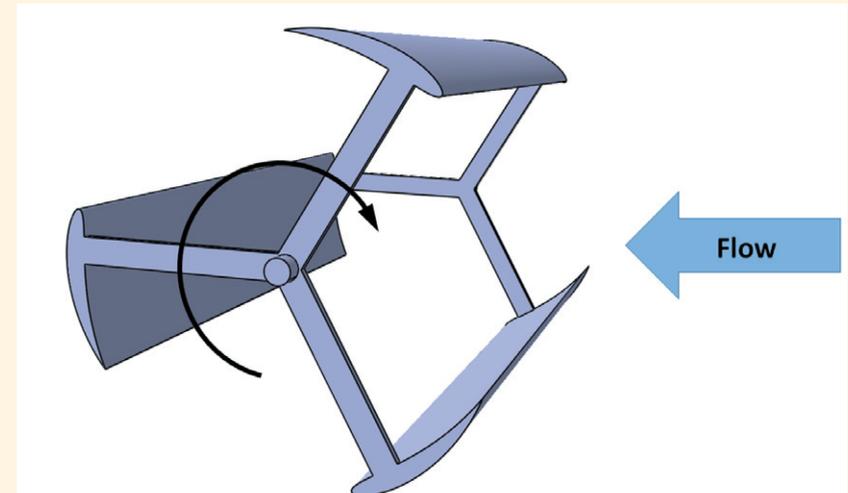
- Introduction
- The FV-MLS method
- A high-order formulation for incompressible flows
- High-order Fluid-Structure-Interaction techniques
- Conclusions





# High-order Fluid-Structure-Interaction techniques

- ▶ Incompressible flow around a cross-flow turbine.



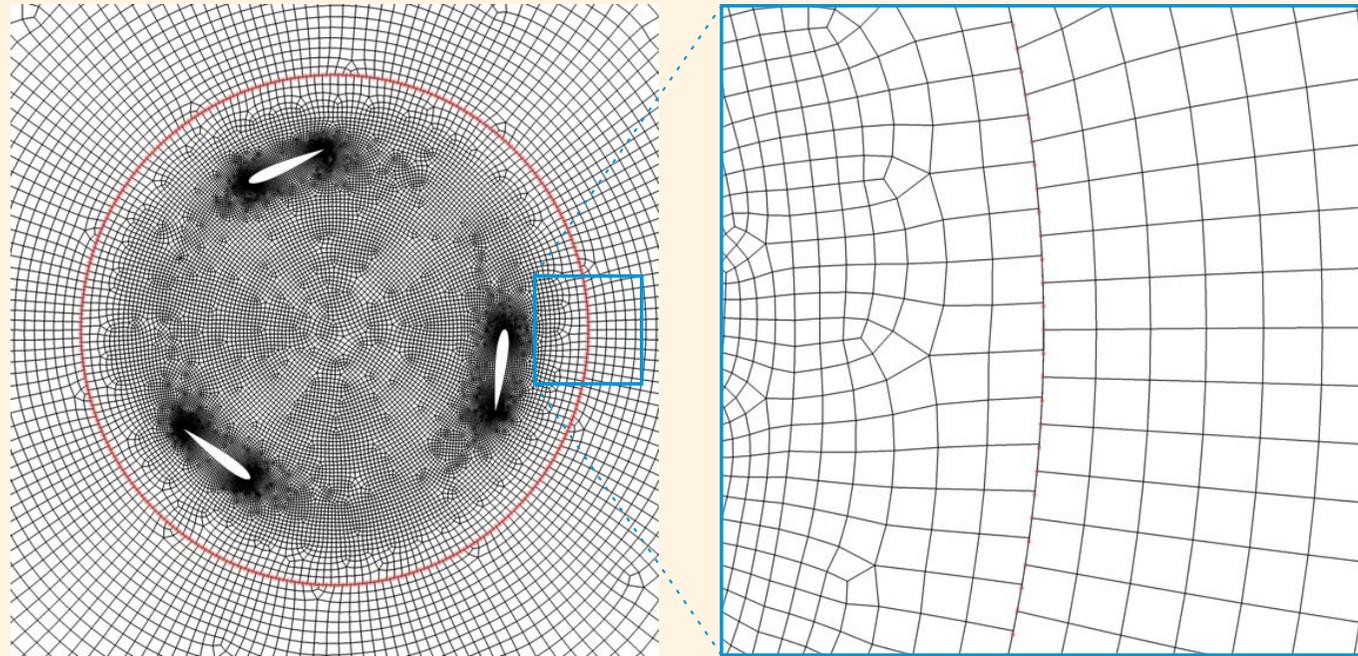
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# High-order Fluid-Structure-Interaction techniques

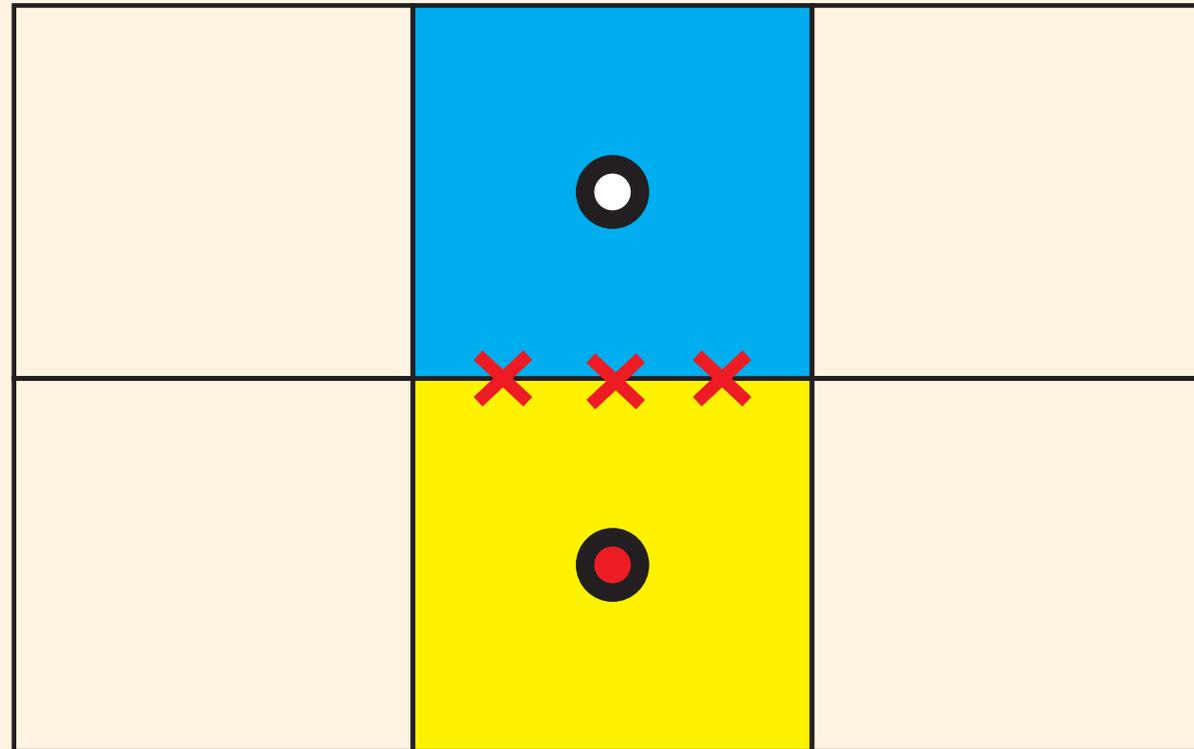
- Incompressible flow around a cross-flow turbine.





# High-order Fluid-Structure-Interaction techniques

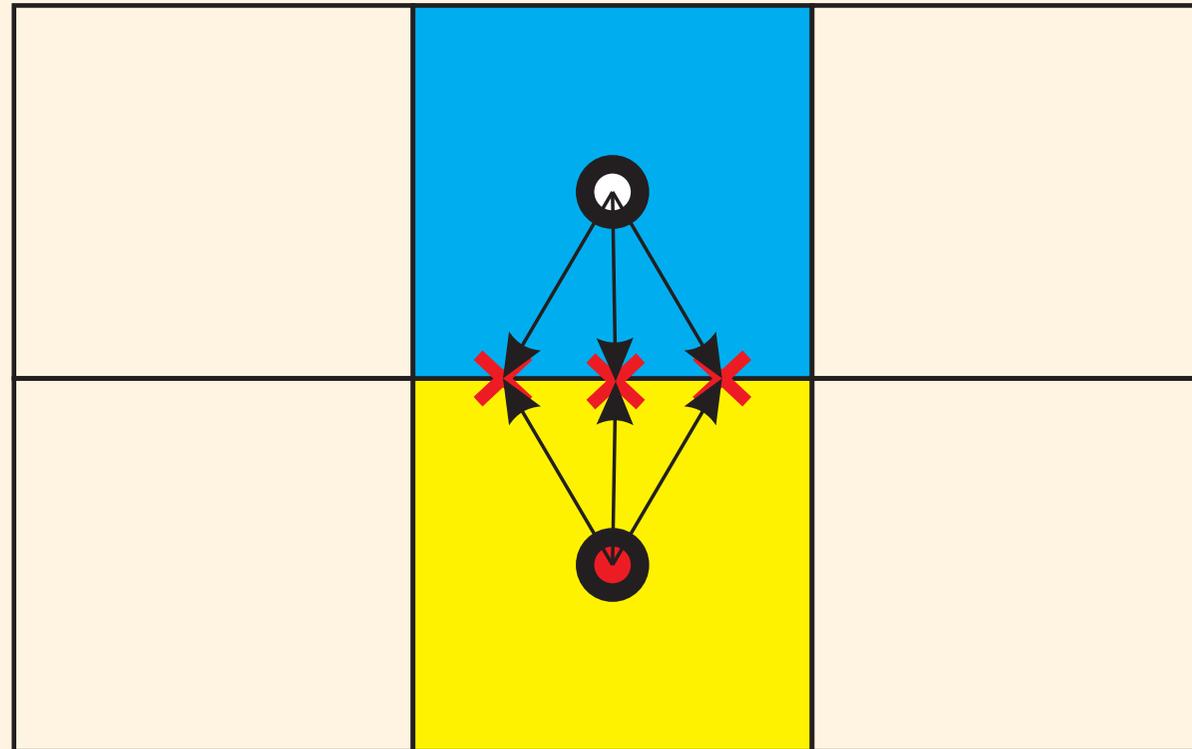
## ► High-order Sliding Mesh Techniques





# High-order Fluid-Structure-Interaction techniques

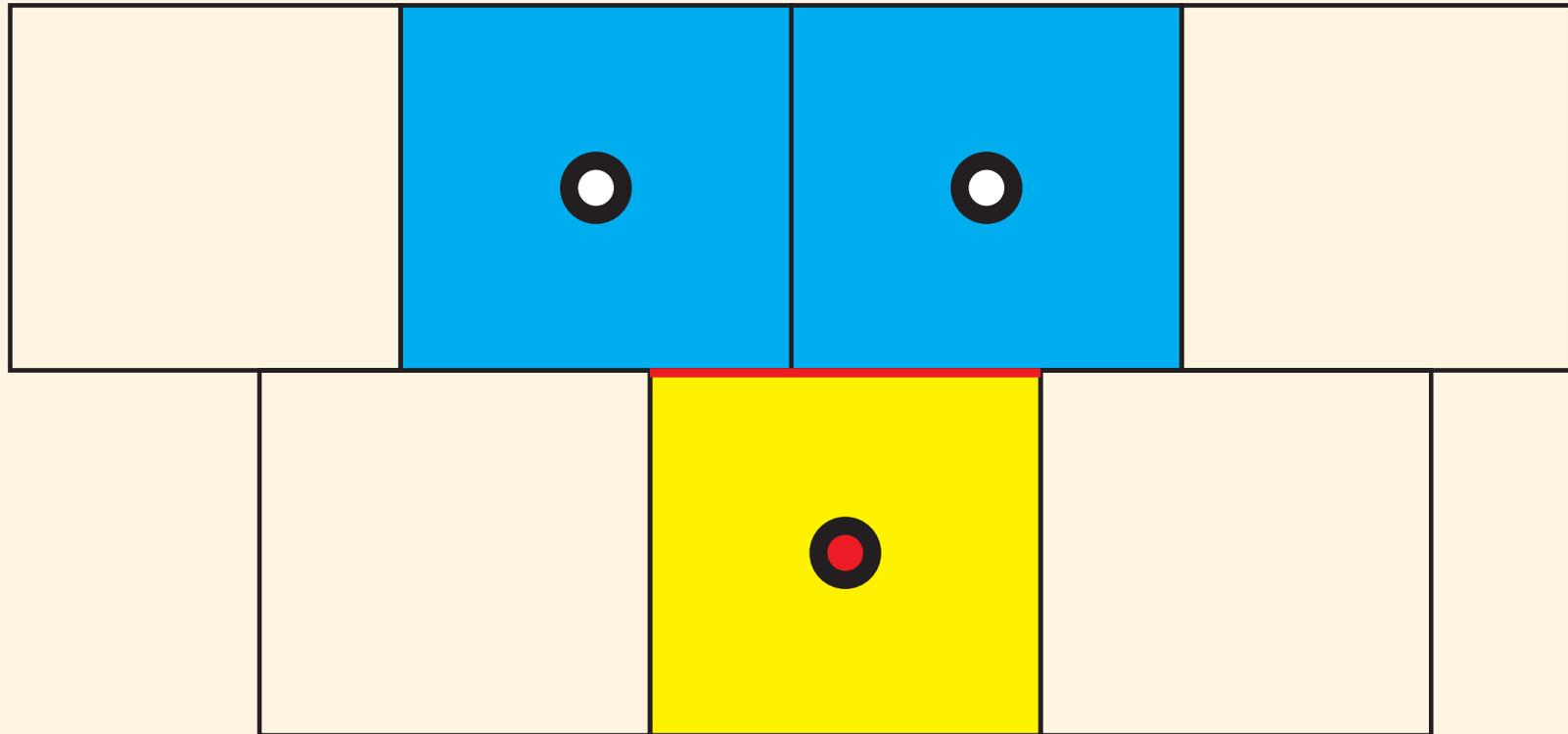
## ► High-order Sliding Mesh Techniques





# High-order Fluid-Structure-Interaction techniques

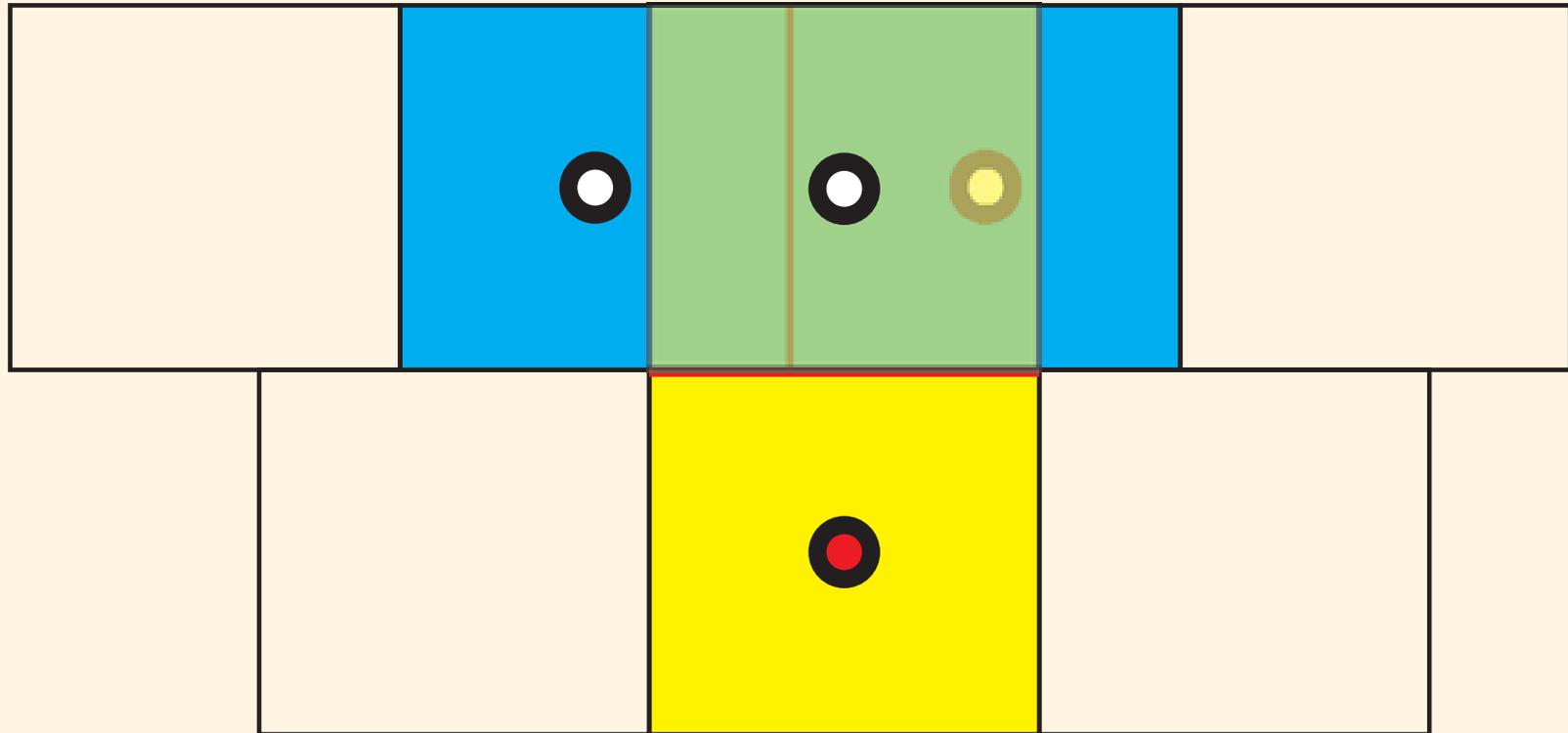
## ► High-order Sliding Mesh Techniques





# High-order Fluid-Structure-Interaction techniques

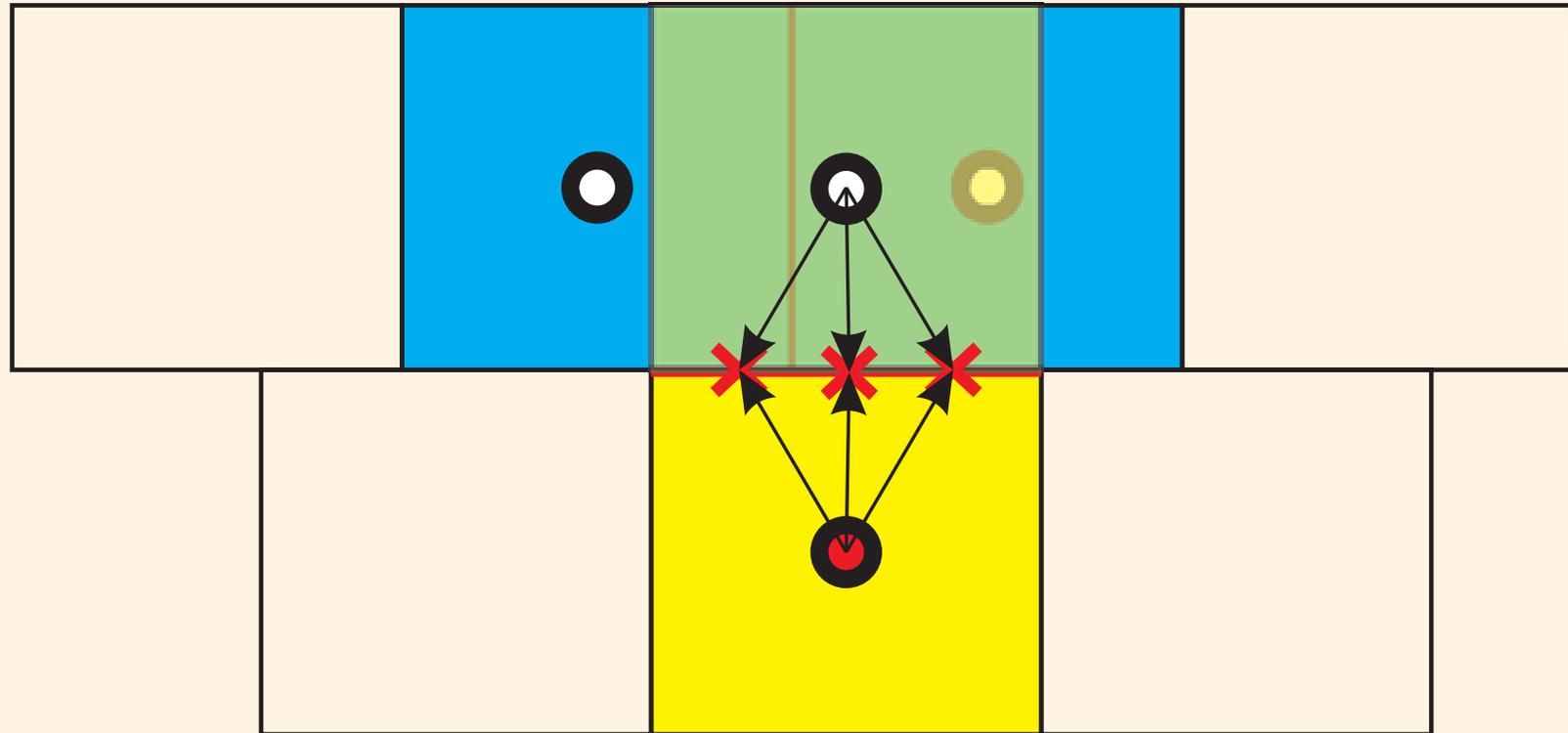
## ► High-order Sliding Mesh Techniques





# High-order Fluid-Structure-Interaction techniques

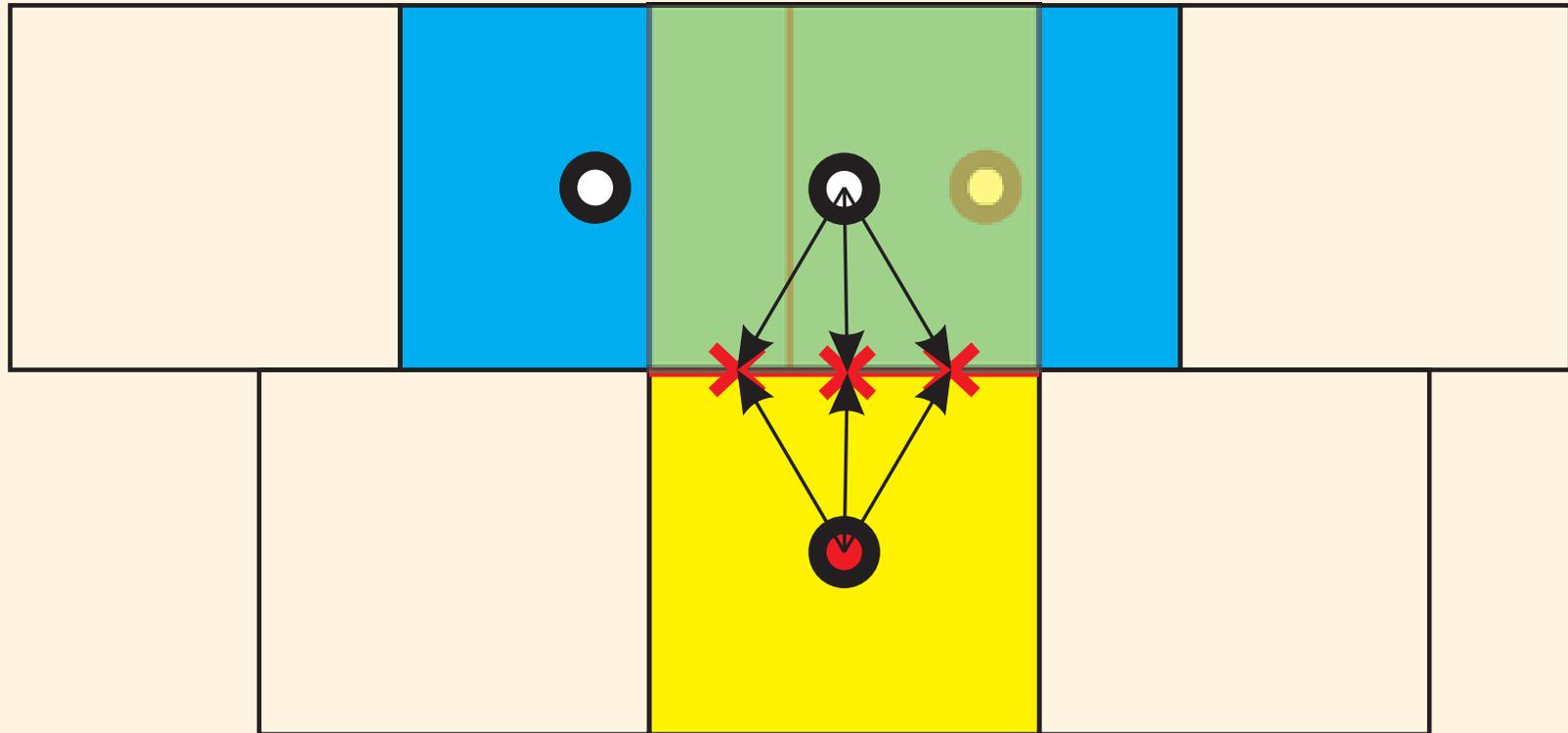
## ► High-order Sliding Mesh Techniques





# High-order Fluid-Structure-Interaction techniques

## ► High-order Sliding Mesh Techniques



It will be presented tomorrow

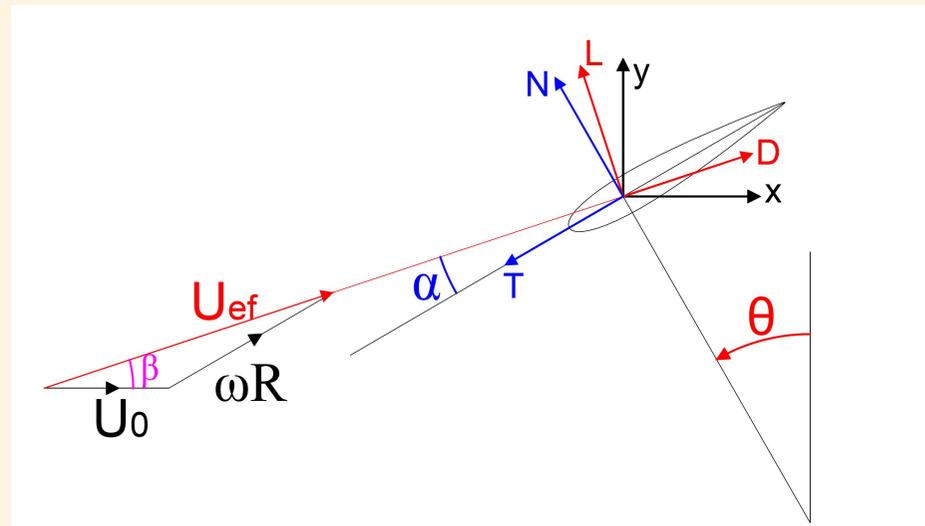
SHARK-FV 2015 Conference  
SHARING HIGHER-ORDER ADVANCED RESEARCH KNOW-HOW on FINITE VOLUME  
Ofir, Portugal  
May 18 - 22, 2015





# Numerical Examples

- Incompressible flow around a cross-flow turbine.



$$\vec{f} = \begin{Bmatrix} f_x \\ f_y \end{Bmatrix} = \oint (p\vec{n} - \nu(\nabla\vec{U} \cdot \vec{n}))d\Gamma$$

$$f_N = f_y \cos\theta - f_x \sin\theta \quad f_T = -f_x \cos\theta - f_y \sin\theta$$



# High-order Fluid-Structure-Interaction techniques

## ► Fluid-Structure Interaction (FSI)

- Flow driven approach  $\rightarrow \omega$  given by the fluid

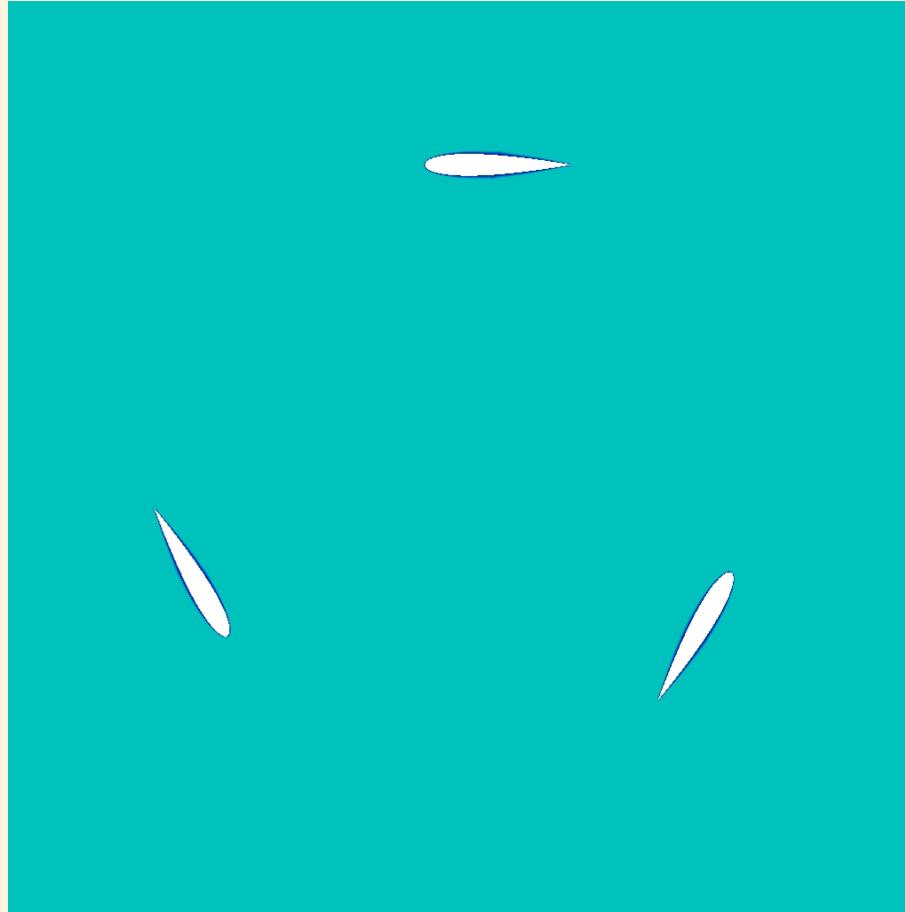
$$\omega^{n+1} = \omega^n + \frac{(T - M) \Delta t}{J}$$

- $T \rightarrow$  Torque
- $M \rightarrow$  Loading Moment
- $\Delta t \rightarrow$  Time step
- $J \rightarrow$  Mass moment of inertia





### ► Fluid-Structure Interaction (FSI)





# Conclusions

- Introduction
- The FV-MLS method
- A high-order formulation for incompressible flows
- High-order Fluid-Structure-Interaction techniques
- **Conclusions**





## Conclusions

- ▶ We have proposed a new higher-order accurate FV formulation for the numerical solution of incompressible fluid flows on unstructured meshes.
- ▶ We have modified the usual linear formulation of MIM to introduce higher-order approximations using MLS.
- ▶ The proposed methodology obtains excellent results.
- ▶ This methodology can be easily included in existing finite volume codes which represents an additional advantage.





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# HIGHER-ORDER FV-MLS METHOD FOR THE INCOMPRESSIBLE NAVIER-STOKES EQUATIONS

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Thank you

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